



GNSS, SBAS essentials and Ionospheric Monitoring in detail

*Based and updated from presentations prepared and performed
by M. Hernández-Pajares*

UPC-IonSAT res. group

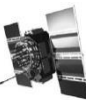
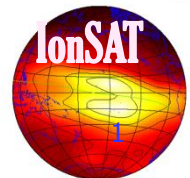
[contact e-mail : manuel@ma4.upc.edu]

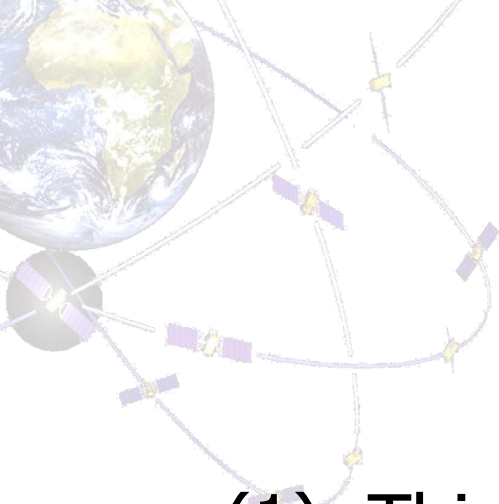
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(Colloquium On Atmospheric Remote Sensing Using The Global Positioning System, Boulder, CO, USA, June 20-July 2, 2004; Satellite Engineering Summer School, UPM, La Granja, Segovia, Spain, July 16, 2009; 40th COSPAR Scientific Assembly 2014, August 2-10, 2014, Moscow, Russia, and contributions from M. Hernández-Pajares, J.M.Juan, J.Sanz, M. García-Fernández, R. Orús-Pérez, P.Ramos, A. Aragón-Ángel, D.Salazar, A.García-Rigo, G. Olivares, among others)



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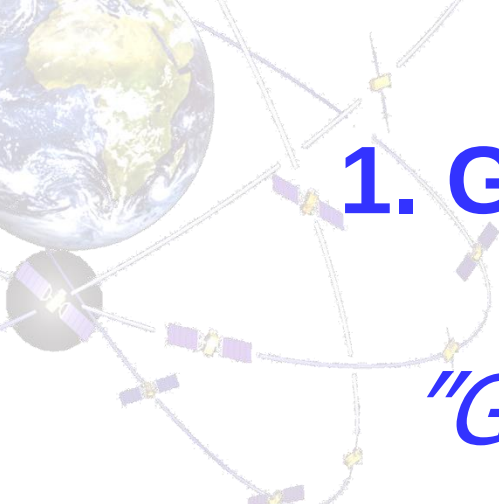


Outline

- (1) Thin-shell model, local model, global model
- (2) L4 linear combination, parameter estimation
- (3) Result examples, Errors / Quality

Outline

- Introduction
- The Ionosphere
- Ionospheric effects on GPS.
- From Ionosphere First principles to GPS preprocessing
- Obtaining the electron content
 - ✓ Using ground-based GNSS data (VTEC)
 - ✓ Using space-based GNSS data as well (Ne)
 - Space-based and ground-based GNSS data
 - Ground-based GNSS and Ionosonde data.
- Ionospheric determination at regional/continental scale
- Determining the electron content at global scale
- Application Example: WARTK and WARTK-3
- Conclusions

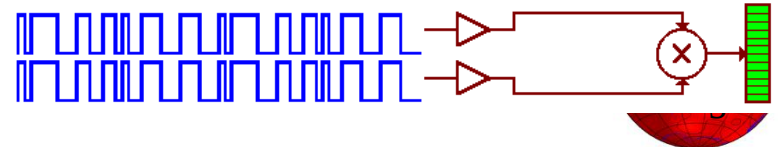
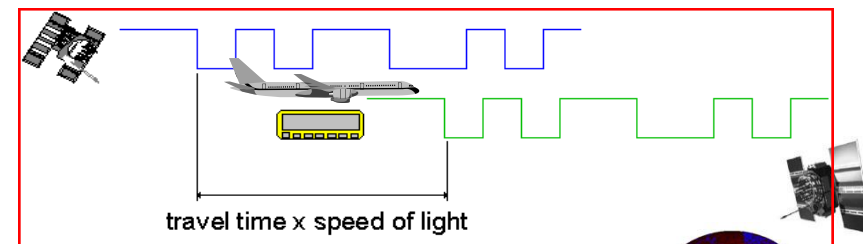
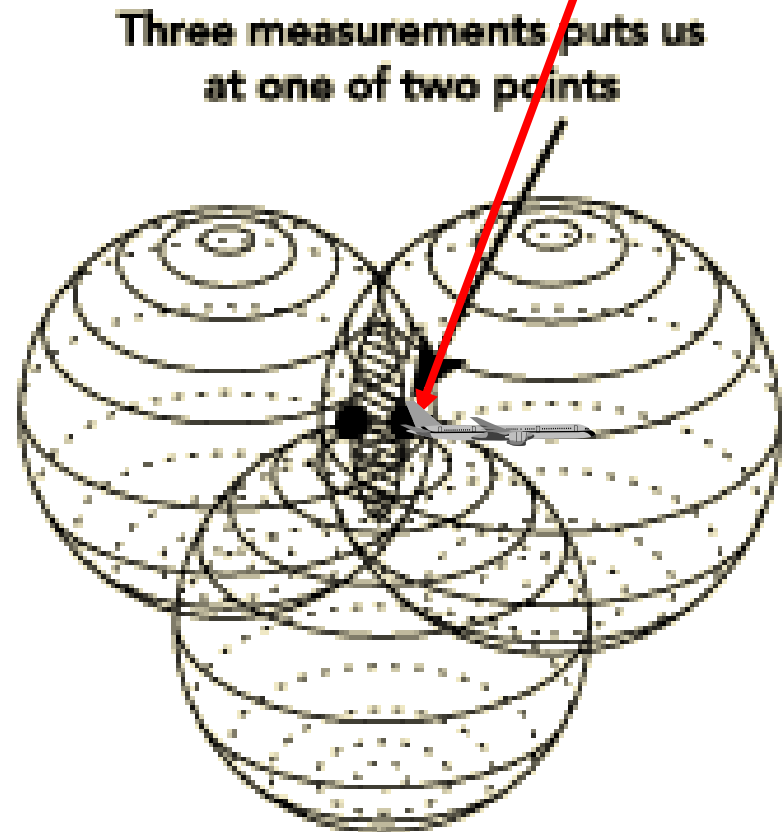
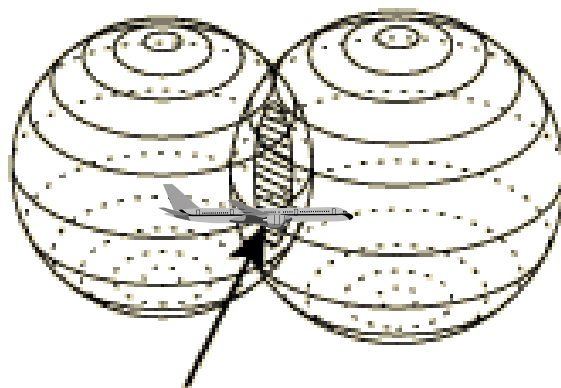
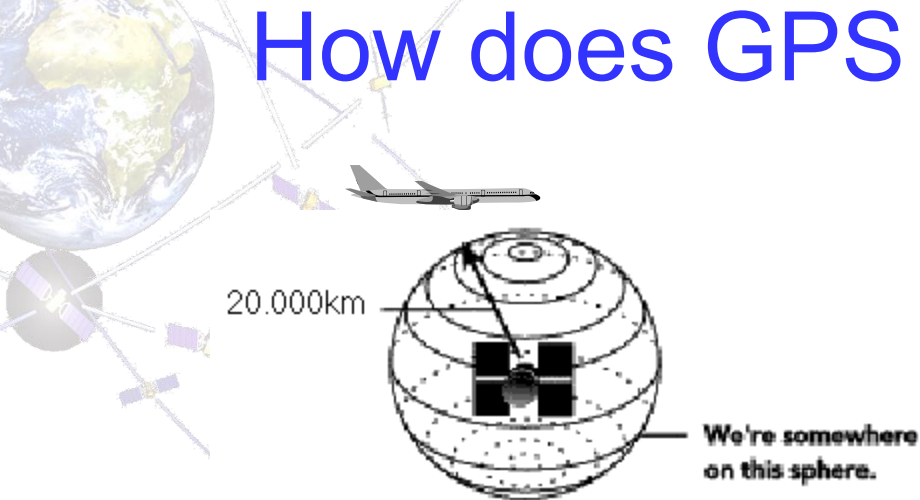


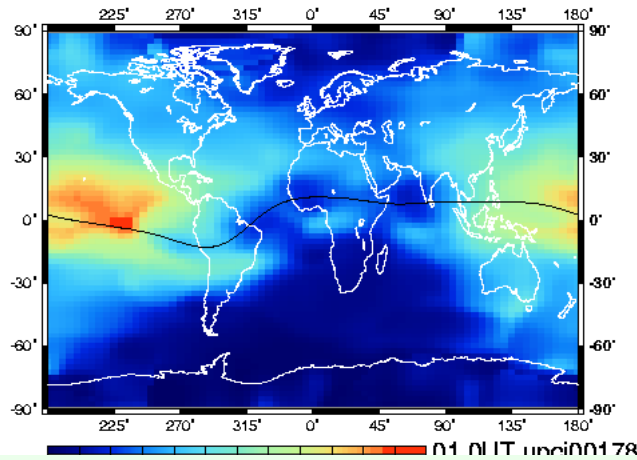
1. Global Navigation Satellite Systems essentials

"GNSS: All time, all weather, anywhere, passive electromagnetic trilateration; supported by many recent advances of Telecommunications, Mathematics and Physics"

How does GPS work?"

One of the solutions is not on the Earth surface.





Atmospheric propagation: IONO, TROPO



Relativistic effects

Main error sources

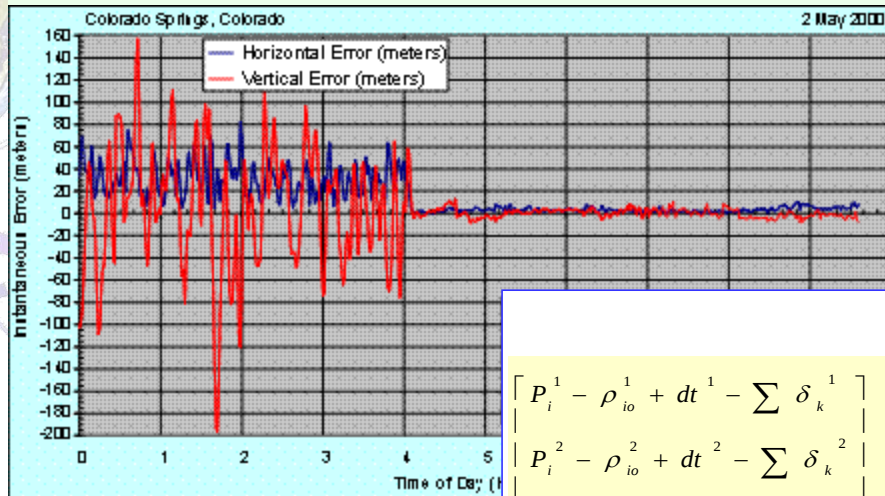
- GPS clocks $\ll 300$ km
- Relativistic effect < 13 m
- Instrumental delays ~ 1 m
- Ionospheric delay 2-50 m
- Tropospheric delay 2-10 m
- Multipath < 5 m
- Thermal noise < 2 m

MODEL:

Atmospheric propag., relativistic effects, clocks and instrum. delays are modeled and removed.

And the navigation equations are built

How satellite navigation systems work



$$\begin{bmatrix} P_i^1 - \rho_{io}^1 + dt^1 - \sum \delta_k^1 \\ P_i^2 - \rho_{io}^2 + dt^2 - \sum \delta_k^2 \\ \dots\dots\dots \\ P_i^n - \rho_{io}^n + dt^n - \sum \delta_k^n \end{bmatrix} =$$

$$\begin{bmatrix} \frac{x_{io} - x^1}{\rho_{io}^1} & \frac{y_{io} - y^1}{\rho_{io}^1} & \frac{z_{io} - z^1}{\rho_{io}^1} & 1 \\ \frac{x_{io} - x^2}{\rho_{io}^2} & \frac{y_{io} - y^2}{\rho_{io}^2} & \frac{z_{io} - z^2}{\rho_{io}^2} & 1 \\ \dots\dots\dots \\ \frac{x_{io} - x^n}{\rho_{io}^n} & \frac{y_{io} - y^n}{\rho_{io}^n} & \frac{z_{io} - z^n}{\rho_{io}^n} & 1 \end{bmatrix}$$

$$\begin{bmatrix} \Delta x_i \\ \Delta y_i \\ \Delta z_i \\ cdt_i \end{bmatrix}$$

Navigation equations

The geometric problem is linearized, and Weighted Least Mean Squares or Kalman filter are used to compute the solution.

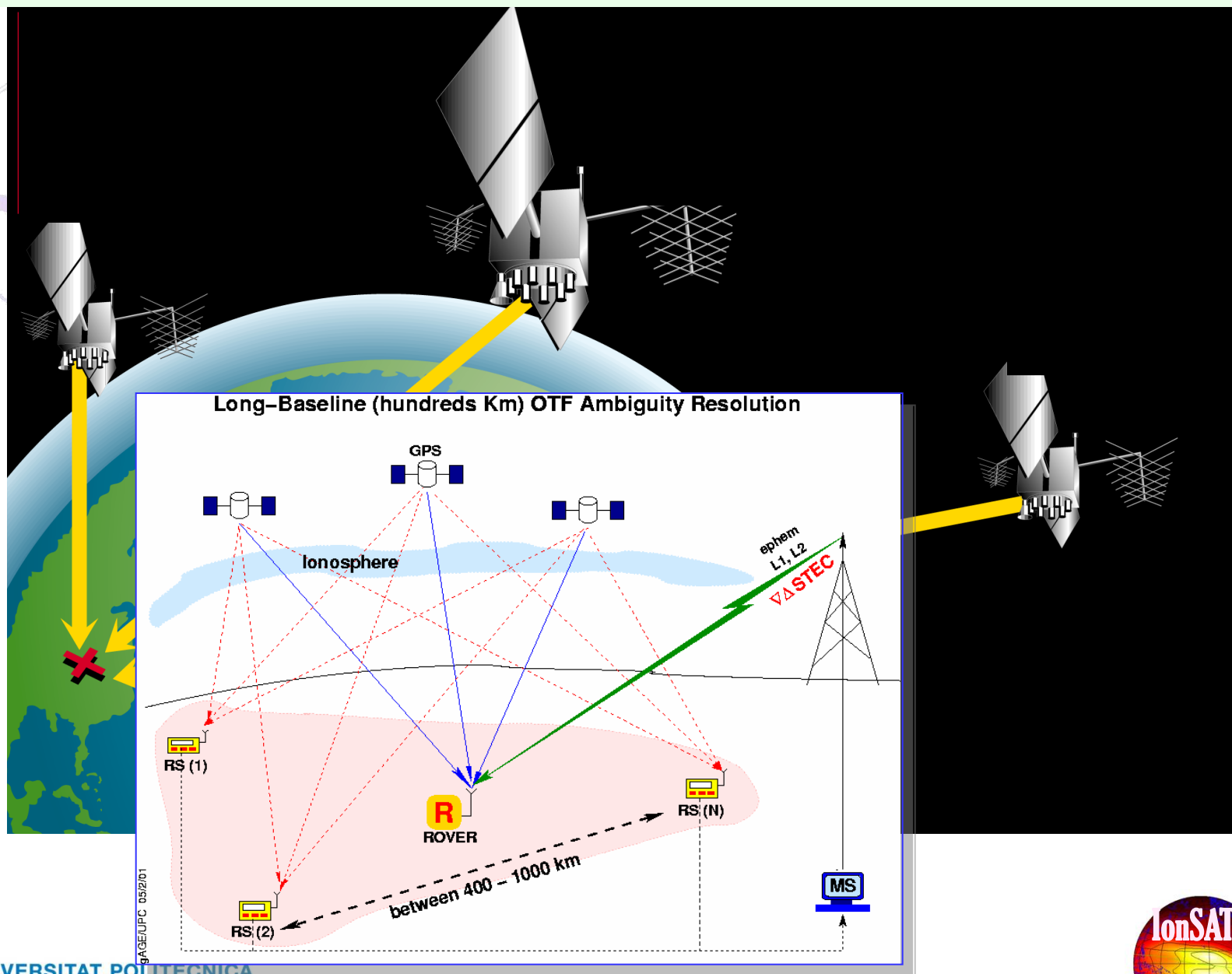
MODEL:

Atmospheric propagation, relativistic effects, clocks and instrument delays are modeled and removed.

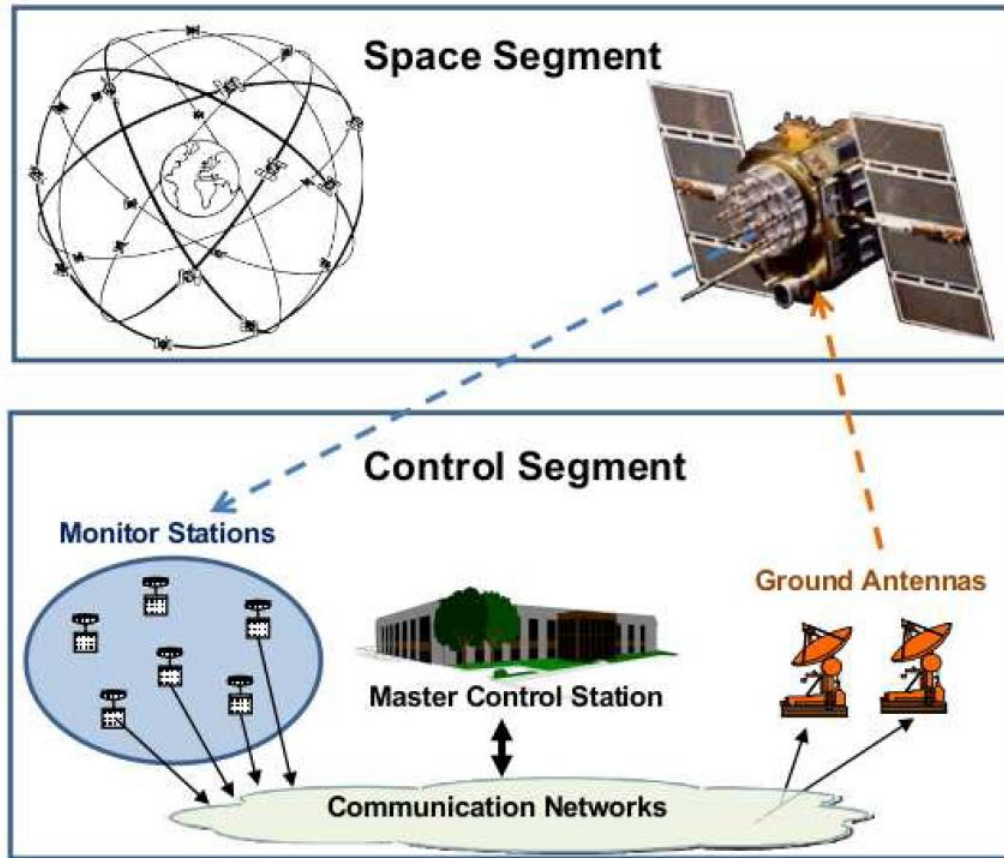
And the navigation equations are built



How satellite navigation systems work



GPS segments



Space segment: Transmitting GPS satellites as excellent reference points (the orbits are highly predictable)

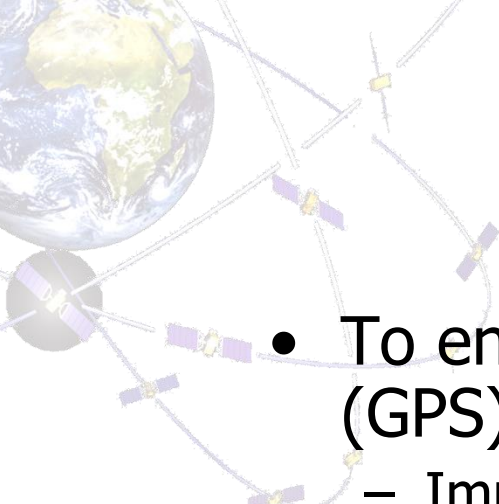
Control segment: Space State representation (providing clocks and orbits to the users, instead of raw data of the control segment stations –more suitable for worldwide usage-)

User segment: receivers from single-frequency code to dual-frequency phase and code measurements.



2. Essentials of the GNSS Augmentation concept, including EGNOS

*“Giving navigation integrity , at
civil aviation standards, by means
of a control segment improved at
continental scale”*



Augmentation concept

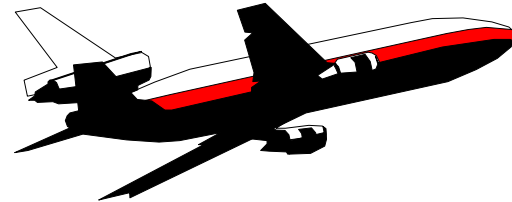
- To enhance the performance of the current GNSS (GPS) with additional information to:
 - Improve INTEGRITY via real-time monitoring
 - Improve ACCURACY via differential corrections
 - Improve AVAILABILITY and CONTINUITY
- Satellite Based Augmentation Systems (**SBAS**)
 - E.g., WAAS, **EGNOS**, MSAS
- Ground Based Augmentation Systems (GBAS)
 - E.g., LAAS
- Aircraft Based Augmentation (ABAS)
 - E.g., RAIM, Inertials, Baro Altimeter



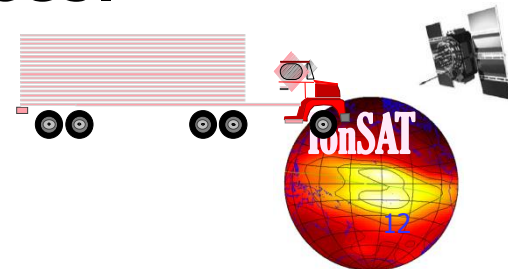
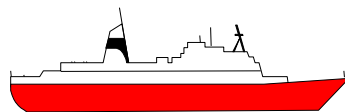
Why Augmentation Systems?

- Current GPS/GLONASS Navigation Systems cannot meet the Requirements for All Phases of Flight:

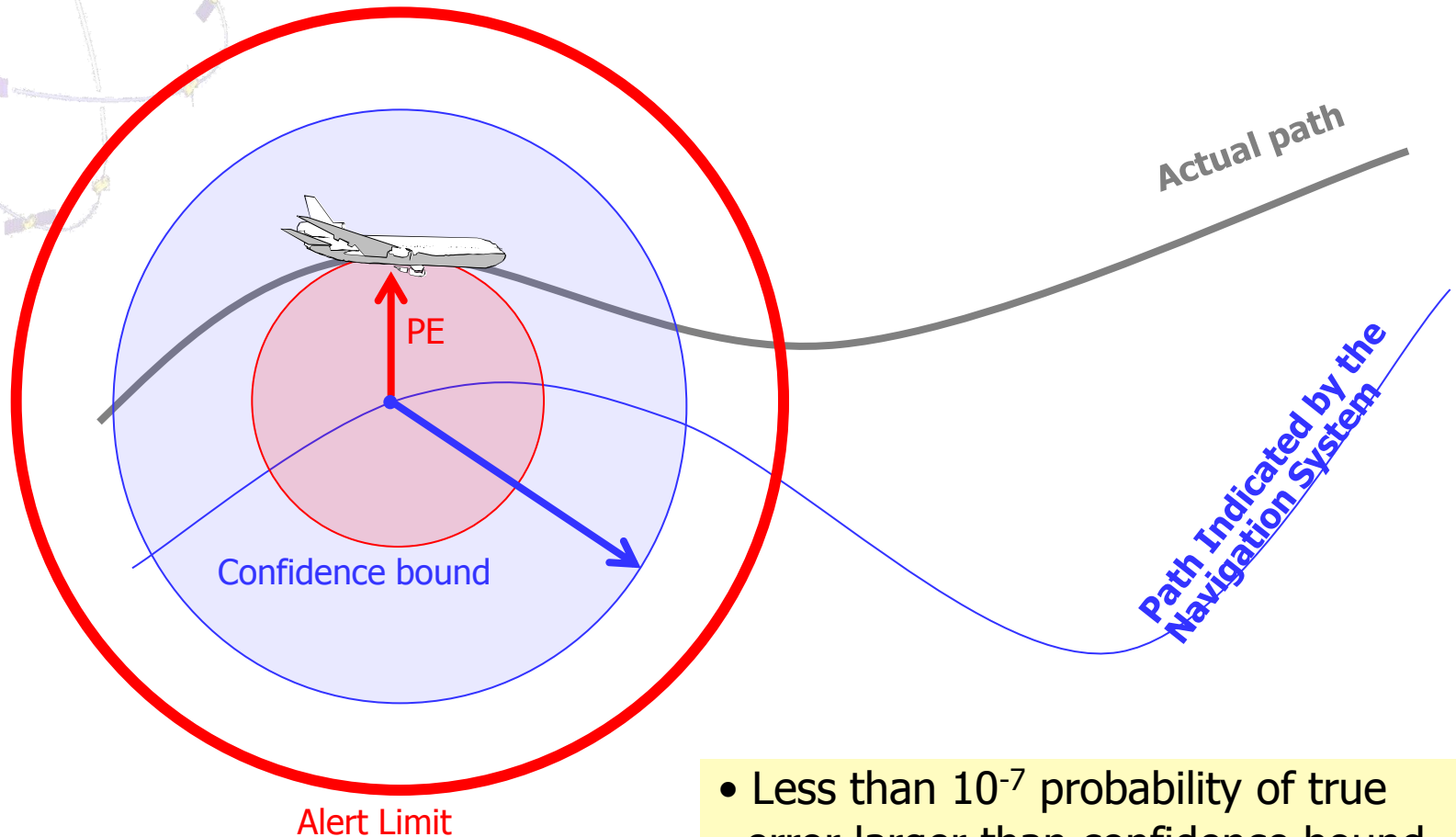
- Accuracy
- Integrity
- Continuity
- Availability



- Marine and land users will also require some sort of augmentation for improving the GPS/ GLONASS performances.



INTEGRITY

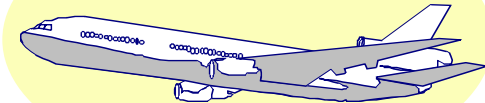


WHY GPS NEEDS AN AUGMENTATION ?

GPS Only



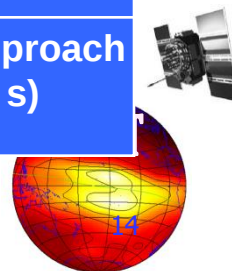
Civil Aviation



PERFORMANCE

CATEGORY I Requirements

H. 13 m V. 22m	ACCURACY (95%)	H 16.0 m V 4.0 m
99% (RAIM)	AVAILABILITY	99% to 99.990%
?	INTEGRITY RISK	$2 \cdot 10^{-7}$ / approach Time to alarm 6 s
?	CONTINUITY OF SERVICE	Risk: 10^{-5} / approach (10^{-6} / 15 s)



ERRORS on the GNSS Signal

- Space Segment Errors:

- Clock errors
- Ephemeris errors

Common

- Propagation Errors

- Ionospheric delay
- Tropospheric delay

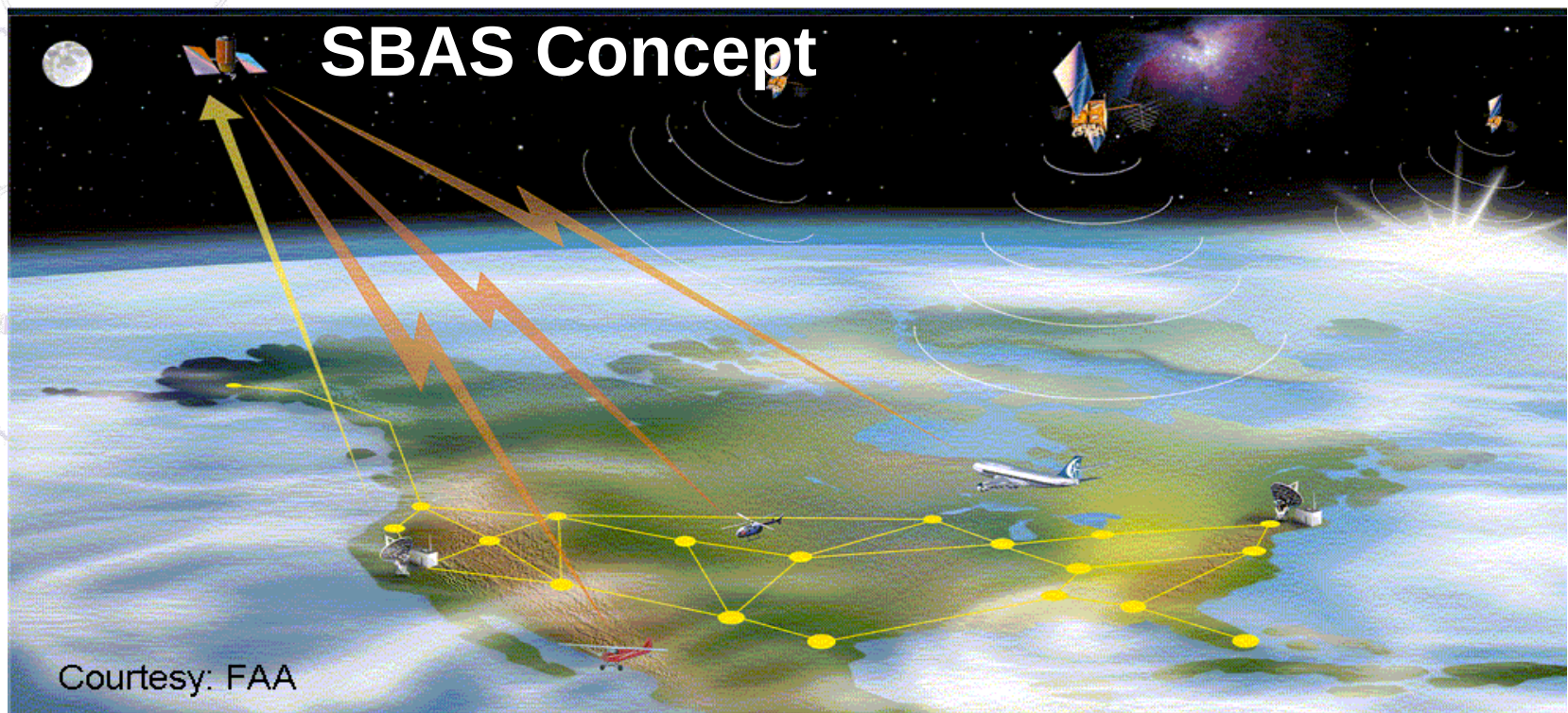
Strong spatial correlation

Weak spatial correlation

- Local Errors

- Multipath
- Receiver noise

No spatial correlation



Courtesy: FAA

Algorithms improvement (ensuring integrity) and augmented control segment in SBAS area (such as WAAS in USA, EGNOS in Europe, MSAS in Japan, GAGAN in India).

The pseudorange error is splitted in its components.

Clock error

Ephemeris error

Ionospheric error

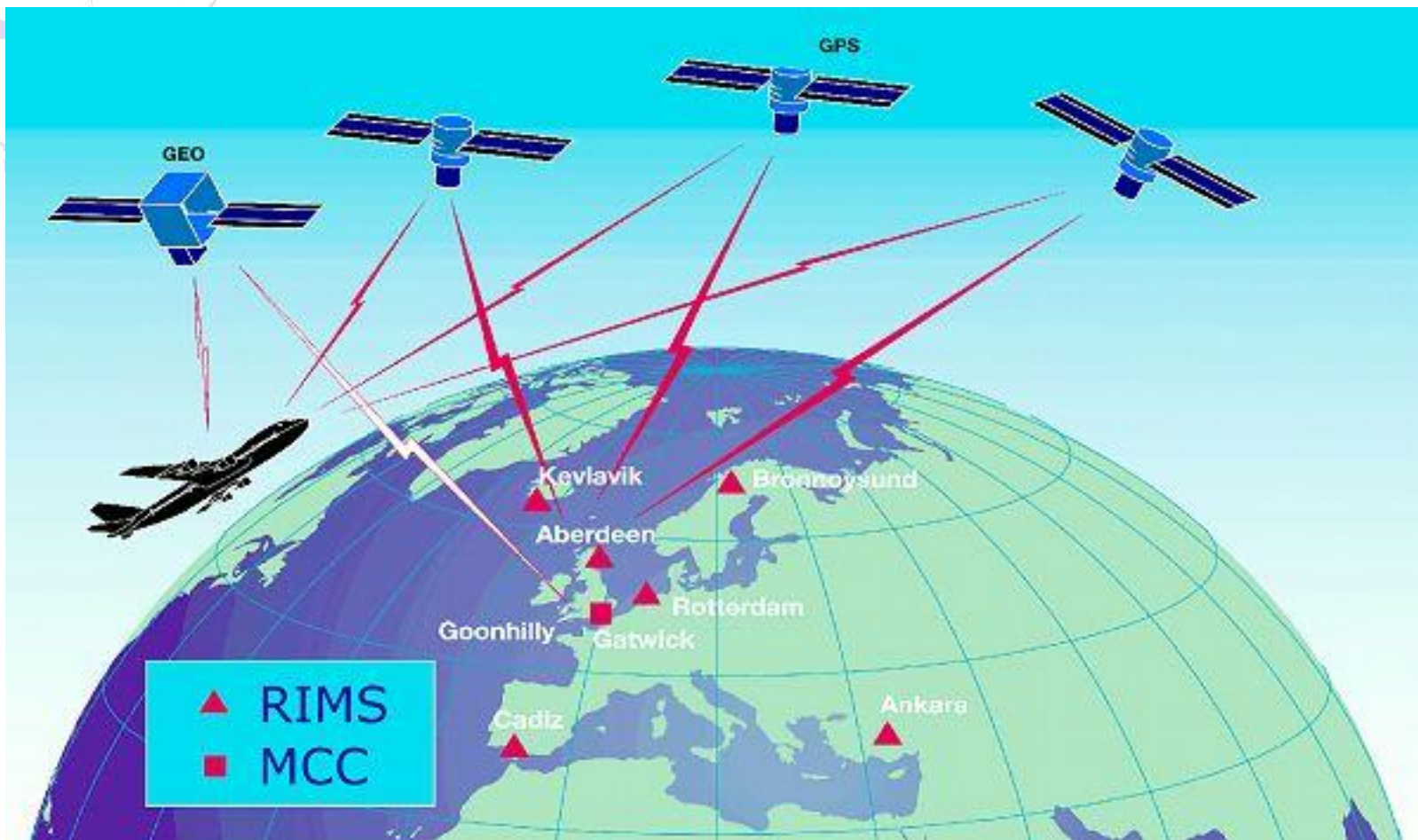
Local errors (troposphere, multipath, receiver noise)

Uses a network of receivers to cover broad geographic area

What is EGNOS?

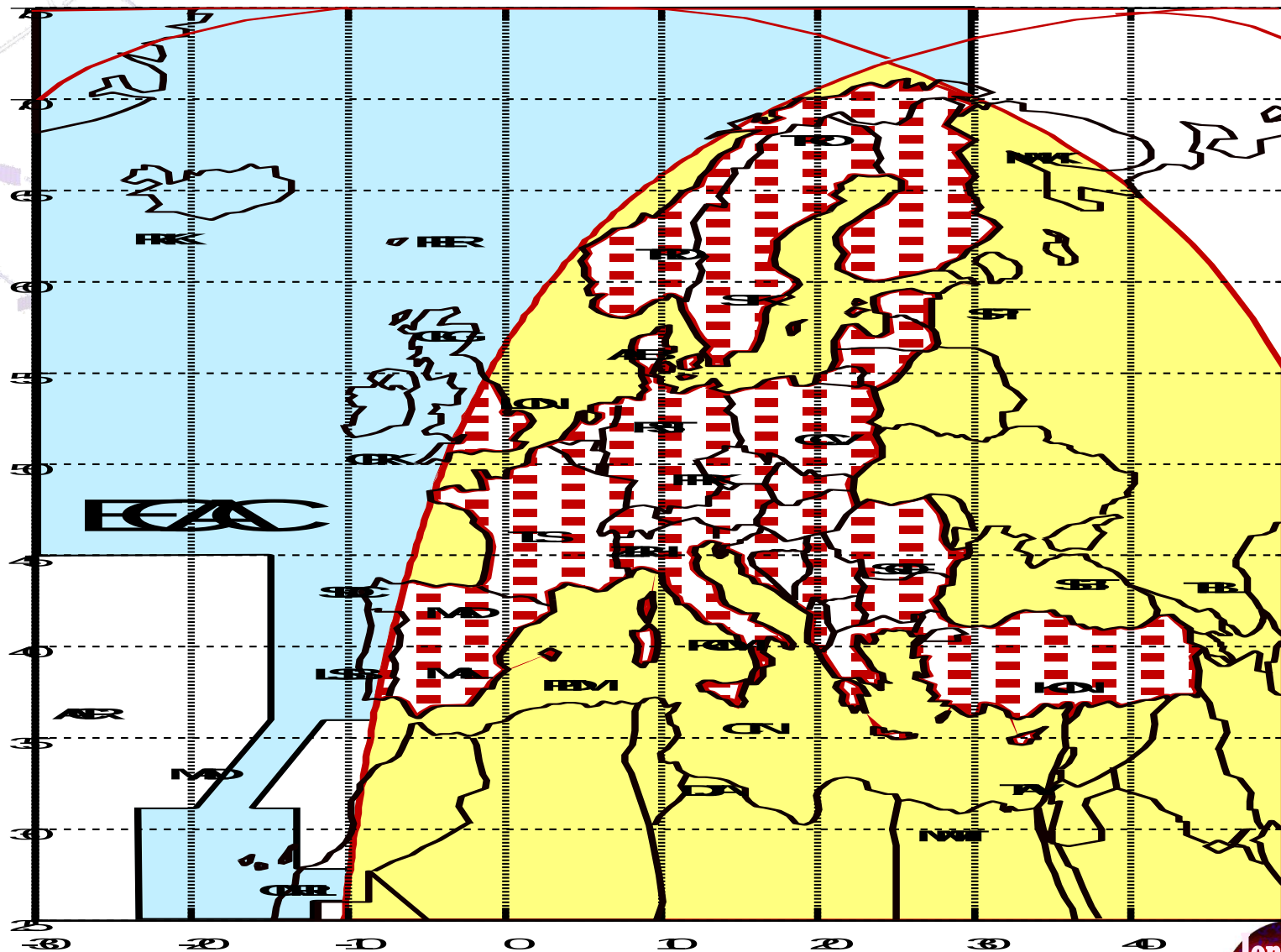
- EGNOS is the European component of a Satellite Based Augmentation to GPS and GLONAS.
- EGNOS has been developed under the responsibility of a tripartite group:
 - The European Space Agency (ESA)
 - The European Organization for the Safety of Air Navigation (EUROCONTROL)
 - The Commission of the European Union.

The European Geostationary Navigation Overlay SERVICE (EGNOS)



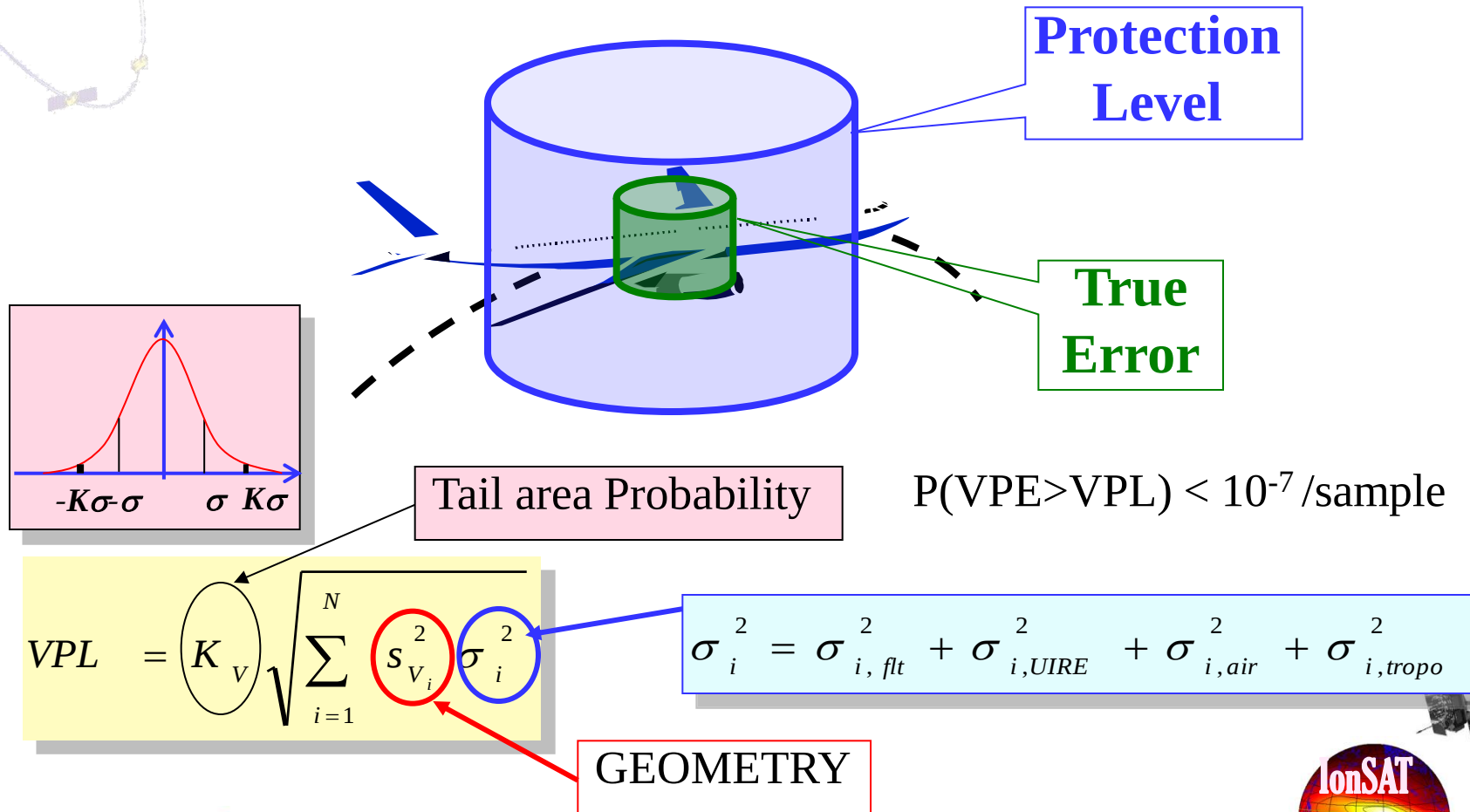


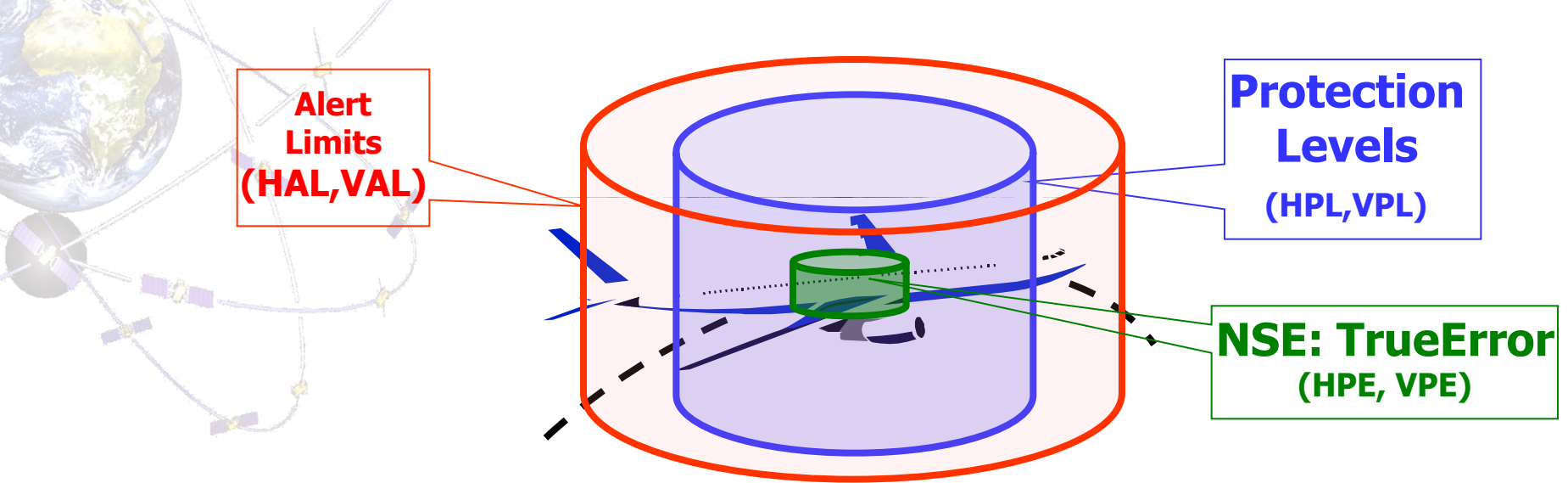
FRS NC NES ASIAE



Users know the receiver-satellites geometry and can compute bounds on the horizontal and vertical position errors.

These bounds are called Protection Levels (HPL and VPL). They provide good confidence (10^{-7} /hour probability) that the true position is within the corresponding cylinder/ellipsoid around the computed position.



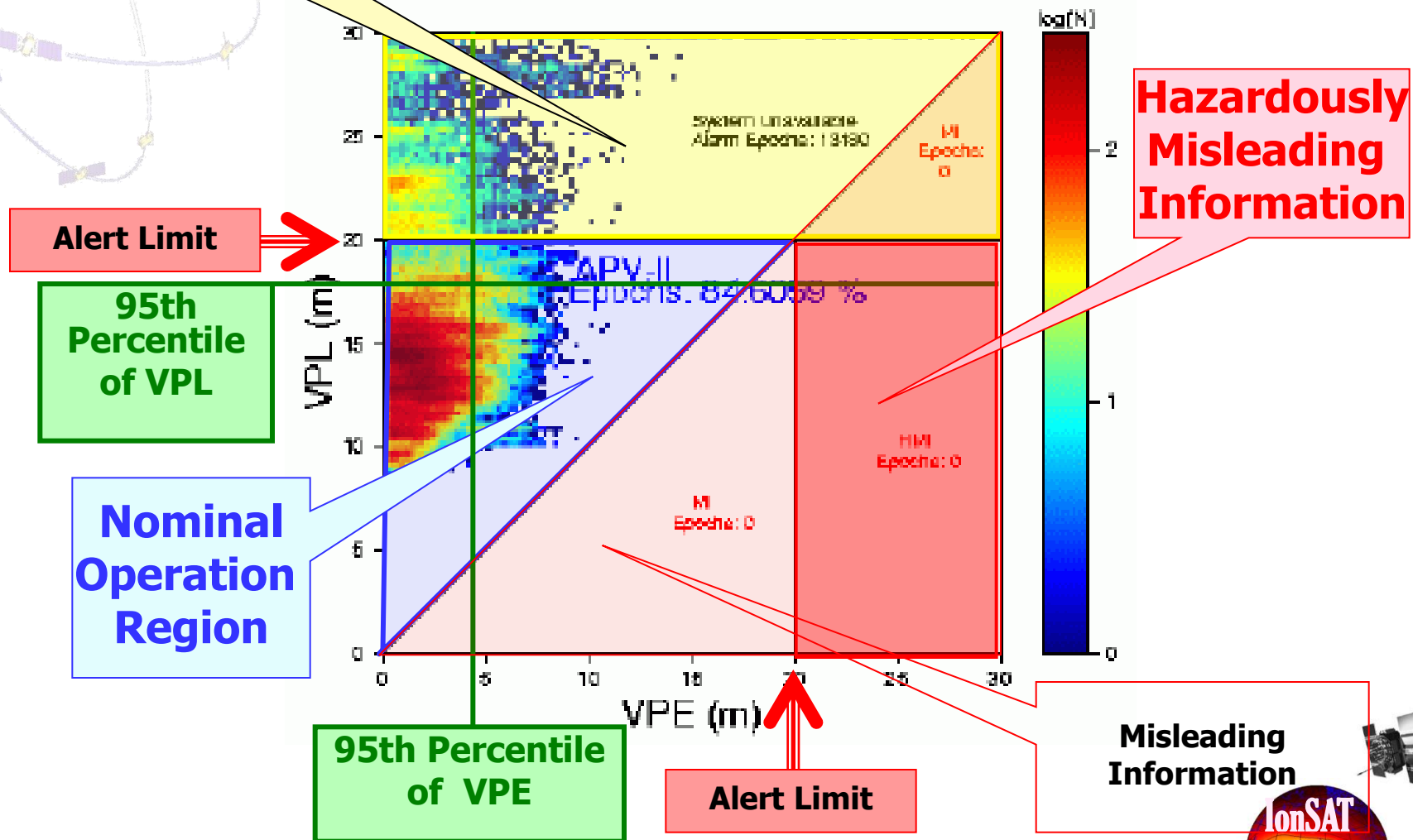


- Each epoch, HPL/VPL are compared with the Alert Limits (HAL/VAL) defined for the operation mode:
 - Hazardously Misleading Information (HMI): $XPE > XAL > XPL$
 $\rightarrow$ **INTEGRITY RISK**
 - Misleading Information (MI): $XPE > XPL > XAL$
 $\rightarrow$ **Out-Of-Tolerance cond.**
- The system is set unavailable when $XPL > XAL$
- Nominal operation: $XAL > XPL > XPE$

STANFORD PLOTS

**Alarm epochs
System Unavailable**

VERTICAL Perf. over 85617 epochs





3. Ionospheric monitoring with GNSS

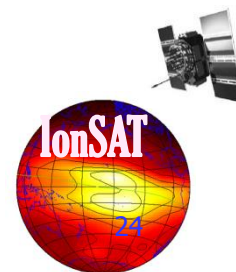
*“Determining in real-time the
electron content distribution with
an unprecedented spatial and
temporal resolution thanks to
GNSS”*



● Introduction

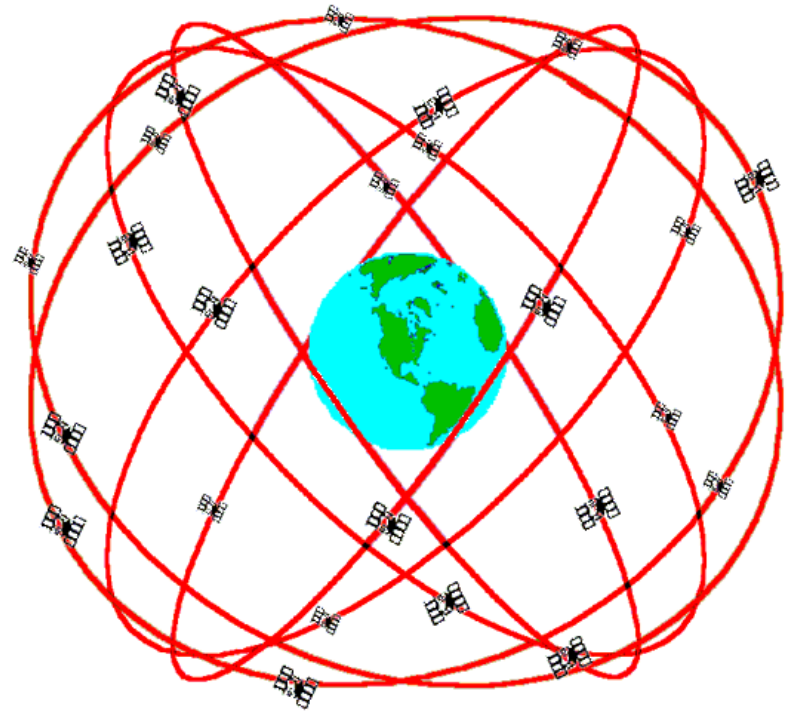


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Global Positioning System, GPS

- Constellation with 24 or more satellites orbiting at 20200km height.
- GPS satellites emit at 2 L-Band frequencies (1.6 and 1.2 GHz approx.)



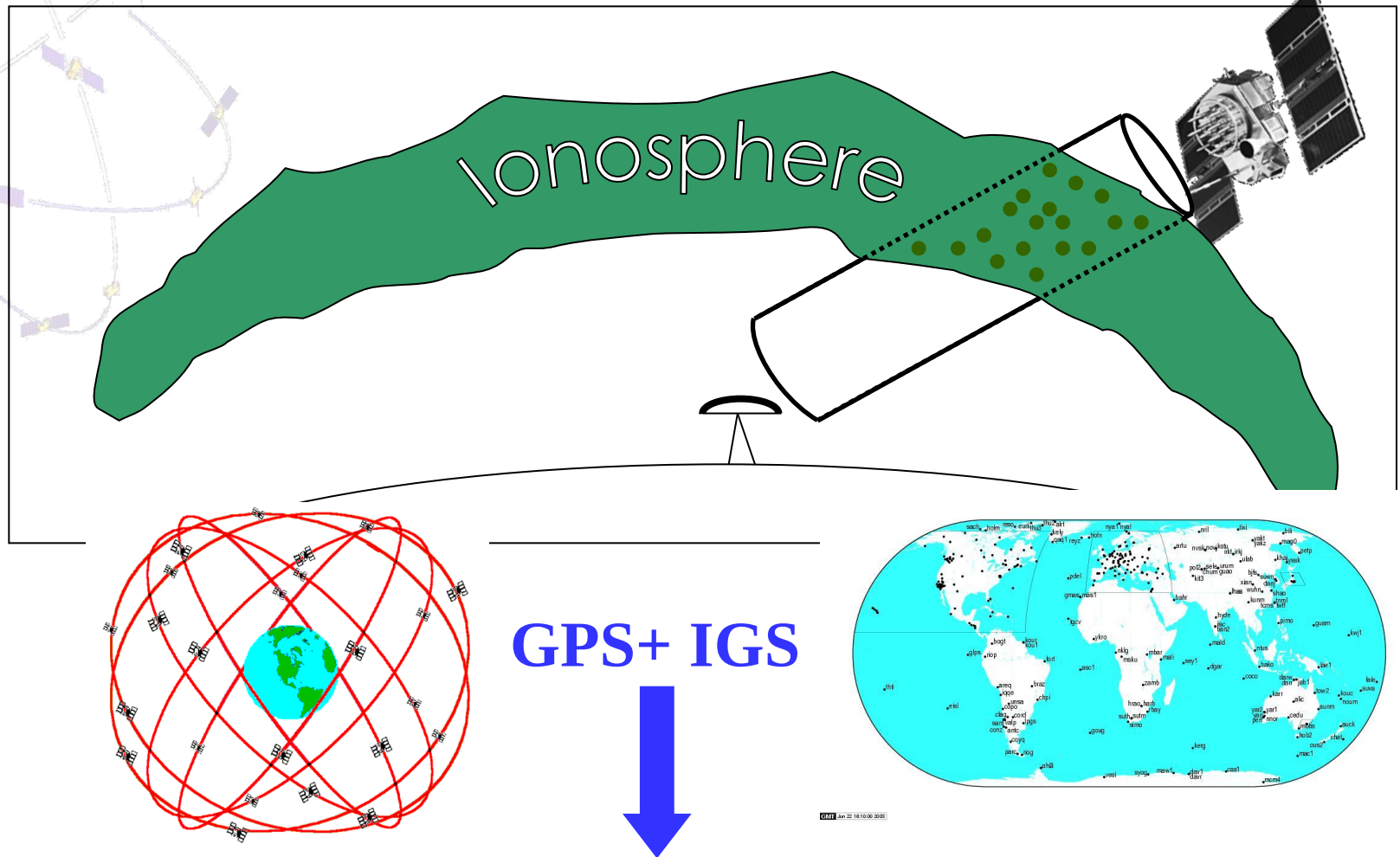
Ionosphere is
dispersive



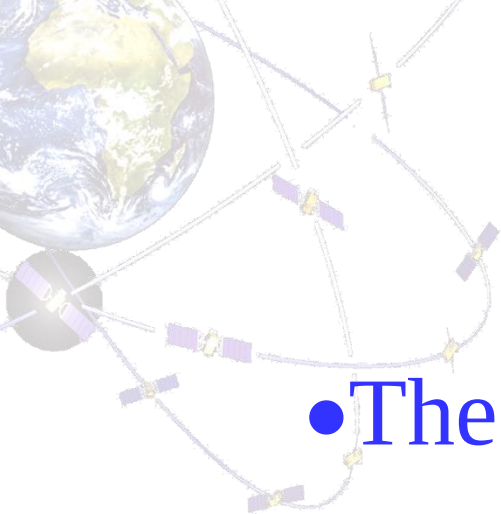
**Ionospheric effect:
signal delay/advance
observed from a receiver,
proportional to STEC**

GMT

GPS+ IGS: Global Iono. scanner



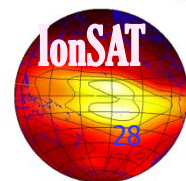
Worldwide scanner of the Ionosphere that can be used to generate global VTEC maps



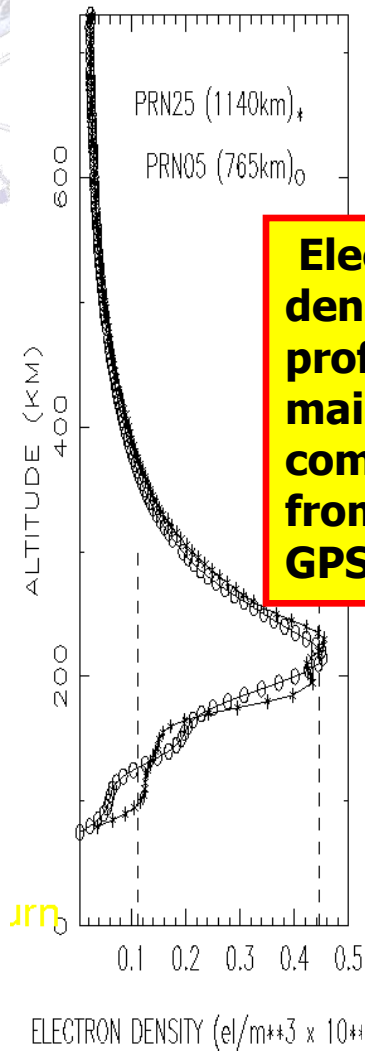
•The Ionosphere



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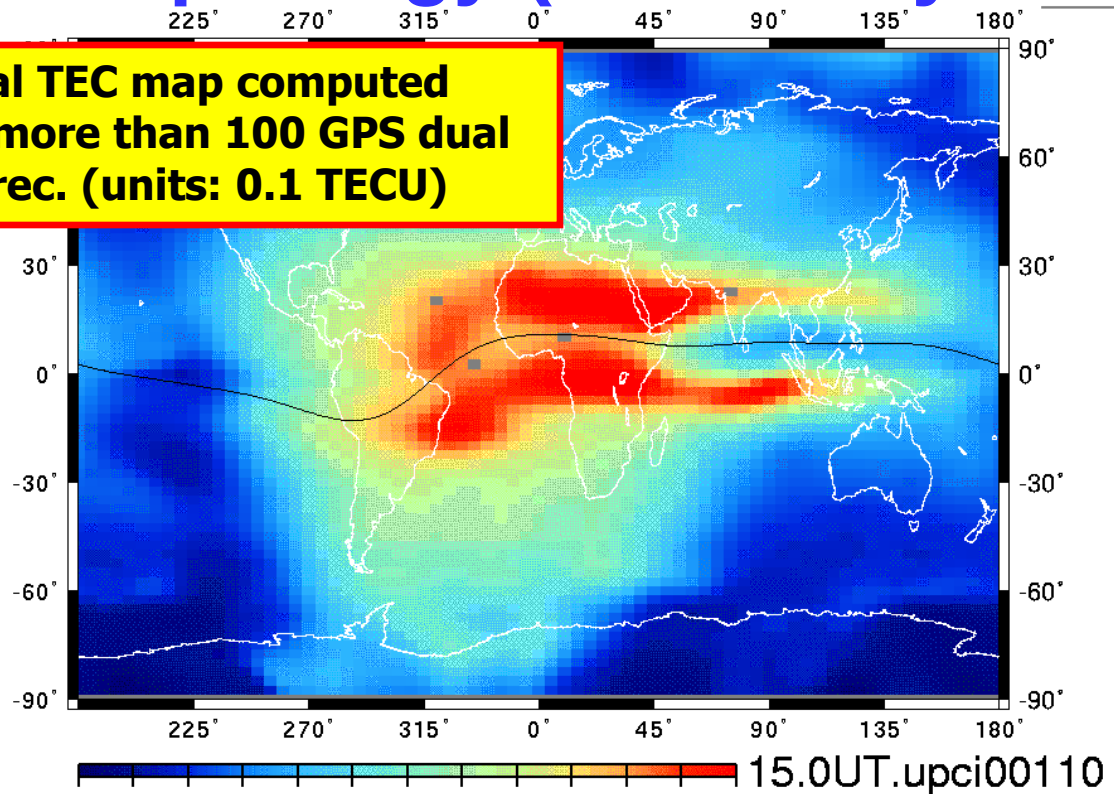


Ionosphere morphology (viewed by GPS)



Electron density profile mainly computed from LEO GPS data.

Global TEC map computed from more than 100 GPS dual freq. rec. (units: 0.1 TECU)

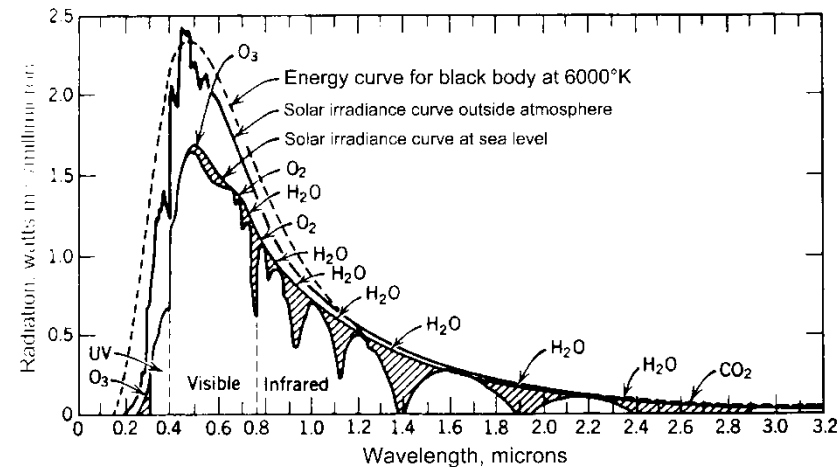
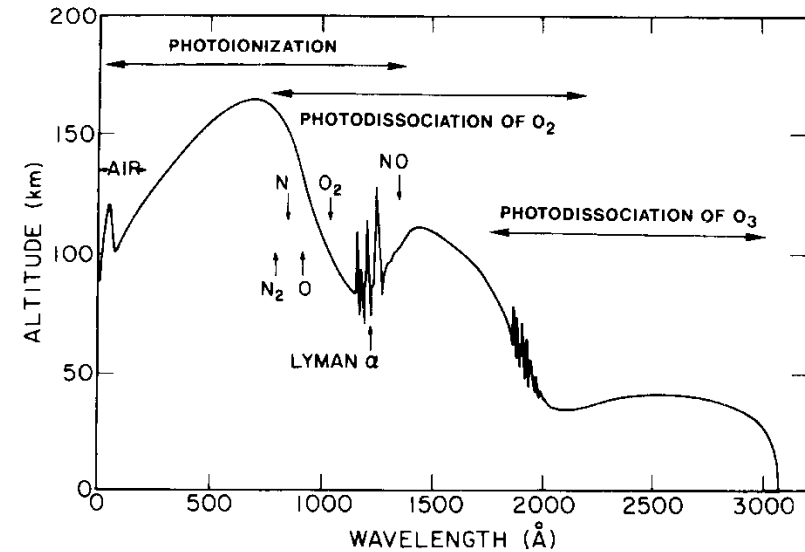


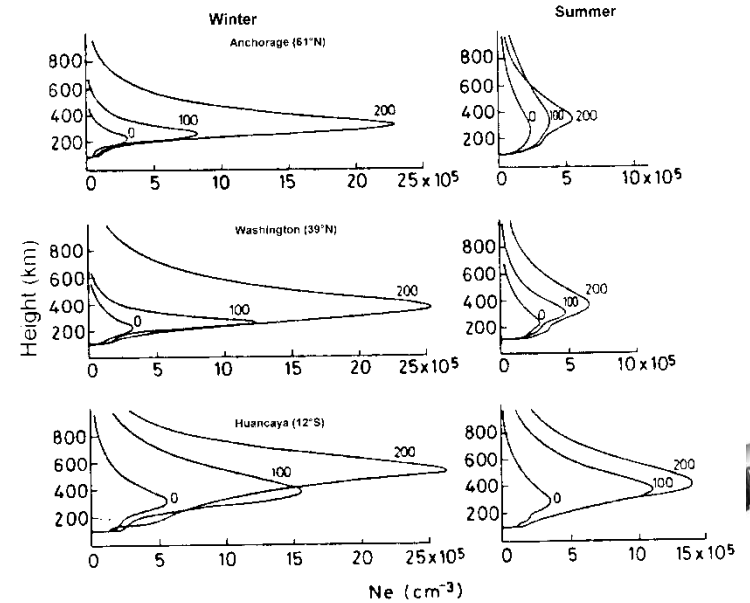
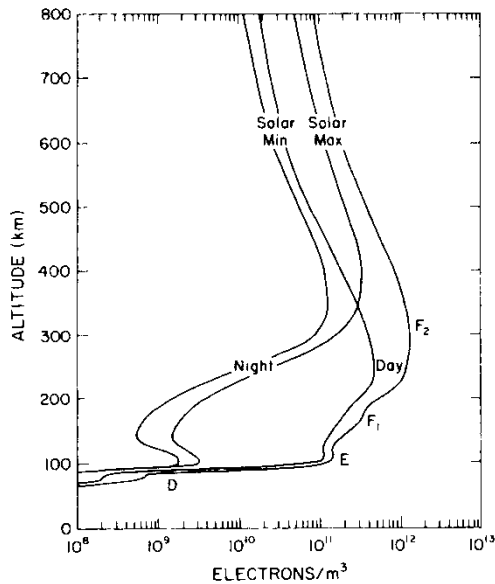
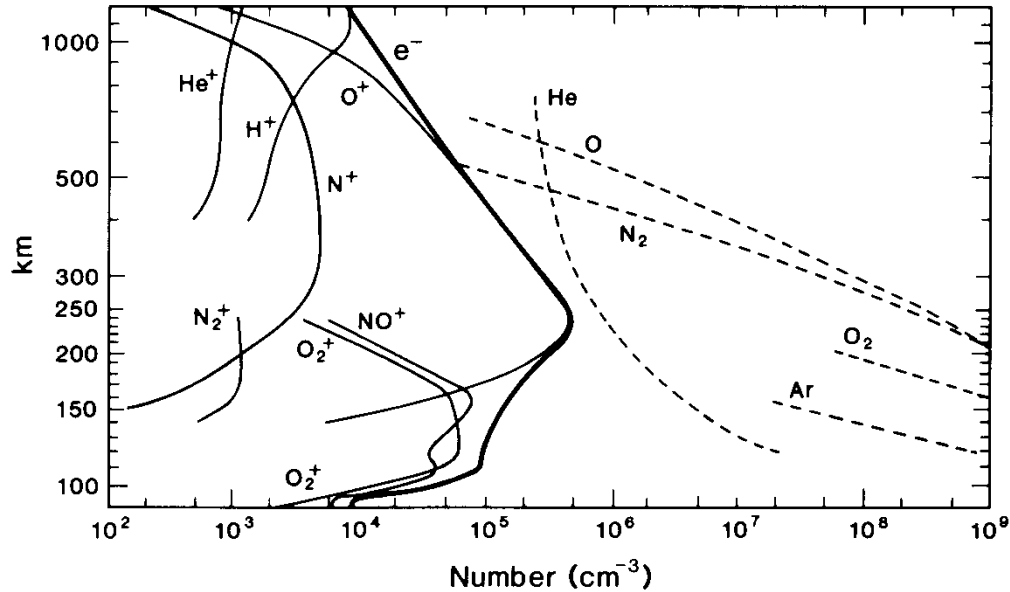
- GPS signal is very sensitive to free electron distribution, which respond to the EM field, oscillating and generating a secondary EM wave which overimposes and change the velocity of the GPS signals (main iono. efect on GPS code and phase).
- Vertical: Maximum density height at 200-400 km.
- Horizontally: Maximum density at both sides of the geomagnetic equator (equatorial anomalies).

The photoionization and the Ionosphere

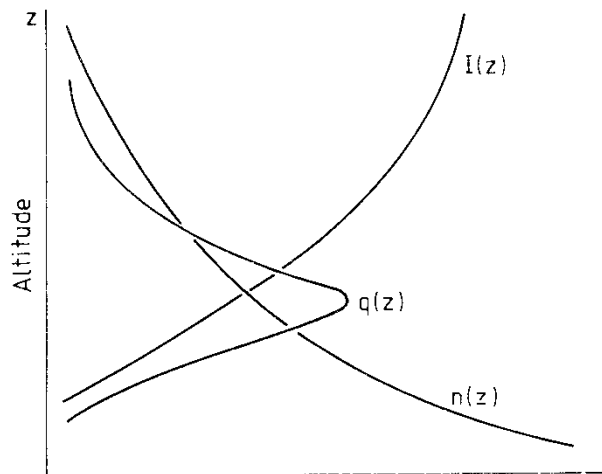
Table 5.4. Typical reaction rates relevant for ionospheric processes as indicated

Processes	Reaction rates (m^3/sec)
Dissociative recombination	
$\text{NO}^+ + e \rightarrow \text{N} + \text{O}$	$\alpha_1 = 2.1 \times 10^{-13} (T_e/300)^{-0.85}$
$\text{O}_2^+ + e \rightarrow \text{O} + \text{O}$	$\alpha_2 = 1.9 \times 10^{-13} (T_e/300)^{-0.5}$
$\text{N}_2^+ + e \rightarrow \text{N} + \text{N}$	$\alpha_3 = 1.8 \times 10^{-13} (T_e/300)^{-0.39}$
Rearrangement	
$\text{O}^- + \text{N}_2 \rightarrow \text{NO}^+ + \text{N}$	$k_1 = 2 \times 10^{-18}$
$\text{O}^+ + \text{O}_2 \rightarrow \text{O}_2^+ + \text{O}$	$k_2 = 2 \times 10^{-17} (T_r/300)^{-0.4}$
$\text{O}_2^+ + \text{NO} \rightarrow \text{NO}^+ + \text{O}_2$	$k_3 = 4.4 \times 10^{-16}$
$\text{O}_2^+ + \text{N}_2 \rightarrow \text{NO}^+ + \text{NO}$	$k_4 = 5 \times 10^{-22}$
$\text{N}_2^+ + \text{O} \rightarrow \text{NO}^+ + \text{NO}$	$k_5 = 1.4 \times 10^{-16} (T_r/300)^{-0.44}$
$\text{N}_2^+ + \text{O}_2 \rightarrow \text{N}_2 + \text{O}_2^+$	$k_6 = 5 \times 10^{-17} (T_r/300)^{-0.8}$
Radiative recombination	
$\text{O}^- + e \rightarrow \text{O} + h\nu$	$\alpha_r = 7.8 \times 10^{-14} (T_e/300)^{-0.5}$
Electron attachment	
$\text{O}_2 + M + e \rightarrow \text{O}_2^- + M$	$\beta_1 = 10^{-43} (300/T_e) e^{-600/T_e} [\text{N}_2]$
	$\beta_2 = 10^{-41} [\text{O}_2]$
$\text{O} + e \rightarrow \text{O}^- + h\nu$	$\beta_r = 10^{-21}$
Electron detachment	
$\text{O}_2^- + h\nu \rightarrow \text{O}_2 + e$	$\rho = 0.33 \text{ sec}^{-1}$
$\text{O}_2^- + \text{N}_2 \rightarrow \text{O}_2 + \text{N}_2 + e$	$\gamma_1 = 2 \times 10^{-18} (T_r/300)^{1.5} \exp(5 \times 10^{-3}/T_r)$
$\text{O}_2 + \text{O}_2 \rightarrow \text{O}_2 + \text{O}_2 + e$	$\gamma_2 = 3 \times 10^{-16} (T_r/300)^{0.5} \exp(6 \times 10^{-3}/T_r)$
$\text{O}_2 + \text{O} \rightarrow \text{O}_3 + e$	$\gamma_3 = 5 \times 10^{-16}$
$\text{O}^- + \text{O} \rightarrow \text{O}_2 + e$	$\gamma_4 = 3 \times 10^{-16}$
Ion-ion recombination	
$\text{O}_2^+ + \text{O}_2^- \rightarrow \text{O}_2 + \text{O}_2$	$\alpha_{i1} = 2 \times 10^{-13}$
$\text{O}_2^+ + \text{O}_2^- + M \rightarrow \text{O}_2 + \text{O}_2 + M$	$\alpha_{i2} = 3 \times 10^{-31} (T/300)^{-2.5} [M]$





Ionosphere & photoionization: Chapman model

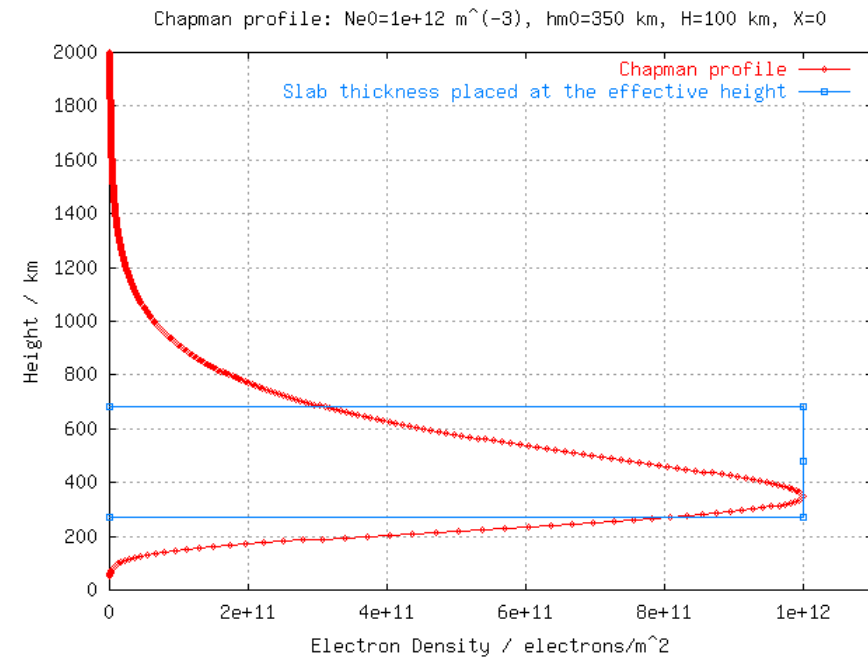
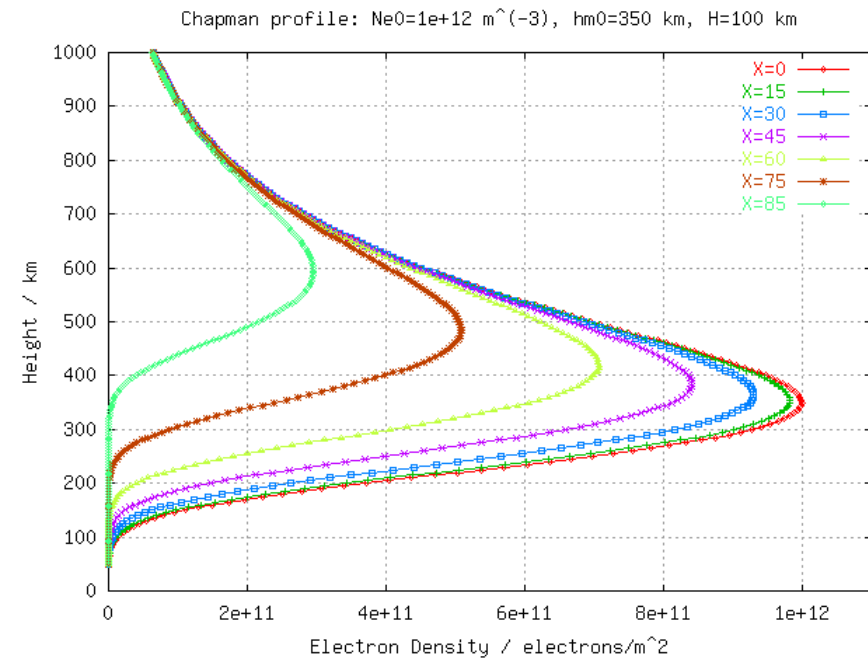


Neutral density, radiation intensity and ionization rate in arbitrary scales

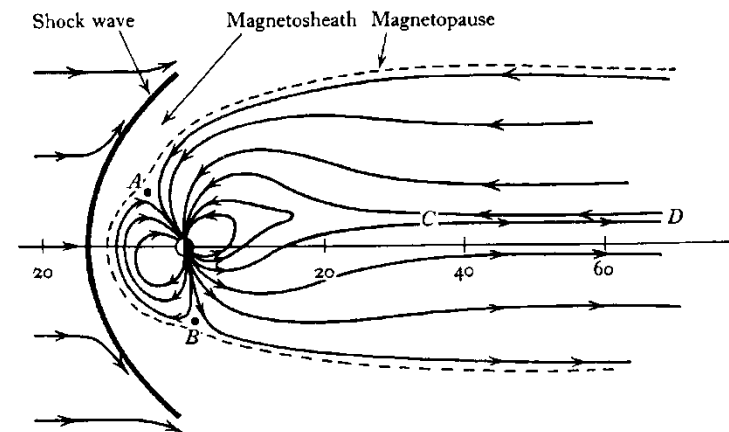
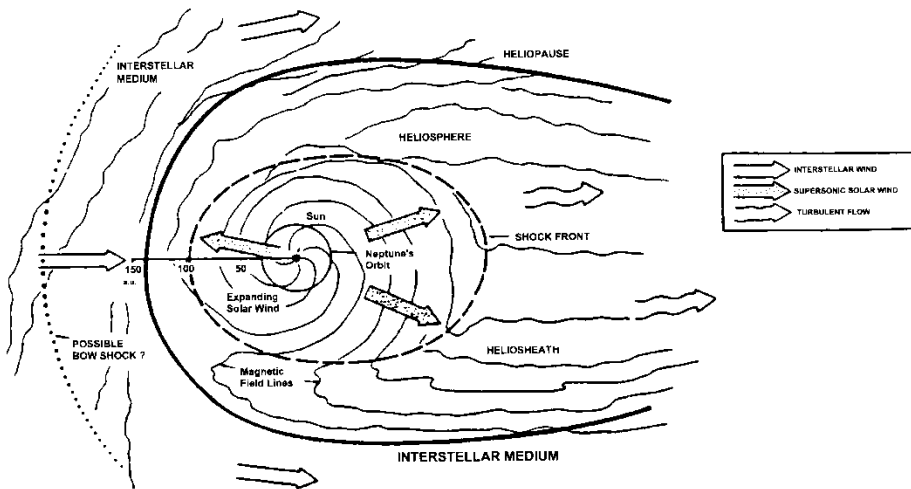
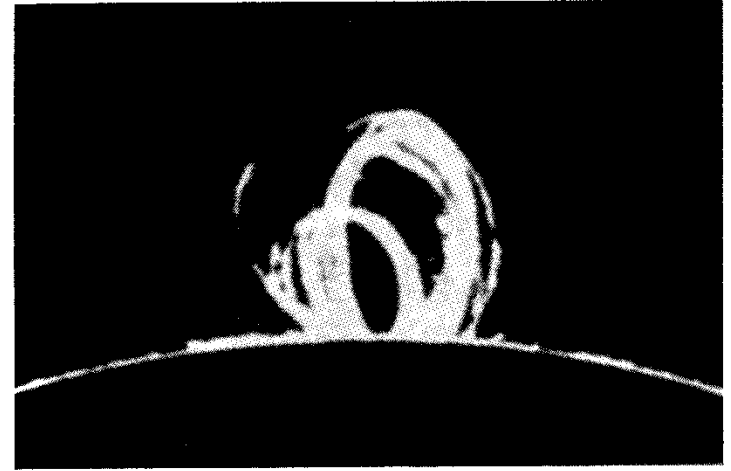
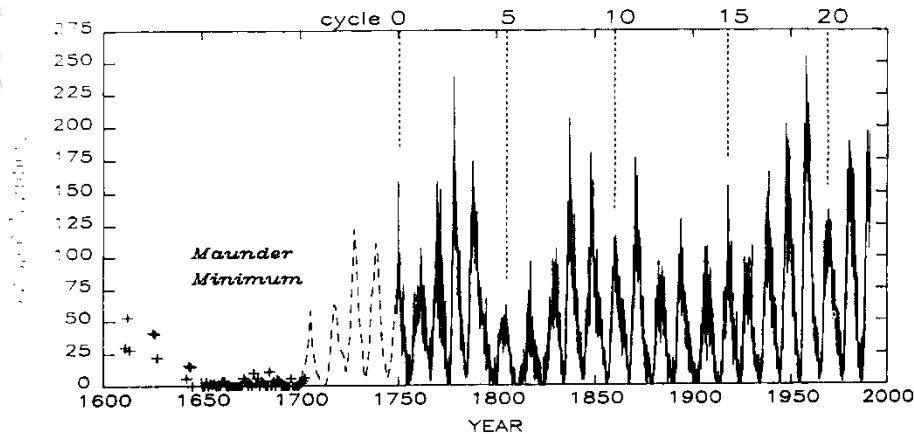
Under hydrostatic equilibrium (with pressure height scale smaller than the temperature one), assuming an ideal gas, a monochromatic beam acting on the lower part of the atmosphere, and considering photochemical equilibrium, it can be demonstrated that:

$$N_e(h, \chi) = N_{e,0} e^{\frac{1}{2} \left(1 - \frac{h - h_{m,0}}{H} - \sec(\chi) e^{-\frac{h - h_{m,0}}{H}} \right)}$$

Being $N_{e,0}$ and $h_{m,0}$ the electron density peak and corresponding height, and being H the scale height.



The Solar Cycle, the Solar Wind, the Geomagnetic Storms



Physical background: Maxwell equations, Lorentz and Electrostatic forces

$$\vec{\nabla} \cdot \vec{D} = \rho$$

$$\vec{\nabla} \cdot \vec{B} = 0$$

$$\vec{\nabla} \times \vec{H} = \vec{J} + \frac{\partial \vec{D}}{\partial t}$$

$$\vec{\nabla} \times \vec{E} = - \frac{\partial \vec{B}}{\partial t}$$

Poisson's equation (Gauss law)

Divergenceless B, there are not magnetic monopoles

Ampère's law

$$\hat{n} \cdot (\vec{D}_+ - \vec{D}_-) = \sigma$$

$$\hat{n} \times (\vec{E}_+ - \vec{E}_-) = 0$$

$$\hat{n} \cdot (\vec{B}_+ - \vec{B}_-) = 0$$

$$\hat{n} \times (\vec{H}_+ - \vec{H}_-) = \hat{K}$$

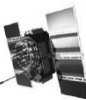
Being $D=\epsilon E$ in and $B=\mu H$ in normal and homogeneous media, representing E and B to the electric and magnetic fields.

Being σ the surface charge density and K the surface density of current

$$m \ddot{\vec{r}} = q \vec{E} + q \vec{v} \times \vec{B} - \gamma m \dot{\vec{r}}$$

Electrostatic, Lorentz and collision forces

Being m , q and v the mass, charge and velocity of the particle (i.e. electrons)



Equatorial anomalies and E cross B drift

Assuming only electric and magnetic fields acting on the charged particles, the component of $\vec{E}$ along $\vec{B}$ will vanish (it is the only force acting in this direction, generating a polarized field which suppress the external one).

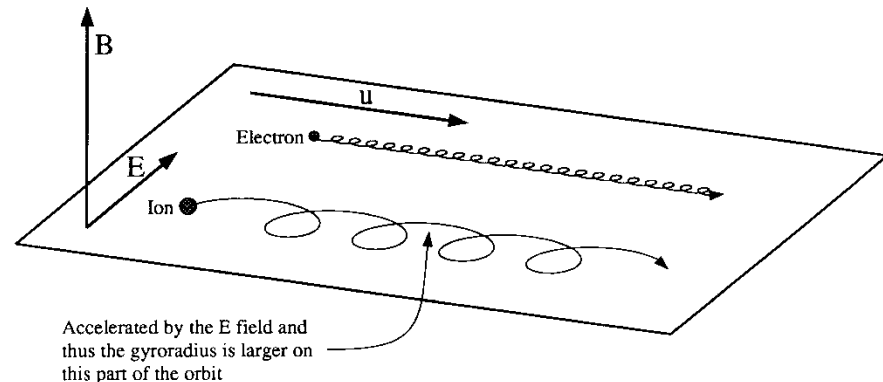
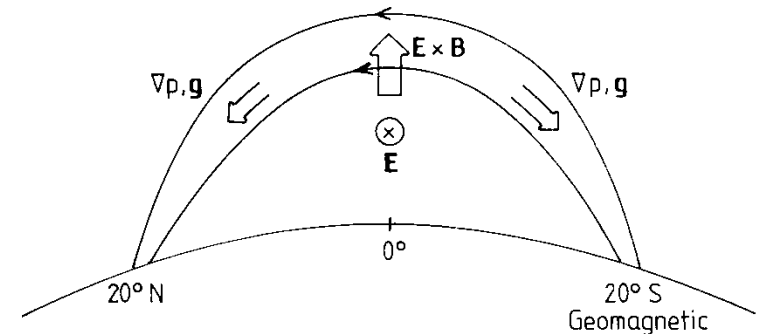
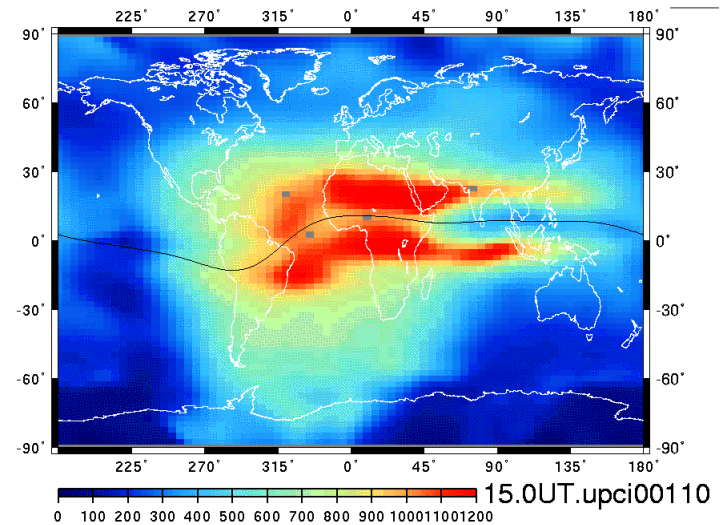
$$\underbrace{\left\langle m \frac{d\vec{v}}{dt} \right\rangle}_{\times \vec{B}} = q \vec{E} + q \vec{u}_E \times \vec{B} = \vec{0}$$

$$\vec{0} = q \vec{E} \times \vec{B} + q \left[\underbrace{(\vec{u}_E \cdot \vec{B})}_{0} \vec{B} - B^2 \vec{u}_E \right]$$

$$\vec{u}_E = \frac{1}{B^2} \vec{E} \times \vec{B}$$

Being u_E the displacement velocity generated by the E cross B drift. For a general force $\vec{F}$:

$$\vec{u}_F = \frac{1}{qB^2} \vec{F} \times \vec{B}$$

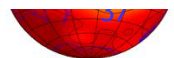
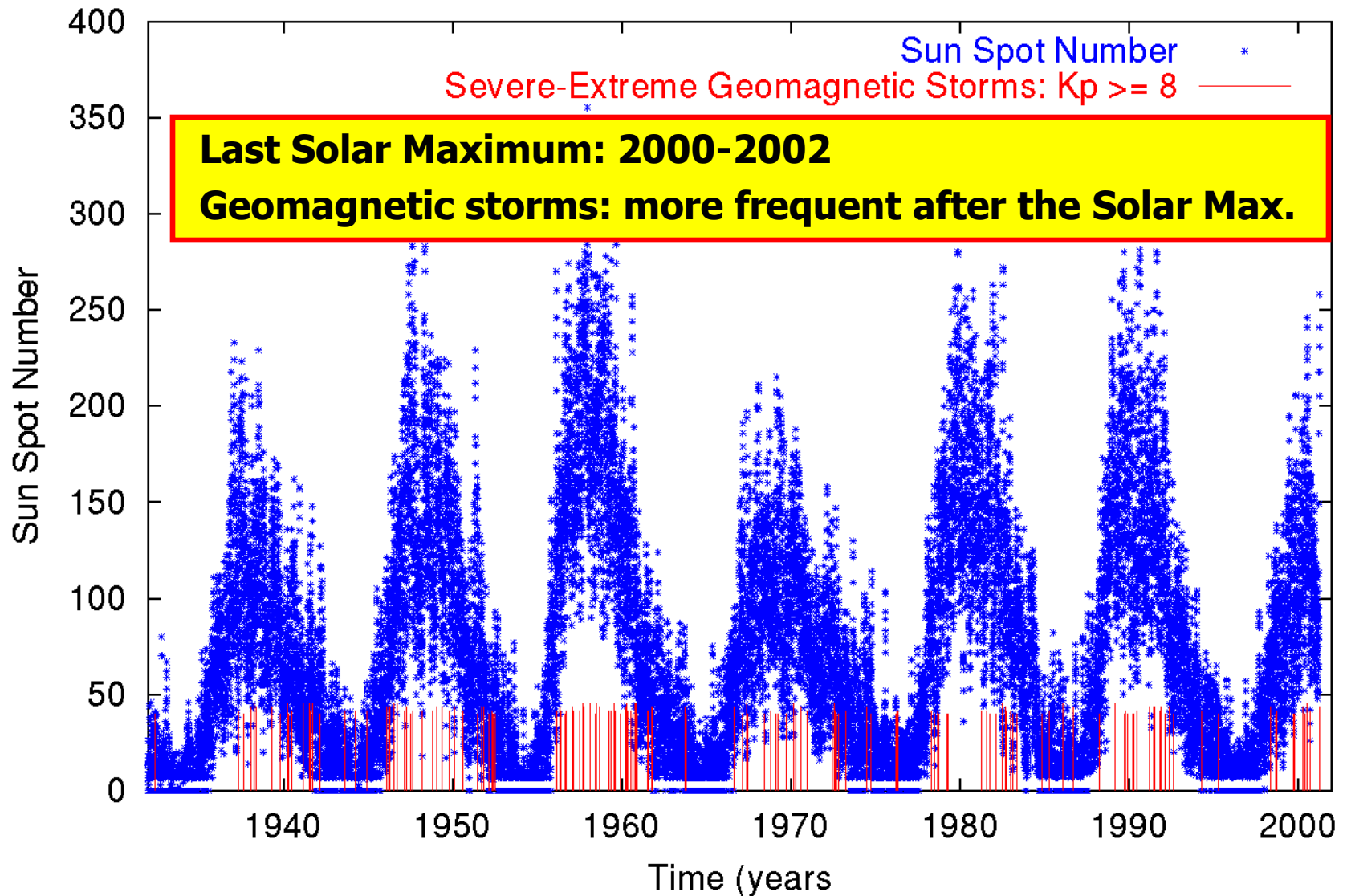




• Ionospheric effects on GPS

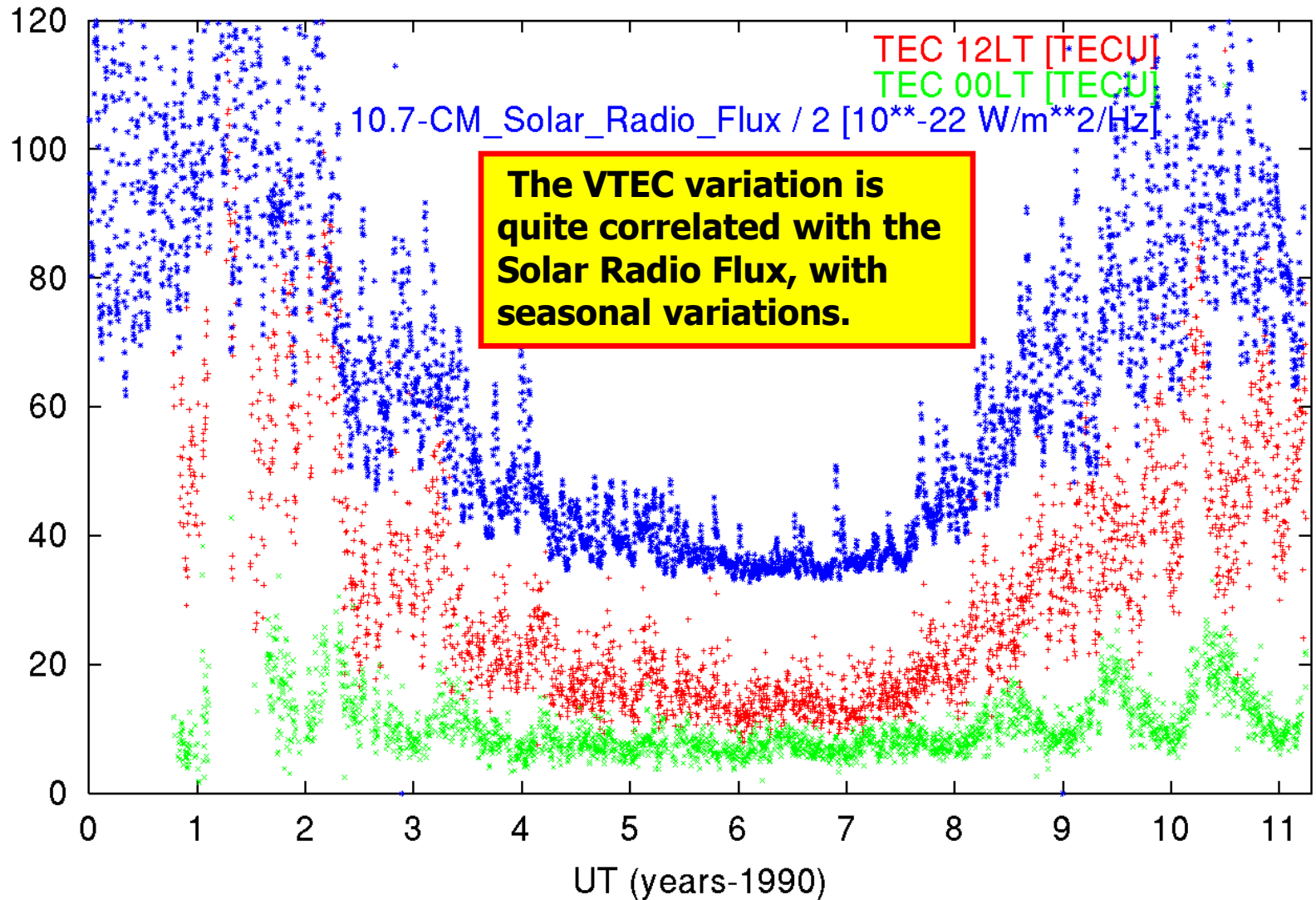
11-years Solar cycle

Sun Spot Number and Severe-Extreme Geomagnetic Storms: 1932-2002



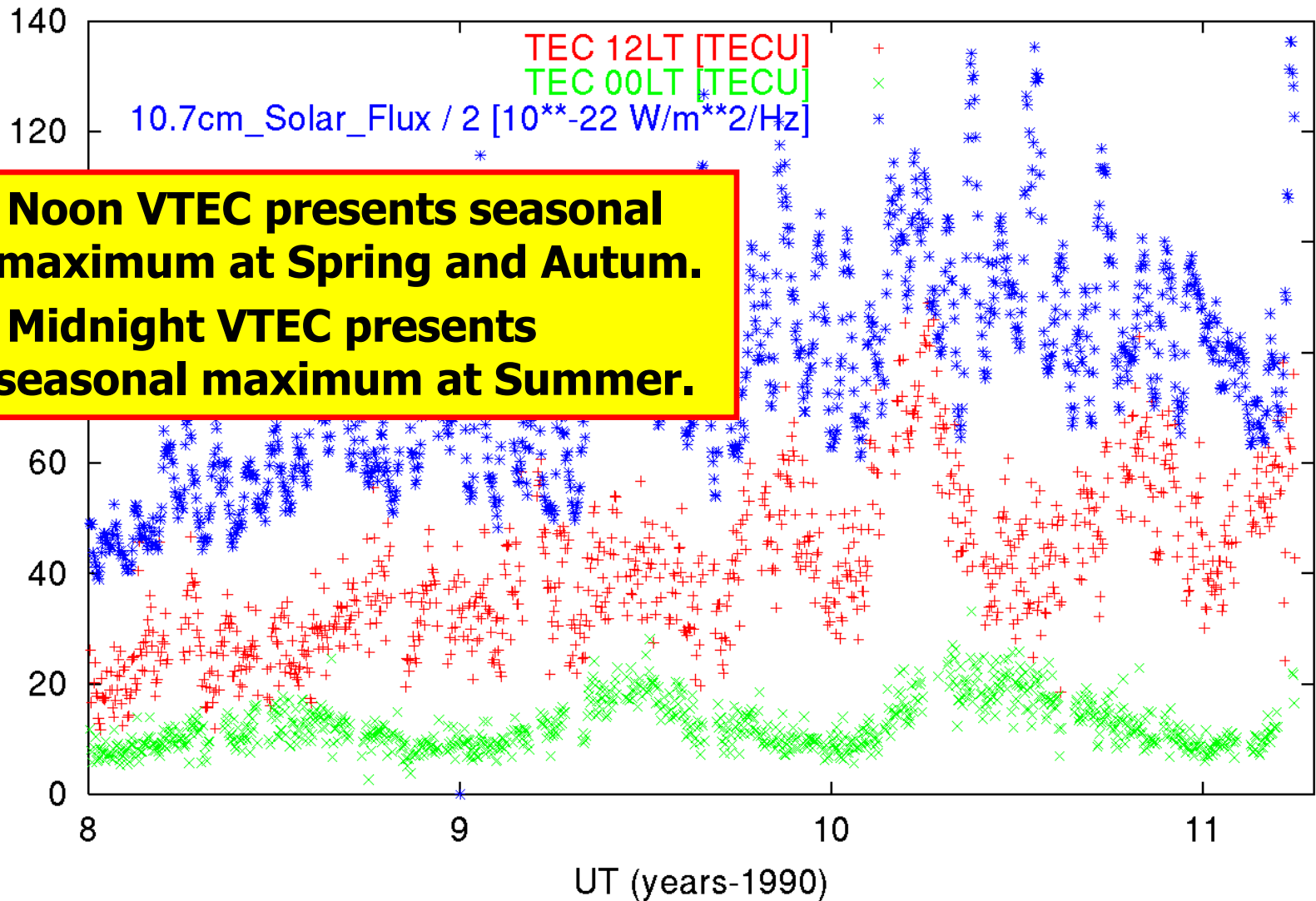
VTEC: Modulated by the Solar Cycle

Solar Flux and TEC at IGS GPS station JPLM (242,34) during the last 11 years



VTEC modulated by the Season

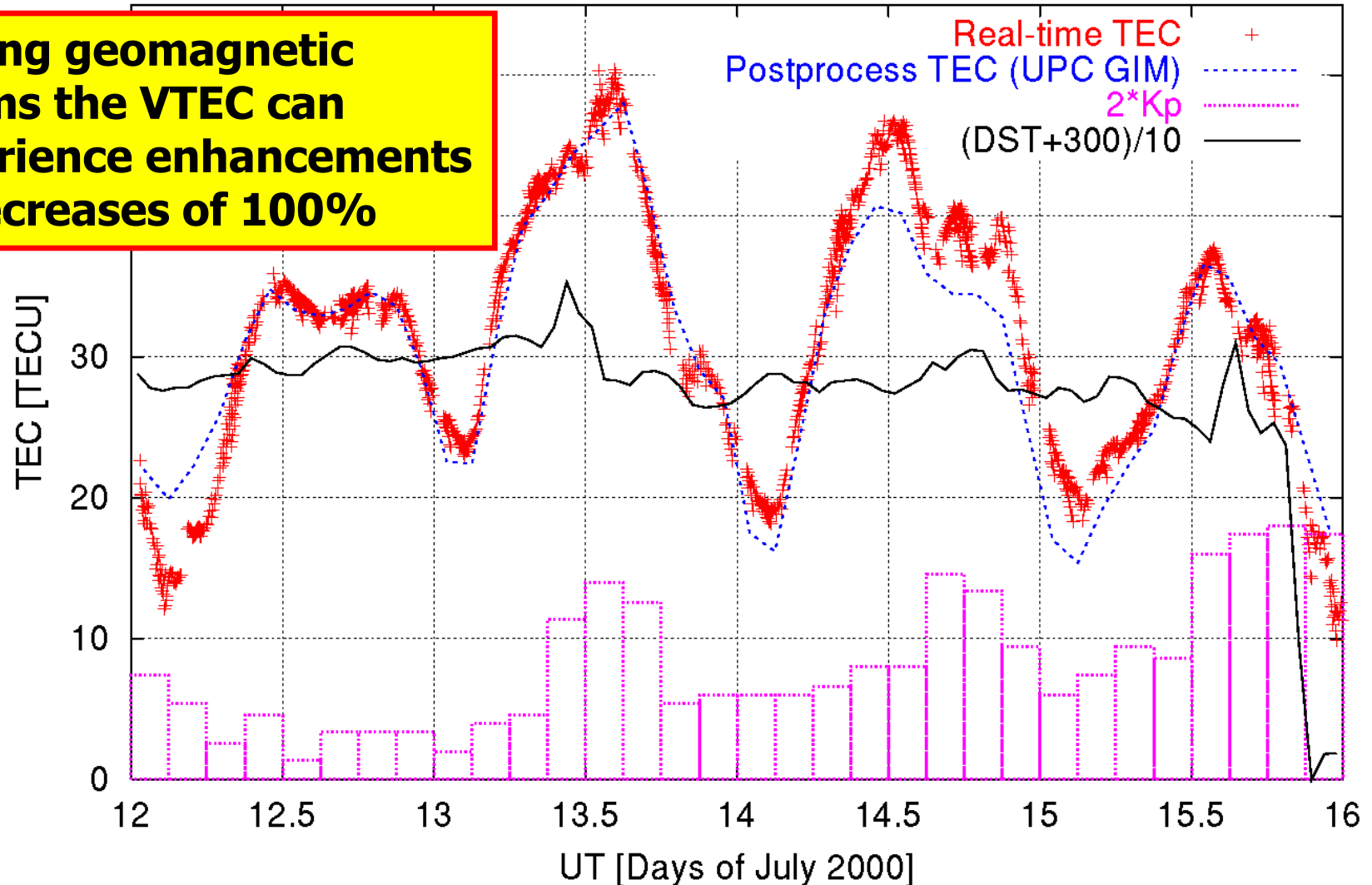
TEC at IGS GPS station JPLM (242,34) and Solar_Flux



VTEC can change during geomagnetic storms

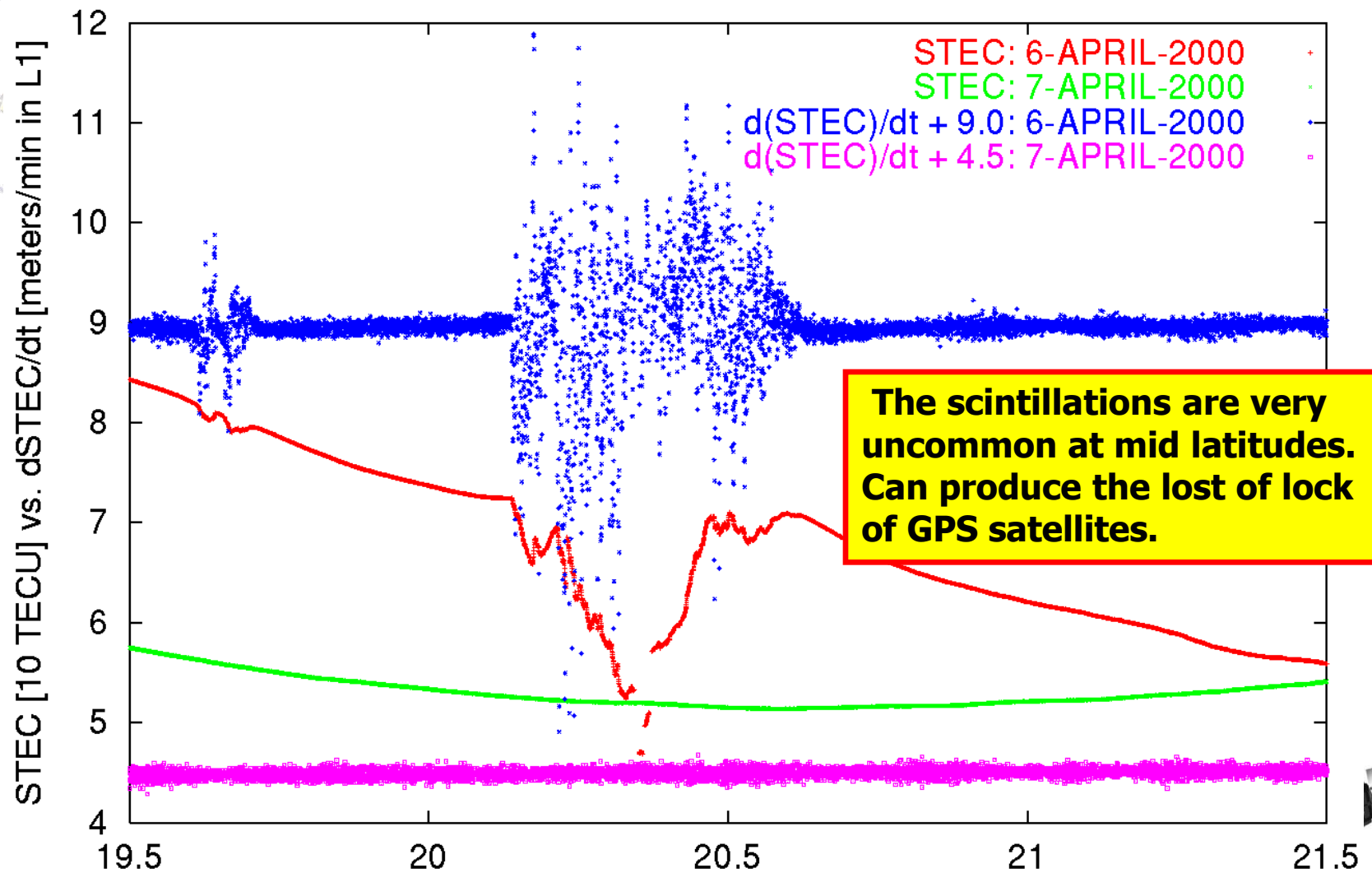
Real-time Total Electron Content: Brussels (Lon=4, Lat=51 deg.)

During geomagnetic storms the VTEC can experience enhancements or decreases of 100%



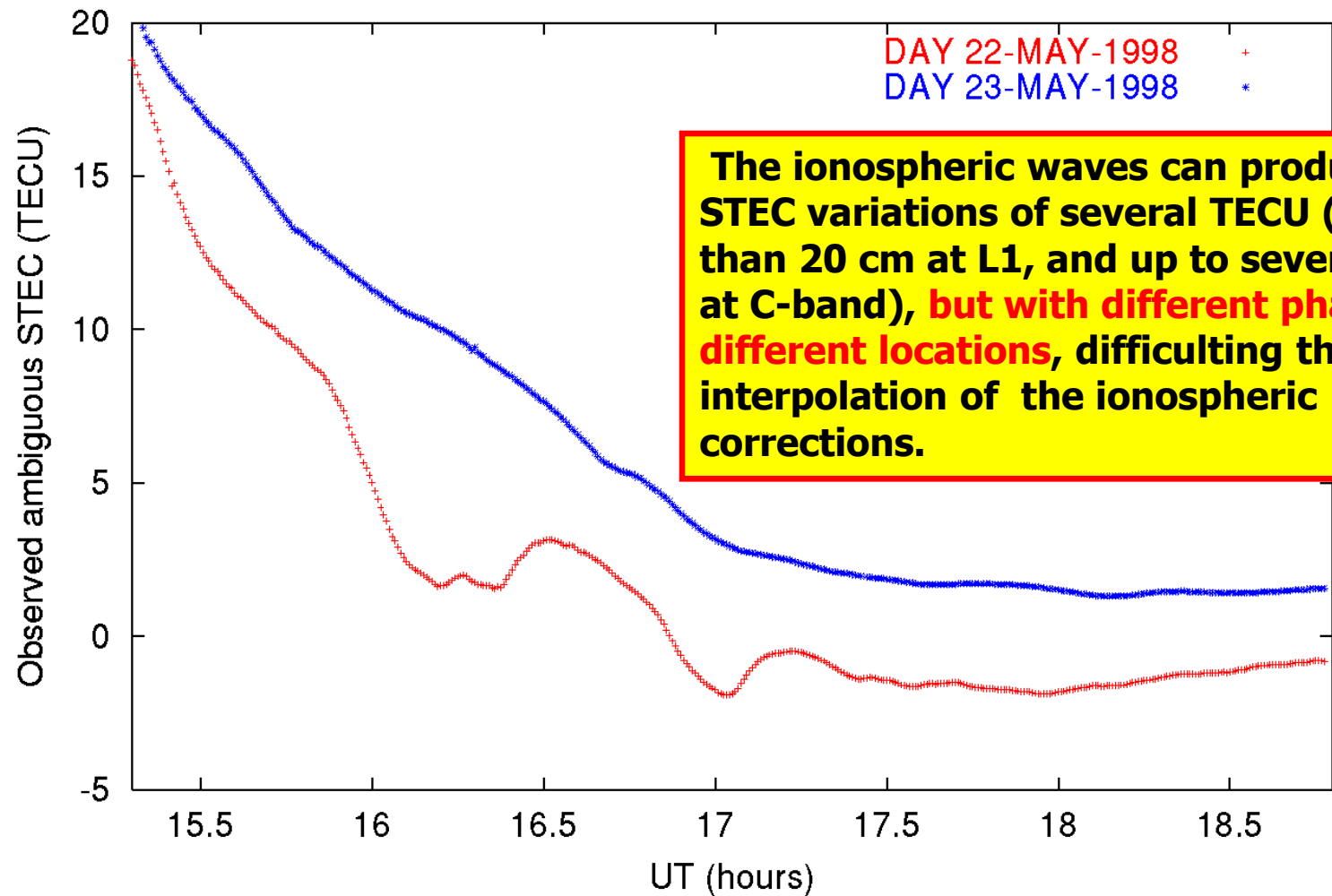
Scintillation: Very rapid VTEC variations

Observed ambiguous STEC for PRN01 (UPC, Barcelona, Spain)



STEC can be affected by Travelling Ionospheric Disturbances

PRN01 from USUD: Lon=138 deg, Lat=36 deg (LT=UT+9.22 hours)



The ionospheric waves can produce STEC variations of several TECU (more than 20 cm at L1, and up to several cm at C-band), but with different phases at different locations, diffculting the interpolation of the ionospheric corrections.



- From Ionosphere First principles to GPS preprocessing

GPS signal propagation: Appleton Formula

$$m \ddot{\vec{r}} = q \vec{E} + q \vec{v} \times \vec{B} - \gamma m \dot{\vec{r}}$$

$$\begin{cases} m \ddot{x} = qE_x - qB_T \dot{z} - \gamma m \dot{x} \\ m \ddot{y} = qE_y + qB_L \dot{z} - \gamma m \dot{y} \\ m \ddot{z} = qE_z + q(B_T \dot{x} - B_L \dot{y}) - \gamma m \dot{z} \end{cases}$$

$$m \ddot{\vec{r}} = e \vec{E} + (e \vec{r} \times \vec{B}) - m \nu \dot{\vec{r}} \quad (3.1)$$

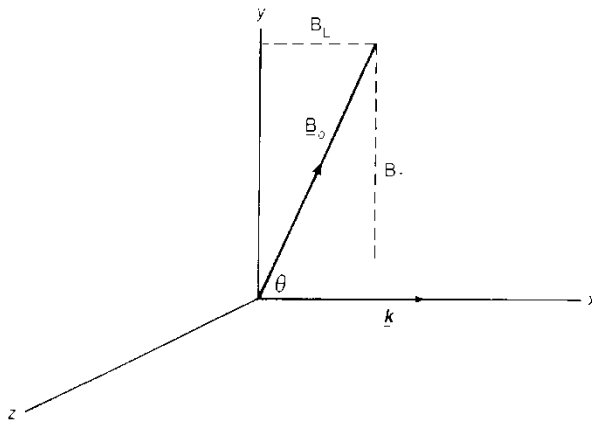


Fig. 3.1 System of orthogonal axes x , y , and z . The x -direction is the direction of phase propagation, and the external magnetic field lies in the x y plane. The y z plane is the plane of the wavefront

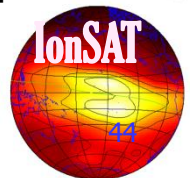
Multiplying throughout by the charge density N_e , replacing first and second x derivative by $i\omega x$ and $-\omega^2 x, \dots$, using $P_x = Nq x, \dots$, the components of the volume polarization, taking into account the wave polarization relationships ($R = Dz/Dy = -Hy/Hz = Ez/Ey = Pz/Py = z/y$), we can get the **Appleton formula**:

$$n^2 = (\mu - i\chi)^2 = 1 + \frac{P_y}{\epsilon_0 E_y} = 1 - \frac{X}{1 - iZ + iY_L R} = 1 - \frac{X}{1 - iZ - \frac{Y_T^2}{2(1 - X - iZ)} \pm \sqrt{\frac{Y_T^4}{4(1 - X - iZ)} + Y_L^2}}$$

$$X = \frac{\omega_p^2}{\omega^2}, Y_L = -\frac{\omega_g}{\omega} \cos \theta, Y_T = -\frac{\omega_g}{\omega} \sin \theta, Z = \frac{\gamma}{\omega}$$

Being the plasma and gyro frequencies:

$$\omega_p^2 = \frac{Nq^2}{m \epsilon_0}, \omega_g = \frac{Bq}{m}$$

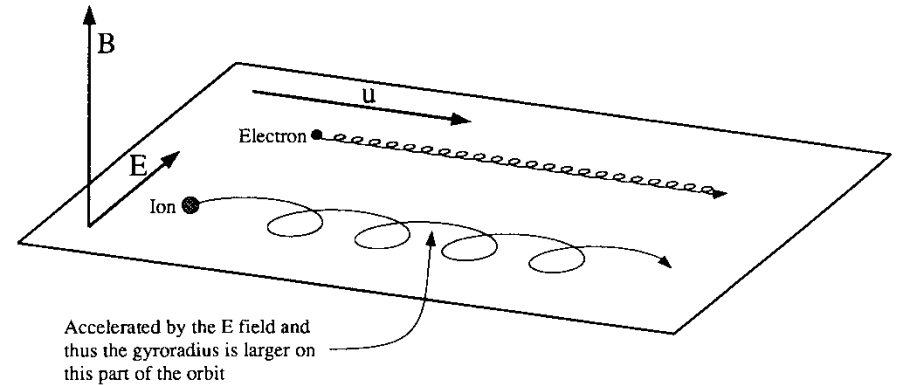


Plasma and gyro frequencies

$$m \vec{r} = q \vec{v} \times \vec{B}$$

$$m \frac{v^2}{r} = qvB_{\perp}$$

$$\omega_g = \frac{qB_{\perp}}{m}$$

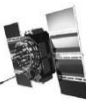


$$\nabla \cdot \vec{E} = \frac{\rho}{\epsilon_0} \Rightarrow \frac{\partial E}{\partial x} = \frac{\rho}{\epsilon_0} \Rightarrow E \approx - \frac{N_e |q_e|}{\epsilon_0} \delta$$

$$\frac{F}{A} = \left(- \frac{N_e |q_e|}{\epsilon_0} \delta \right) (N_e |q_e| L) = (LN_e m_e) \ddot{\delta}$$

$$\ddot{\delta} = - \frac{N_e q_e^2}{m_e \epsilon_0} \delta \Rightarrow \omega_p^2 = \frac{N_e q_e^2}{m_e \epsilon_0}$$

From the Appleton formula, with the first order approximation and neglecting collisions, the plasma frequency produces no propagation ($n=0$, $X=1$, $\omega=\omega_p$, reflexion in vertical sounding for instance).



GPS signal propagation and ionospheric terms (first and second order)

$$n_p^2 = 1 - \frac{X}{1 - iZ - \frac{Y_T^2}{2(1 - X - iZ)}} \pm \sqrt{\frac{Y_T^4}{4(1 - X - iZ)^2} + Y_L^2} \approx$$

$$\approx \{Z = 0 \text{ \& high freq.}\} \approx 1 - \frac{X}{1 \pm Y_L} \Rightarrow n \approx 1 - \frac{1}{2} X (1 \mp Y_L)$$

$$\delta L_I = \int_{\vec{r}_{rec}}^{\vec{r}_{tra}} \left(\frac{c}{v_p} - 1 \right) ds = \int_{\vec{r}_{rec}}^{\vec{r}_{tra}} (n_p - 1) ds = - \int_{\vec{r}_{rec}}^{\vec{r}_{tra}} \frac{1}{2} \frac{\omega_p^2}{\omega^2} \left\{ 1 \pm \underbrace{\frac{\omega_g}{\omega} \cos \theta}_{(GPS) \approx \frac{10^6}{1.6 \cdot 10^9} \leq 6 \cdot 10^{-4}} \right\} ds =$$

$$= \{ordinary - wave\} = \left\{ K_1 = \frac{q^2}{8 \pi^2 m \varepsilon_0} \approx 40.3 (MKS) \right\} =$$

$$- \frac{K_1}{f^2} \left[\underbrace{\int_{\vec{r}_{rec}}^{\vec{r}_{tra}} N_e ds}_{STECH} + \frac{q}{2 \pi m} \frac{1}{f} \int_{\vec{r}_{rec}}^{\vec{r}_{tra}} N_e B \cos \theta ds \right] = \underbrace{- \frac{K_1}{f^2} STECH}_{FIRST \text{ ORDER}} - \underbrace{\frac{K_2}{f^3} \int_{\vec{r}_{rec}}^{\vec{r}_{tra}} N_e B \cos \theta ds}_{SECOND \text{ ORDER}}$$

Being $K_1=40.3$ and $K_2=1.13 \times 10^{12}$ in MKS system units.



GPS ionospheric terms: group delay

$$\begin{aligned}\psi_1 &= Ae^{i(\omega t - kx)} \\ \psi_2 &= Ae^{i([\omega + \delta\omega]t - [k + \delta k]x)} = Ae^{i(\delta\omega \cdot t - \delta k \cdot x)} e^{i(\omega t - kx)} \\ \psi_1 + \psi_2 &= A \cos\left(\frac{\delta\omega}{2}t - \frac{\delta k}{2}x\right) e^{i(\delta\omega \cdot t - \delta k \cdot x)/2} e^{i(\omega t - kx)}\end{aligned}$$

Modulated .signal

$$v_g = \frac{\delta\omega}{\delta k} \approx \frac{d\omega}{dk} = \left\{ v_p = \frac{\omega}{k} \right\} = v_p + k \frac{dv_p}{dk}$$

$$n_g - n_p = \frac{c}{v_p v_g} (v_p - v_g) \approx \dots \approx f \frac{dn_p}{df} = \omega \frac{dn_p}{d\omega}$$

$$n_g - n_p = \left\{ n_p = 1 - \frac{1}{2} \frac{\omega_p^2}{\omega^2} - \frac{1}{2} \frac{\omega_p^2 \omega_g}{\omega^3} \right\} = \frac{\omega_p^2}{\omega^2} + \frac{3}{2} \frac{\omega_p^2 \omega_g}{\omega^3}$$

$$n_g = 1 + \frac{1}{2} \frac{\omega_p^2}{\omega^2} + \frac{\omega_p^2 \omega_g}{\omega^3}$$

$$\begin{aligned}\delta\rho_L &= \underbrace{-\frac{K_1}{f^2} STEC}_{\text{FIRST .ORDER}} - \underbrace{\frac{K_2}{f^3} \int_{\vec{r}_{rec}}^{\vec{r}_{tr}} N_e B \cos \theta . ds}_{\text{SECOND .ORDER}} \\ \delta\rho_p &= \underbrace{+\frac{K_1}{f^2} STEC}_{\text{FIRST .ORDER}} + \underbrace{2 \frac{K_2}{f^3} F}_{\text{SECOND .ORDER}}\end{aligned}$$

Being $K_1=40.3$ and $K_2=1.13 \times 10^{12}$ in MKS system units.

Ionospheric combination of GPS obs.

- The ionospheric (geometry-free) combination of carrier phases and pseudoranges **provides the STEC** (in first order, <0.1% error), **but affected by biases** (ambiguities for LI -can be large- and instrumental delays for PI -typically of few meters-).
- The **pseudoranges (PI) can be very affected by multipath and noise measurement (1 meter or more).**
- Thus the ionospheric sounding with GPS should be based on carrier phase measurements, dealing with the carrier phase ambiguity estimation.

$$\left. \begin{aligned} L &= \rho^* + \lambda \cdot b + \lambda \cdot w + pc_L - \frac{K_1}{f^2} STEC - \frac{K_2}{f^3} F \\ P &= \rho^* + D + pc_P + \frac{K_1}{f^2} STEC + \frac{2K_2}{f^3} F \end{aligned} \right\} \begin{aligned} STEC &= \int_{\vec{r}_{rec}}^{\vec{r}_{tra}} N_e ds \\ F &= \int_{\vec{r}_{rec}}^{\vec{r}_{tra}} N_e B \cos \theta \cdot ds \end{aligned}$$

$$L_I = L_1 - L_2 = \left\{ \begin{array}{l} 1st - order \\ error < 0.1\% \end{array} \right\} = \underbrace{K_1 \frac{(f_1^2 - f_2^2)}{f_1^2 f_2^2}}_{\alpha \approx 1.05 \cdot 10^{-17} \text{ m/e- / m}^2} STEC + \underbrace{\lambda_1 b_1 - \lambda_2 b_2}_{BI} + W_I + pc_I$$

$$P_I = P_2 - P_1 = \dots = \underbrace{K_1 \frac{(f_1^2 - f_2^2)}{f_1^2 f_2^2}}_{\alpha \approx 1.05 \cdot 10^{-17} \text{ m/e- / m}^2} STEC + D_I + pc_I$$

$$L_I - P_I = B_I + W_I - D_I$$

Being $K_1=40.3$ and $K_2=1.13 \times 10^{12}$ in MKS system units, ρ^* the non-dispersive terms (geometric distance, clocks errors, tropospheric delay,...), λb the ambiguity and D the instrumental delay.

Ionospheric-free combinations

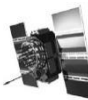
$$\left. \begin{aligned} L &= \rho^* + \lambda \cdot b + \lambda \cdot w + pc_L - \frac{K_1}{f^2} STEC - \frac{K_2}{f^3} F \\ P &= \rho^* + D + pc_P + \frac{K_1}{f^2} STEC + \frac{2K_2}{f^3} F \end{aligned} \right\} \begin{aligned} STEC &= \int_{\vec{r}_{rec}}^{\vec{r}_{tra}} N_e ds \\ F &= \int_{\vec{r}_{rec}}^{\vec{r}_{tra}} N_e B \cos \theta \cdot ds \end{aligned}$$

$$L_c = \frac{f_1^2 L_1 - f_2^2 L_2}{f_1^2 - f_2^2} = \rho^* + \lambda_n \left(\frac{f_1 b_1 - f_2 b_2}{f_1 - f_2} + \frac{f_1 w_1 - f_2 w_2}{f_1 - f_2} \right) + pc_c + K_2 \underbrace{\frac{STEC \cdot B \cdot \cos \theta}{f_1 f_2 (f_1 + f_2)}}_{I_2}$$

$$P_c = \frac{f_1^2 P_1 - f_2^2 P_2}{f_1^2 - f_2^2} = \rho^* - 2K_2 \frac{STEC \cdot B \cdot \cos \theta}{f_1 f_2 (f_1 + f_2)} + pc_c$$

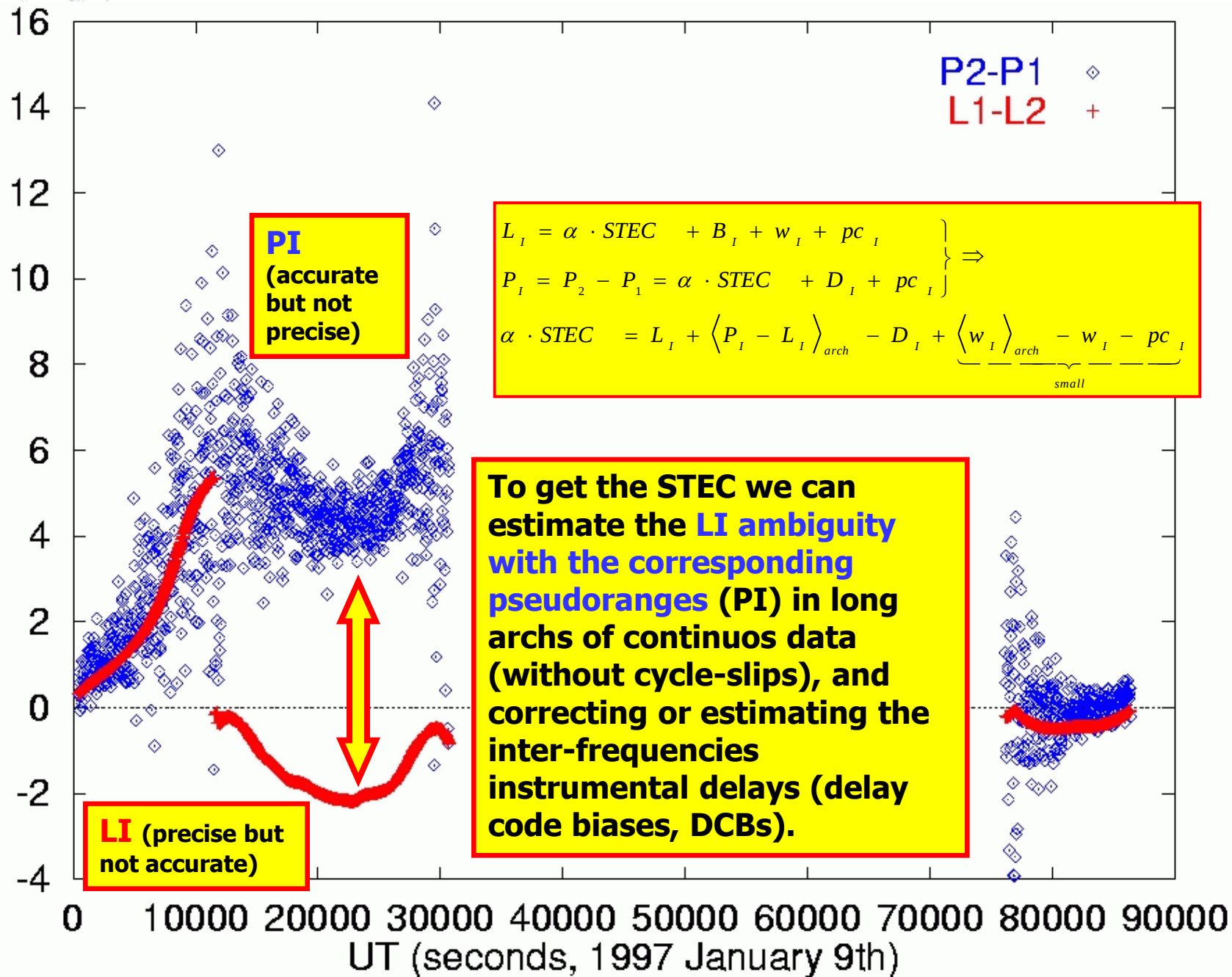
$$L_c - P_c = \lambda_n \left(\frac{f_1 b_1 - f_2 b_2}{f_1 - f_2} + \frac{f_1 w_1 - f_2 w_2}{f_1 - f_2} \right) + 3K_2 \frac{STEC \cdot B \cdot \cos \theta}{f_1 f_2 (f_1 + f_2)}$$

Being $K_1=40.3$ and $K_2=1.13 \times 10^{12}$ in MKS system units, ρ^* the non-dispersive terms (geometric distance, clocks errors, tropospheric delay,...), λb the ambiguity and D the instrumental delay.

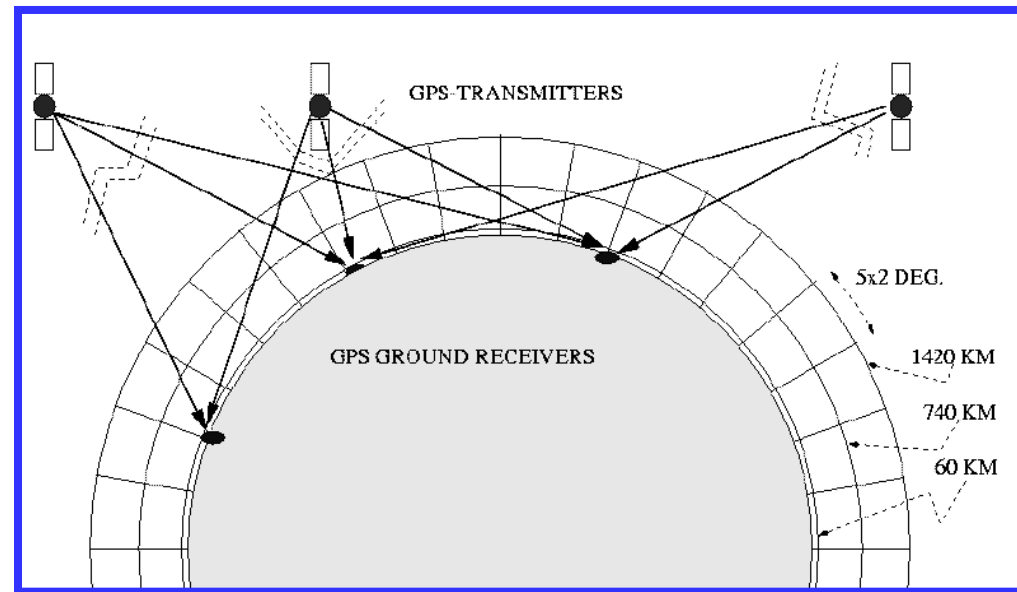


Extracting the STEC from iono. Li (1)

Ionospheric combination (meters, PRN01)



Extracting the STEC from iono. Li (2)



• **Another way to estimate the LI ambiguity is to use just carrier phase data LI, estimating (and decorrelating) it simultaneously to electron content by taking into account the geometry variation (and in the context of a Kalman filter). Not affected by PI multipath & quite convenient for real-time.**

$$L_I = STEC + B_I = \int_{REC}^{SAT} N_e dl + B_I = \sum_i \sum_j \sum_k (N_e)_{i,j,k} \Delta s_{i,j,k} + B_I$$

Extracting the STEC from iono. Li (3)

We can get the STEC variation directly (without DCBs and reducing the multipath error) in one sidereal day, (23h56m approx.) in which the geometry transmitter-receiver repeats (the GPS satellite describes 2 orbits and the fixed GPS receiver on the ground describes 1 rotation).

$$\alpha \cdot STEC = L_I + \left\langle P_I - L_I \right\rangle_{arch} - D_I + \underbrace{\left\langle w_I \right\rangle_{arch} - w_I - pc_I}_{small} \Rightarrow$$

$$\alpha \cdot \delta STEC = \delta L_I + \left\langle \delta P_I - \delta L_I \right\rangle_{\delta - arch}$$

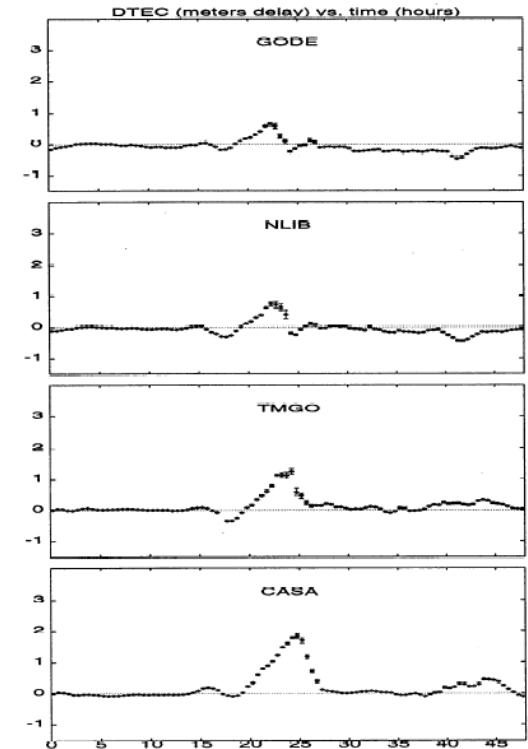


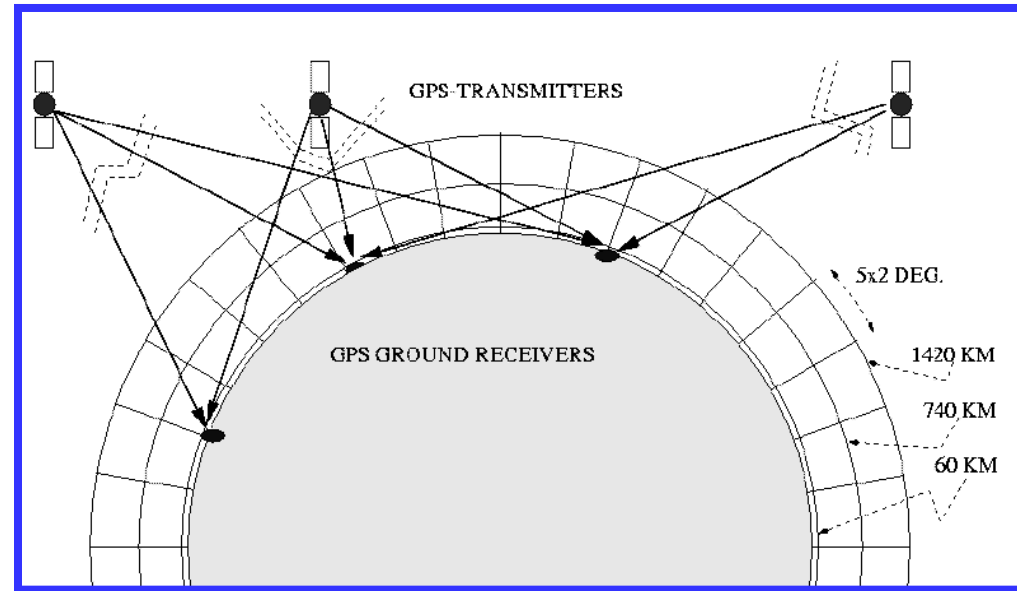
Figure 3. Mean TEC variation at 30-minute intervals are plotted for four permanent GPS stations in North America which are distributed along similar declinations (40 deg. approximately) and with longitudes separated by ≈ 10 -15 deg.: GODE at 283 deg., NLIB at 268 deg., TMGO at 255 deg. and CASA at 241 deg. The data were also collected on 17, 18 and 19 October 1995 and the variation has been calculated referred to 17 October 1995 in meters of delay. The horizontal axis is the time from 18 October 0h UT in hours.

(More details in Hernández-Pajares, M., J.M. Juan, J. Sanz, High resolution TEC monitoring method using permanent GPS receivers, *Geophysical Research Letters*, Vol.24, No.13, 1643-1646, July 1, 1997)



•Obtaining the electron content

Ground-based vs. space-based GPS data



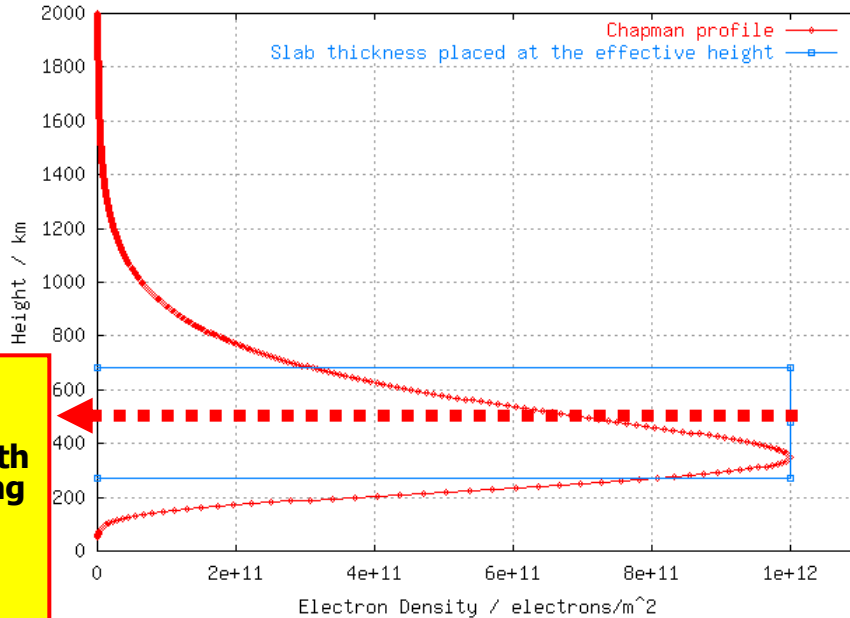
- The ground based GPS observations are mainly vertical: so they can provide VTEC estimates with high horizontal spatial (and temporal) resolution.
- The space-based GPS observations (such as those from LEOs) can provide limb observations, which are mainly horizontal. They can provide high vertical resolution.



- Using ground-based GNSS data (VTEC)

From STEC to VTEC: Single layer model

Chapman profile: $N_0=1e+12 \text{ m}^{-3}$, $h_m=350 \text{ km}$, $H=100 \text{ km}$, $X=0$

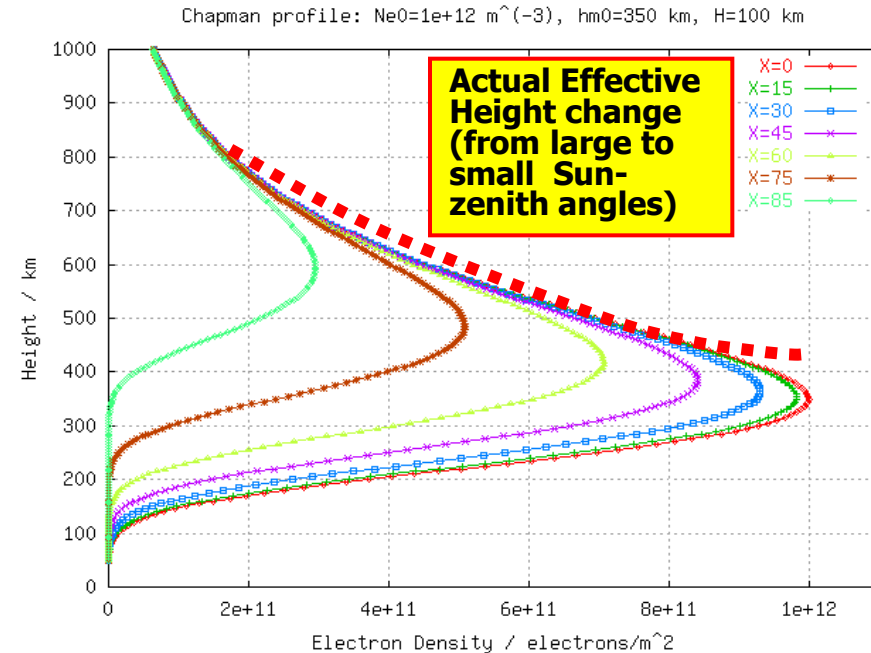


Effective Height
(Sun at zenith and assuming the slab thickness width, $VTEC/N_0$)

One simple way of obtaining the VTEC from STEC is based on the assumption of electron content concentrated at one shell at fixed *effective height* ("single layer model").

But such effective height can change in terms of Local Time (daylight/night), latitude and geomagnetic activity.

You can see at right the effective height change in terms of the Sun-zenith angle predicted by the Chapman model.



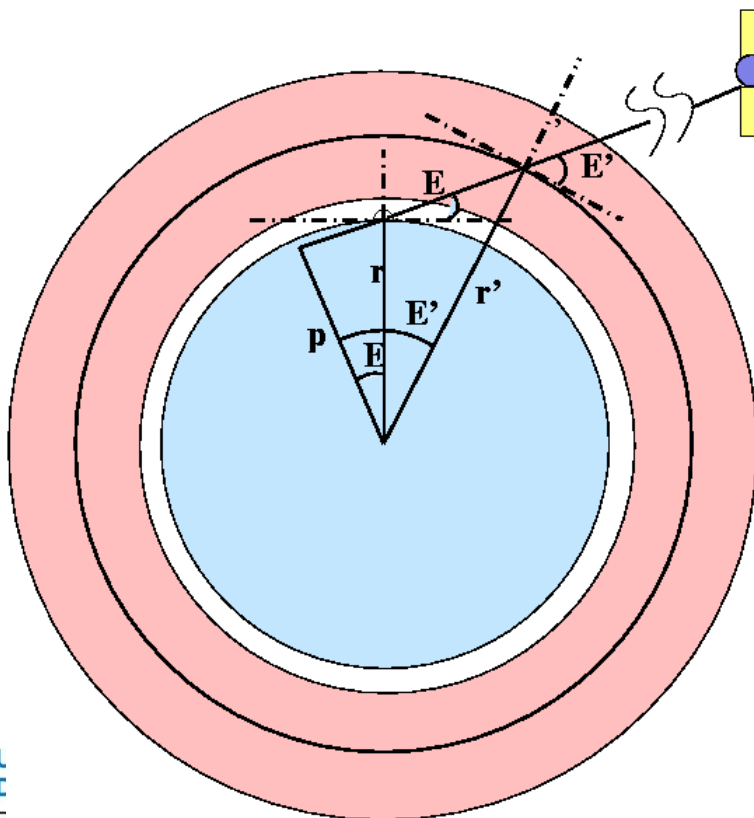
Ionosphere & photoionization: Chapman model

$$STEC = M \cdot VTEC$$

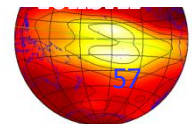
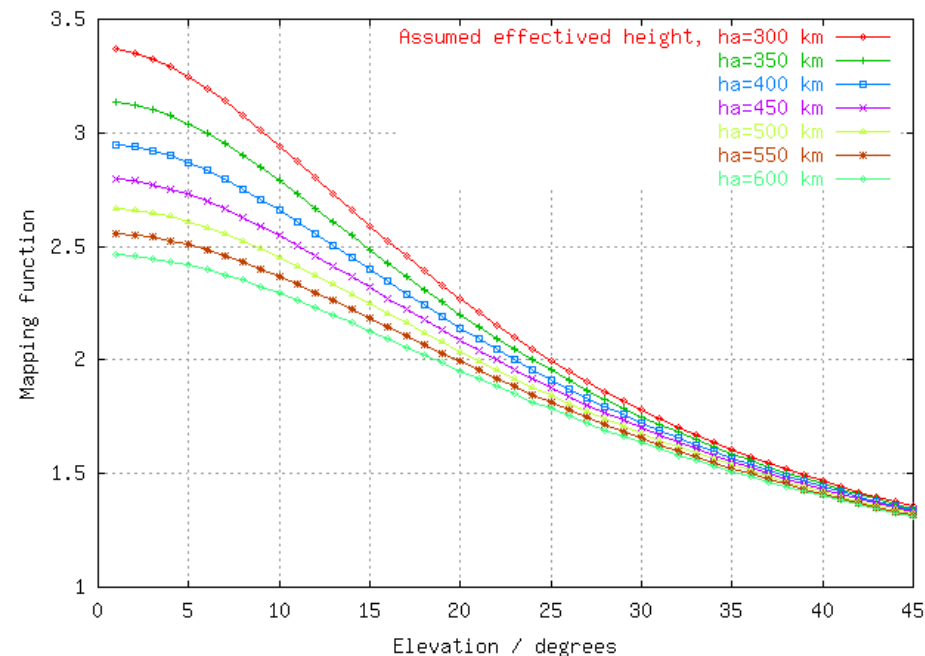
$$M = \frac{1}{\sin E'} \quad \left\{ \begin{array}{l} M = \{r' = r + h\} = \frac{1}{\sqrt{1 - \frac{r^2}{(r+h)^2} \cos^2 E}} \end{array} \right\} VTEC = STEC \cdot \sqrt{1 - \frac{r^2}{(r+h)^2} \cos^2 E} \Rightarrow$$

$$r \cos E = r' \cos E'$$

$$\Rightarrow \underbrace{STEC}_{h \rightarrow 0^+} \cdot \underbrace{\sin E}_{h \rightarrow 0^+} < VTEC < \underbrace{STEC}_{h \rightarrow +\infty}$$



Standard Iono. Mapping in terms of different assumed effective heights



Estimating the ionospheric distribution

Chapman e- Dens. [Ne0,hm0,H=1.e+12,300,100/MKS] / e-/m²

STEC GPS Sat #1

STEC GPS Sat #2

Effective Height
(he)

Assumed (ha)
Effective Height

STEC GPS Sat #3

GPS Rec.

Solar-Zenith angle / degrees

ha > he

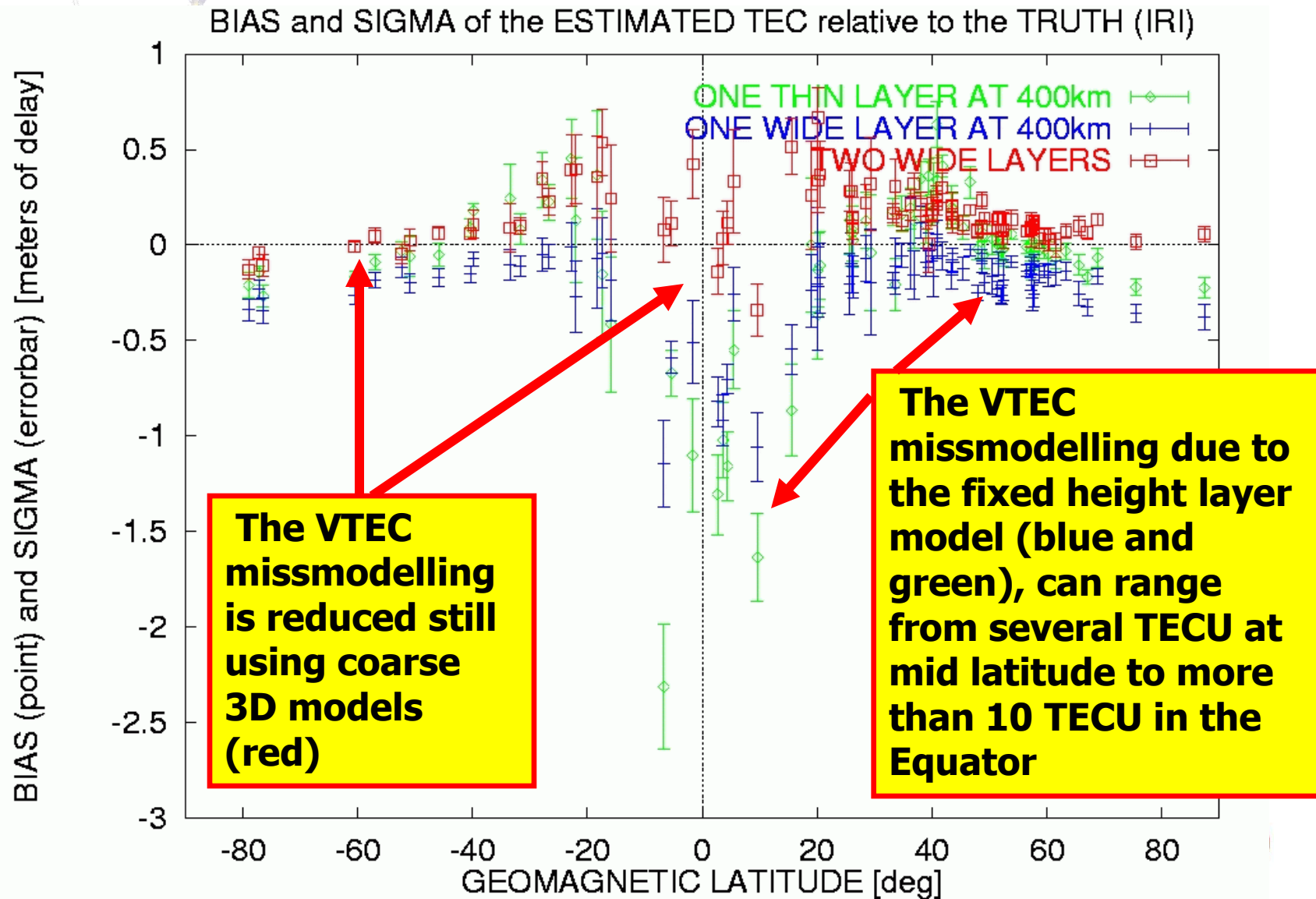
VTEC overestimated

ha < he

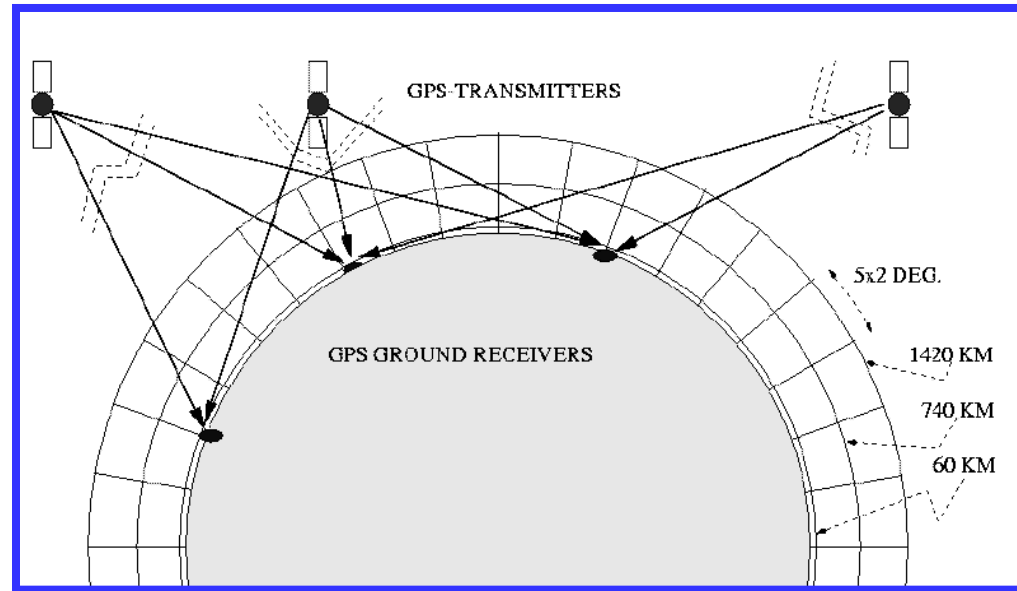
VTEC infraestimated

Height / km

Benefits of combining different kind of Iono. Obs. in a 3D Voxel Model



Real-time iono. model

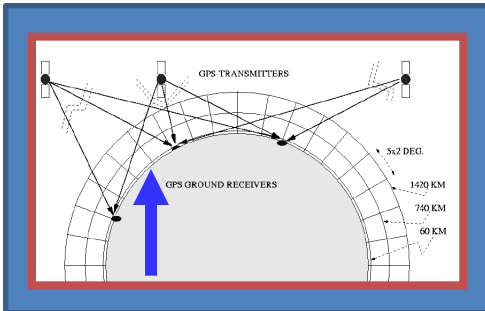


- Only carrier phase data needed
- With tomographic description: more accurate
- DCB's no longer needed
- No affected by pseudorange multipath

$$L_I = STEC + B_I = \int_{REC}^{SAT} N_e dl + B_I = \sum_i \sum_j \sum_k (N_e)_{i,j,k} \Delta s_{i,j,k} + B_I$$

Tomographic implementation

GNSS satellite orbits & clocks,
GNSS measurements



Compute the length
of each ray
intersection with each
illuminated voxel, Δs

Prediction

$$\hat{X}^-(n) = \Phi \hat{X}(n-1)$$

$$P_{\hat{X}^-(n)} = \Phi P_{\hat{X}(n-1)} \Phi^T + Q$$

$$\hat{X}(0)$$

$$P_{\hat{X}(0)}$$

Initialization

Estimation

$$\hat{X}(n) = P_{\hat{X}(n)} \cdot \left[A^T P_{Y(n)}^{-1} Y(n) + P_{\hat{X}^-(n)}^{-1} \hat{X}^-(n) \right]$$

$$P_{\hat{X}(n)} = \left[A^T P_{Y(n)}^{-1} A + P_{\hat{X}^-(n)}^{-1} \right]^{-1}$$

Measurements

$$Y(n)$$

$$P_{Y(n)}$$

$$\hat{X}(n)$$

$$P_{\hat{X}(n)}$$

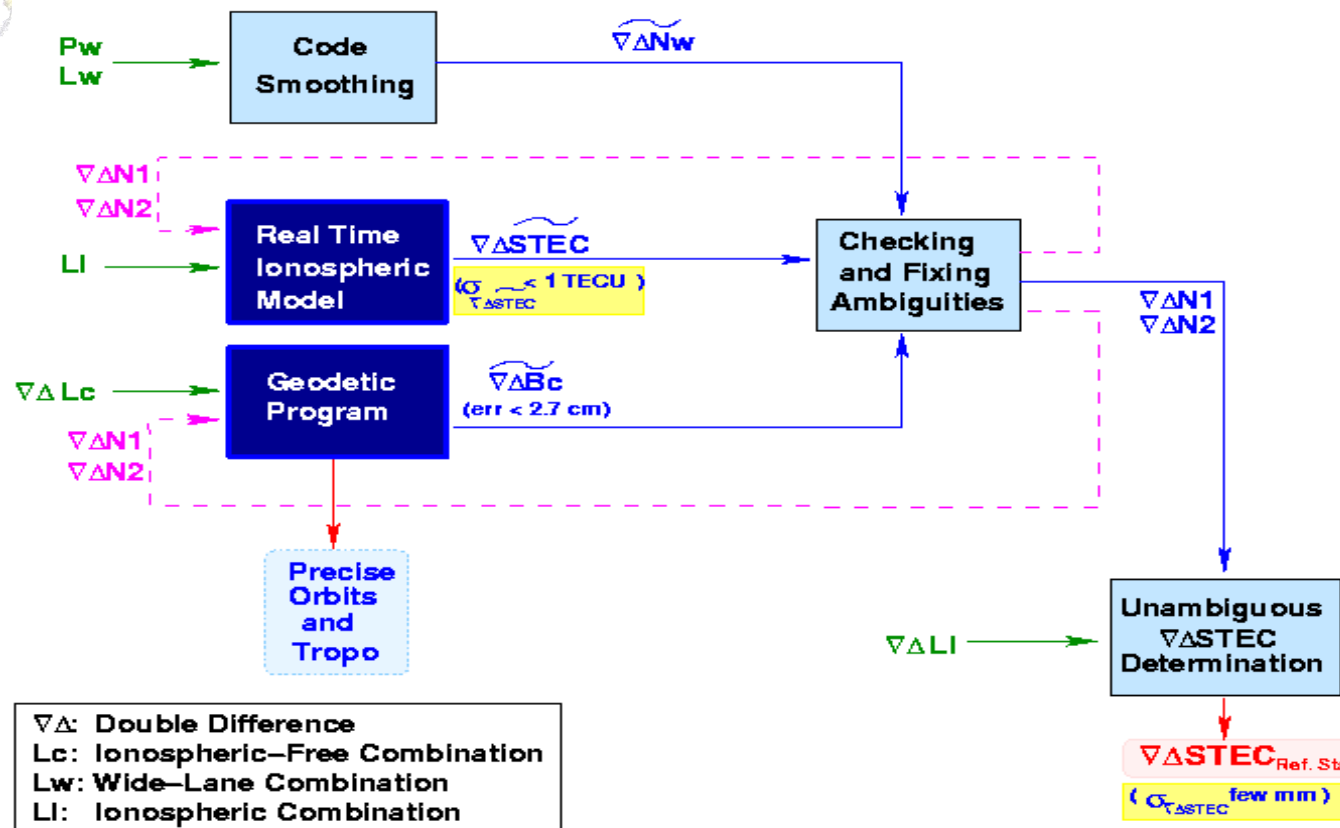
Estimating $X = (N_e, \dots, B_i, \dots)$ in a forward Kalman filter
solving $Y = A X$, with covariance matrices P :
 N_e for each voxel-time (random-walk) and pair receiver-
satellite cont. Arch (random variable)

$$L_I = STEC + B_I = \int_{\text{REC}}^{\text{SAT}} N_e dl + B_I = \sum_i \sum_j \sum_k (N_e)_{i,j,k} \Delta s_{i,j,k} + B_I$$

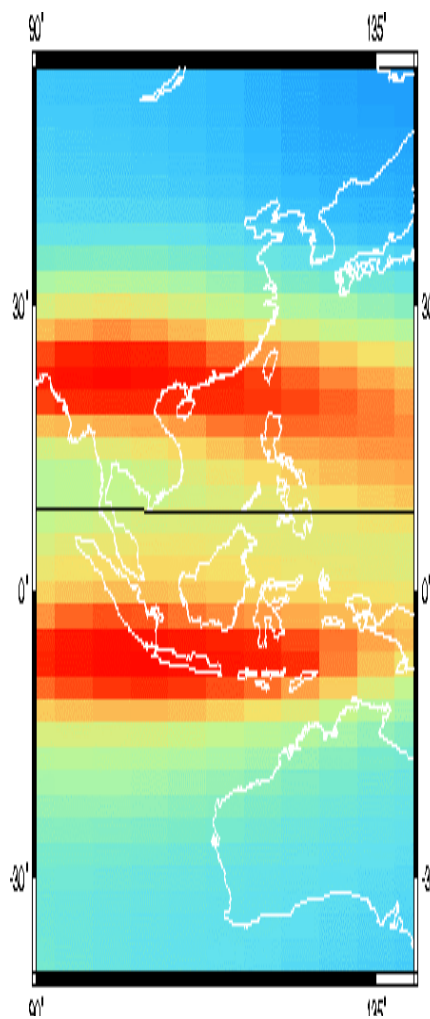
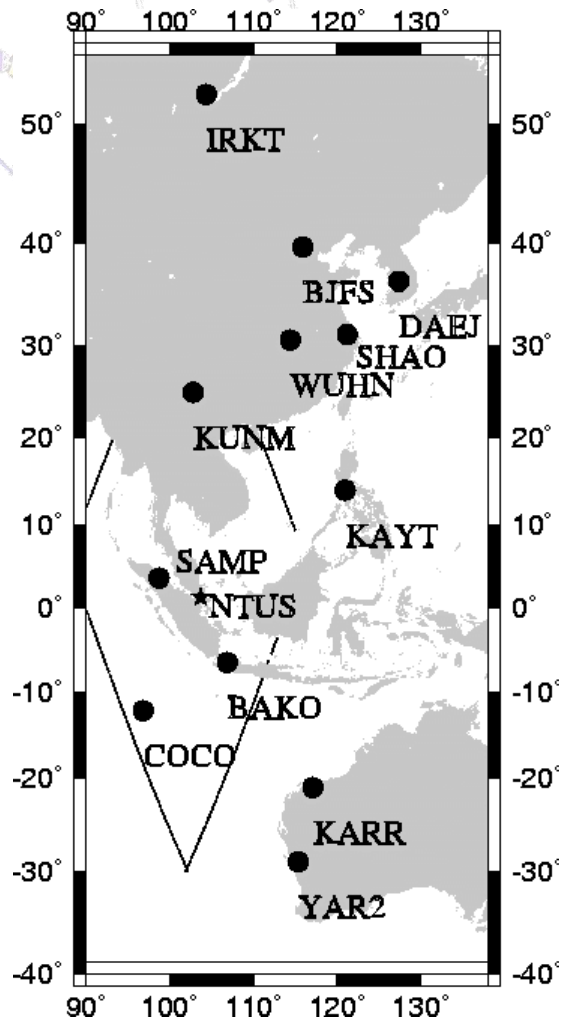
WARTK: Combining real-time Ionospheric & Geodetic models

Resolving the Ambiguous $\nabla\Delta$ STEC in Real Time for the Reference Stations

gAGE/UPC 24/07/01

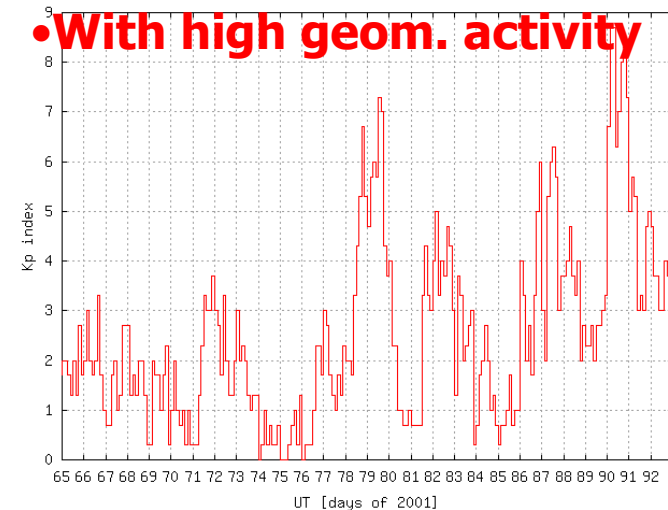


Accurate STEC from fixed GPS sites thousands km far



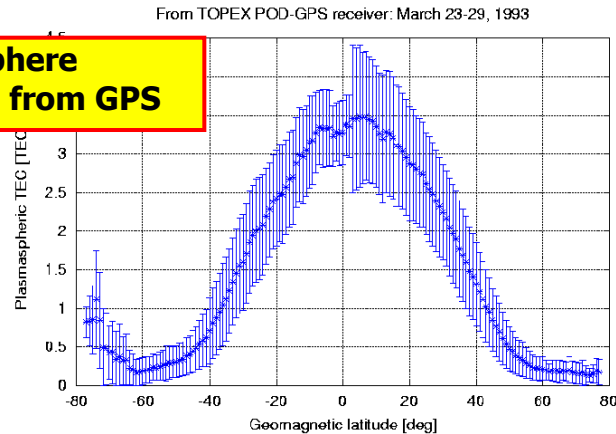
New Experiment:

- 13 IGS stations
- Distances: 1000-3000km
- Including equator, tropics
- 4 weeks 2001 65-92
- Solar Max. conditions
- Seasonal Max.
- With high geom. activity

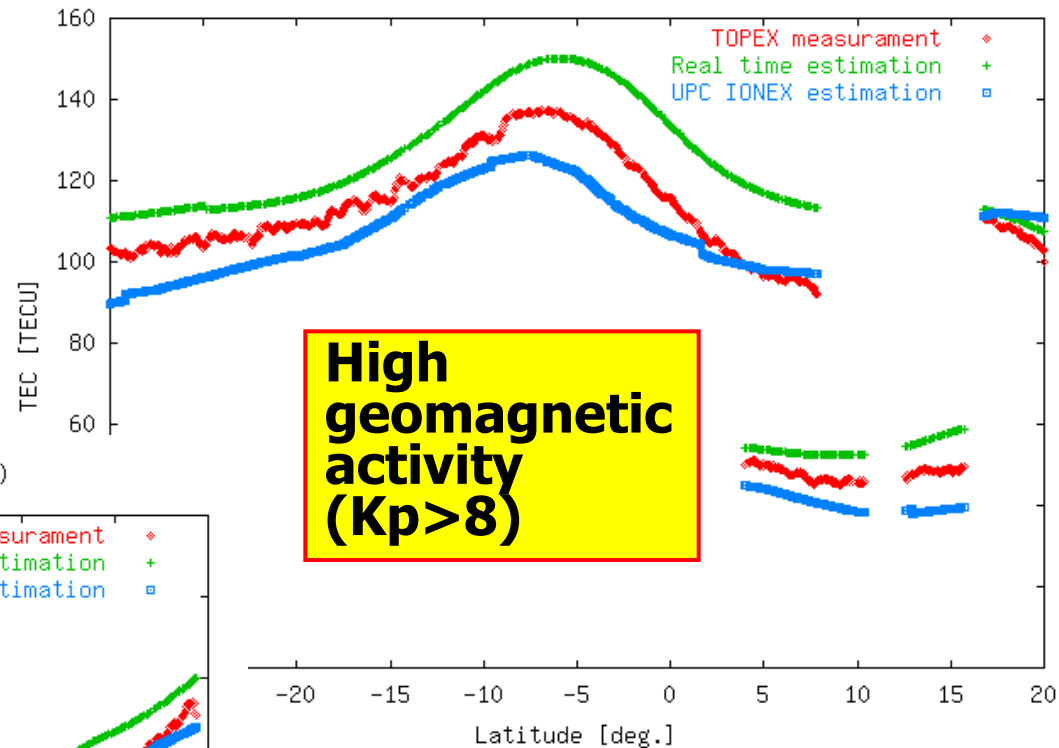


Tomo Real-Time (green) vs. TOPEX VTEC (red)

**Plasmasphere
observed from GPS**

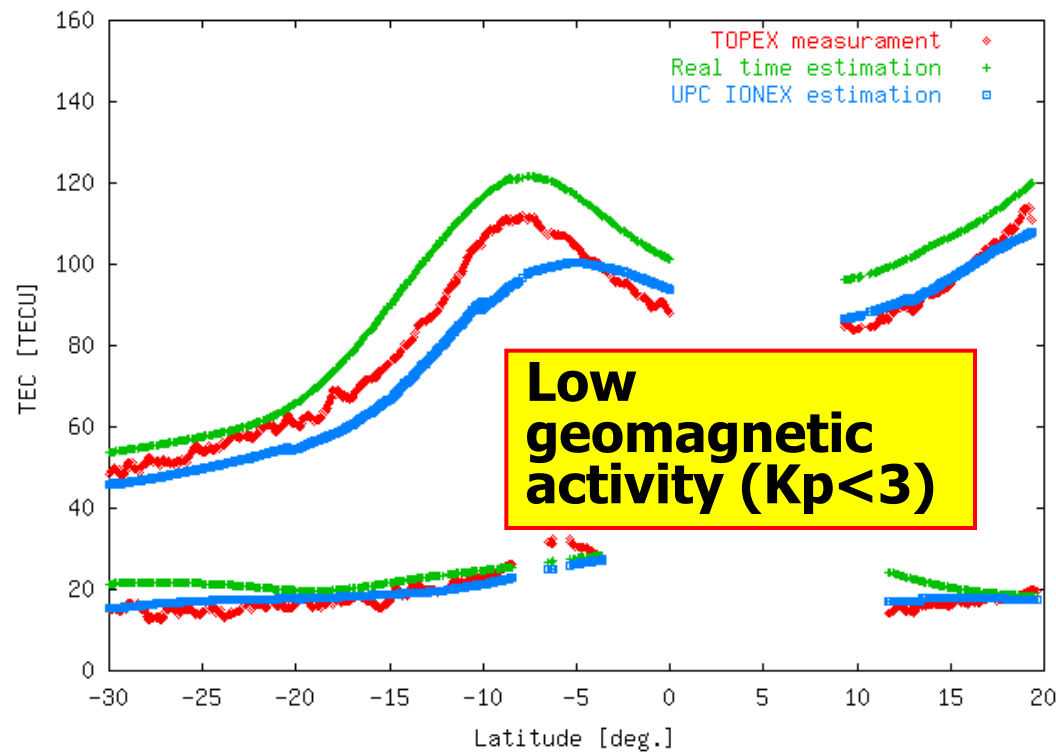


TOPEX TEC estimation; day 90 2001 ($K_p > 8$)

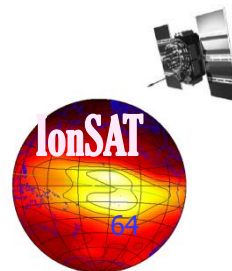


**High
geomagnetic
activity
($K_p > 8$)**

TOPEX TEC estimation; day 67 2001 ($K_p < 3$)



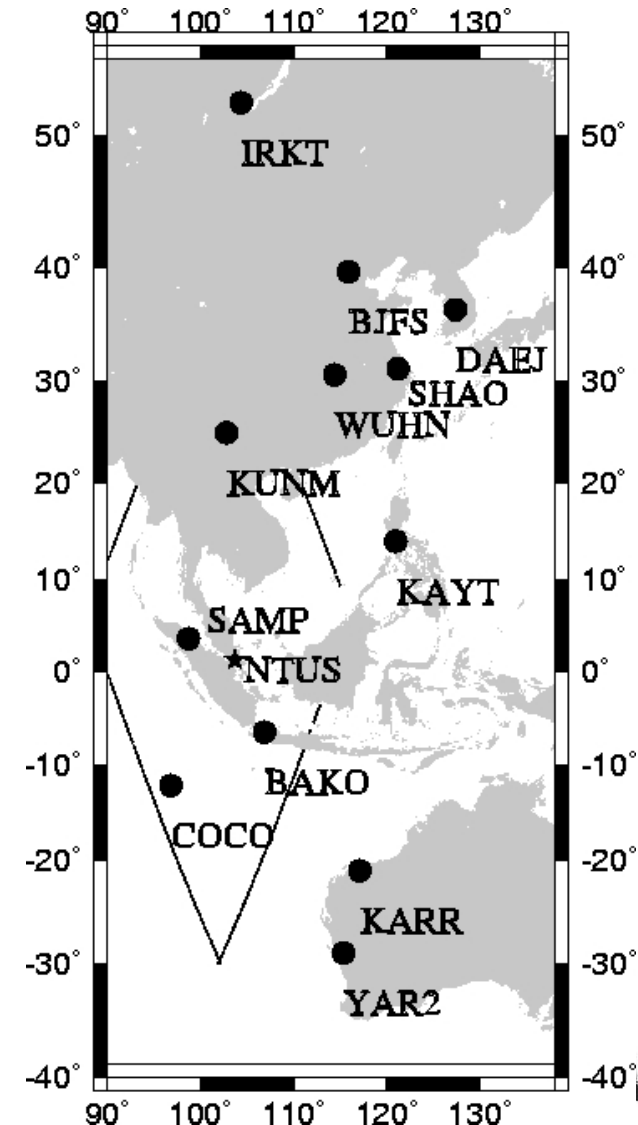
**Low
geomagnetic
activity ($K_p < 3$)**



Real-time DDSTEC vs. truth



Sta.	Ref.	Dist. (km)	% Succ.	RMS [TECU]	# Obs.
IRKT	DAEJ	2507	93	1.2	8329
BJFS	DAEJ	1067	91	1.4	8131
KUNM	DAEJ	2640	95	1.0	3900
WUHN	DAEJ	1369	92	1.4	6358
SAMP	KARR	3341	95	1.1	6441
COCO	KARR	2354	97	0.9	9963
BAKO	KARR	1939	90	1.5	6121
YAR2	KARR	909	97	0.8	12630



(More details in Hernández-Pajares, M., J.M. Juan, J. Sanz, O.L. Colombo, Improving the real-time ionospheric determination from GPS sites at Very Long Distances over the Equator, *J.Geophys.Res.*, 107(A10), 1296, October 2002)

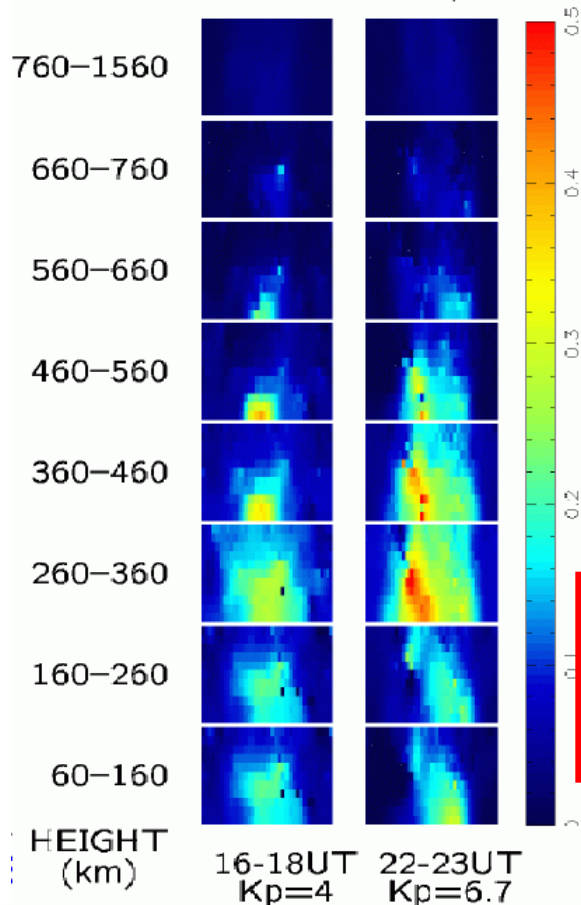
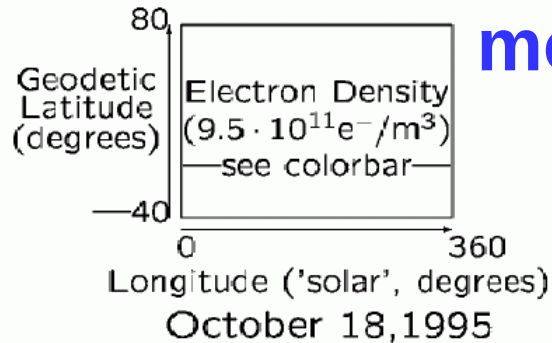


- Using space-based GNSS data as well (N_e)

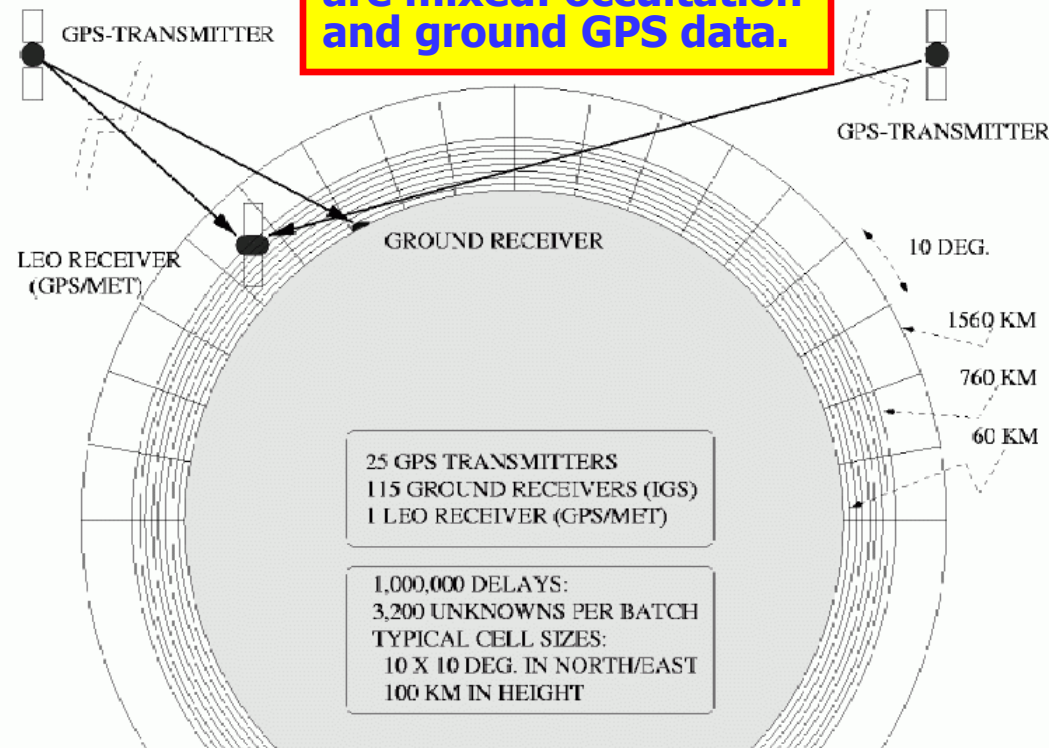


- Space-based and ground-based GNSS data

Estimating electron density in a 3D Voxel model (regional/global)



DE CATALUNYA
BARCELONATECH

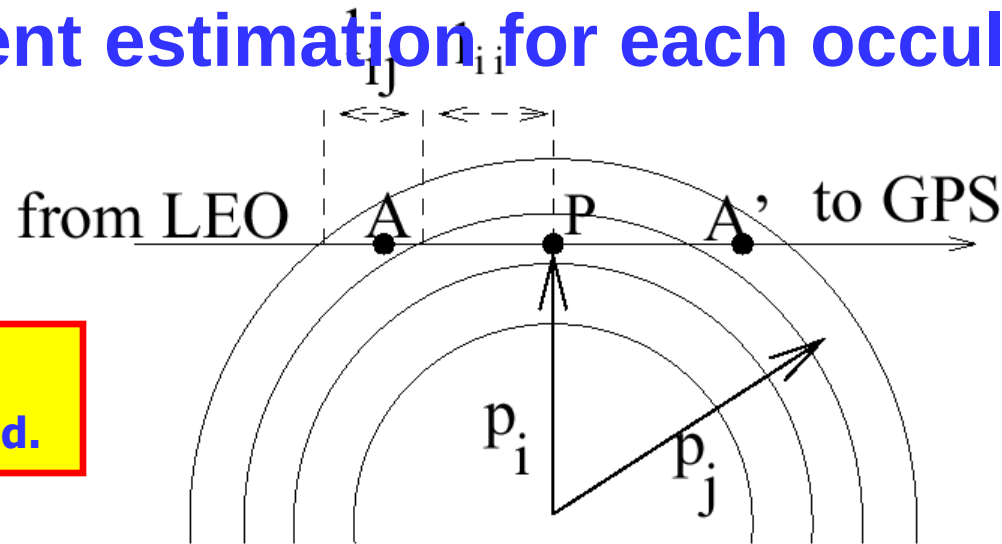


Different kind of data are mixed: occultation and ground GPS data.

This is an example corresponding to the combination of both complementary type of GPS data (ground-based from IGS network and space-based from GPS/MET LEO) in the framework of a 3D voxel tomographic model solved by means of a Kalman filter (Oct.18, 1995 geomagnetic storm).

More details in: Hernández-Pajares M., J.M. Juan, J. Sanz and J.G. Sole, Global observation of the ionospheric electronic response to solar events using ground and LEO GPS data, *Journal of Geophysical Research-Space Physics*, Vol.61, p.1237-1247, 1998.

Classical Abel transform (independent estimation for each occultation)

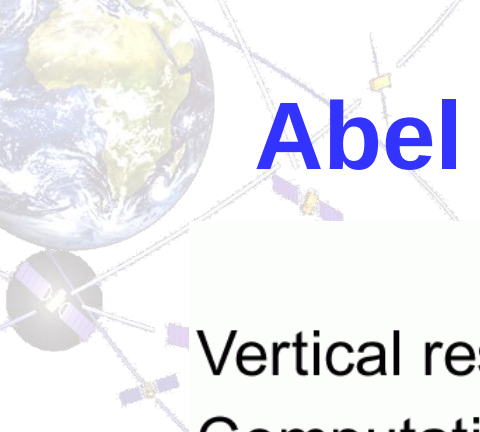


The classical spherical symmetry hypothesis can be formulated as:

$$N_e(LT, LAT, H) = \Phi(H)$$

Recursive solution starting from the outer ray. STEC corresponding to the impact parameter p_i :

$$STEC(p_i) = 2 \cdot l_{ii} \cdot N_e(p_i) + \sum_{j=1}^{i-1} 2 \cdot l_{ij} \cdot N_e(p_j)$$



Abel Transform vs. 3D-Voxel model

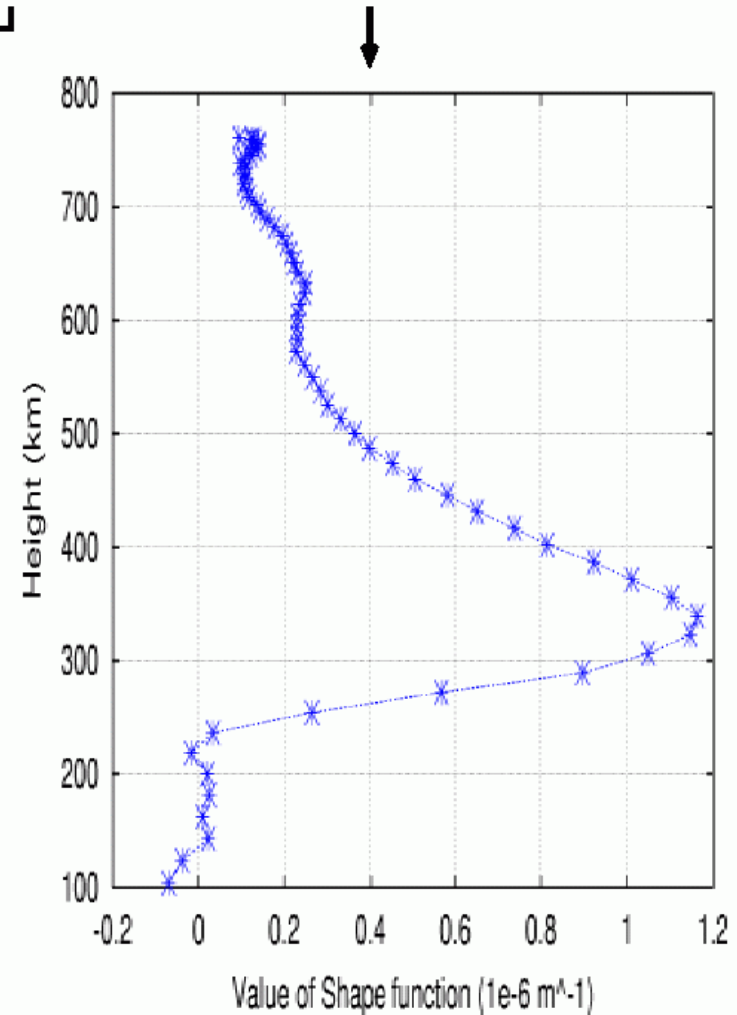
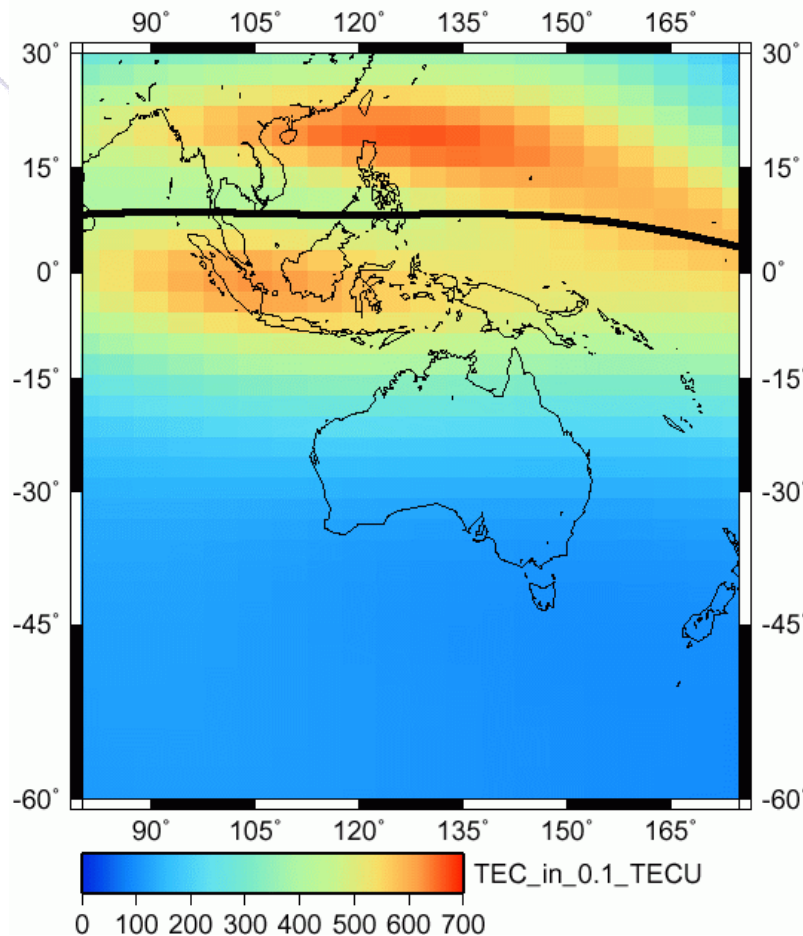
	Abel Transform	3D Voxels
Vertical resolution	$\simeq 1$ km	Tens / hundred km
Computational load	Low ($\simeq 10^2$ unk.)	High ($> 10^3$ unk.)
Sph. Sym. Assump.	Yes	No
Topside mismod.	Yes	No
Assimilation capab.	No	Yes

Our goal: maintaining its other advantages, to avoid these limitations of the classical Abel approach:

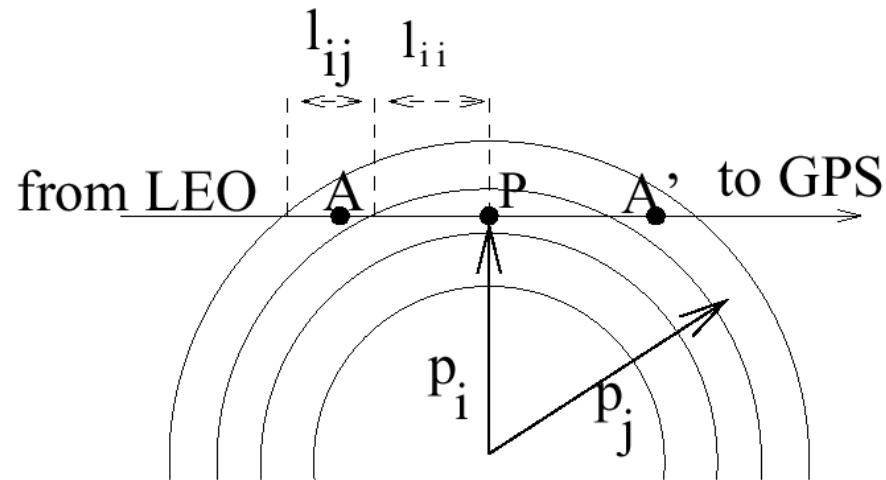
- Overcoming the spherical symmetry assumption.
- Overcoming the topside electron content mismodelling.

Separability concept

$$N_e(LT, AT, H) = VTEC(LT, LAT) \cdot F(H)$$



Improved Abel transform



Let's assume a more general hypothesis that just the spherical symmetry:

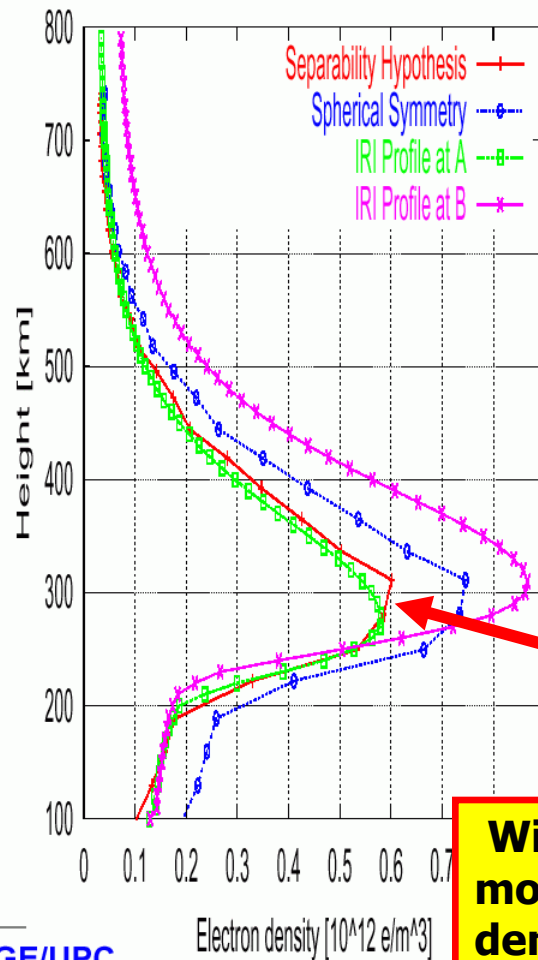
$$N_e(LT, LAT, H) = VTEC(LT, LAT) \cdot F(H)$$

Starting from the outer ray, the STEC corresponding to the impact parameter p_i can be modeled by:

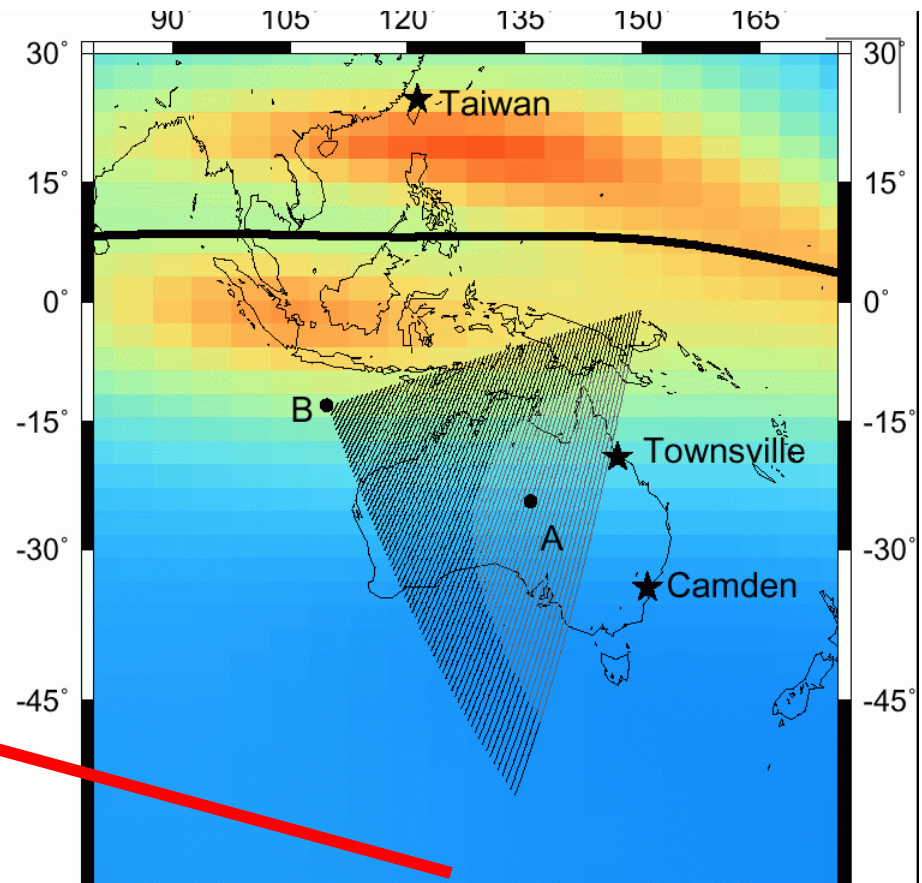
$$STEC(p_i) = 2 \cdot l_{ii} \cdot VTEC(LT_{ii}, LAT_{ii}) \cdot F(p_i) + \sum_{j=1}^{i-1} l_{ij} \cdot [VTEC(LT_{ij}, LAT_{ij}) + VTEC(LT'_{ij}, LAT'_{ij})] \cdot F(p_j)$$

Benefits of separability hypothesis

Retrieval of occultation with GPSRPN 16 at 4hUT with IRI prediction



gAGE/UPC

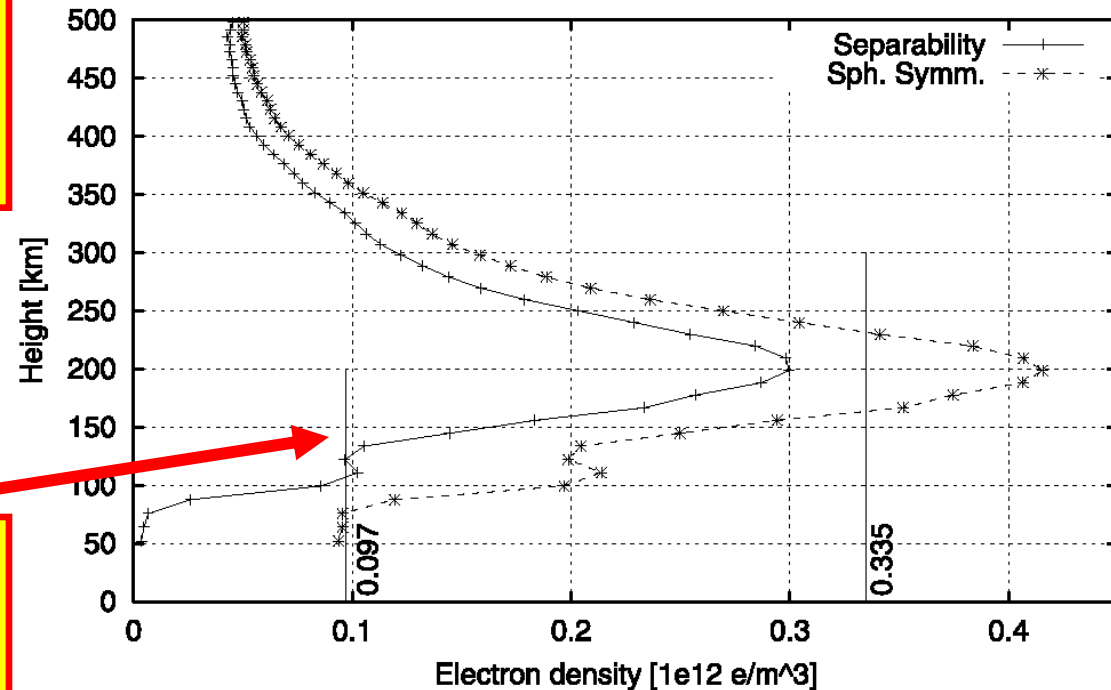


With the TEC (computed from ground GPS data) modelling the horizontal variation, the electron density at tangent point A (green) is estimated from the GPS occultation data better (red) that assuming spherical symmetry (blue).

Separability: improvement in E layer estimation

Comparison with LD160 ionosonde (Leningrad, E30.7 N60.0)

GPSMET prof. at 1443km from LD160 at 10hUT(12hLT), 1995 October 11th



The improved Abel can provide reliable estimates of the NmE value as well (not just NmF2 values).

Table 2. Table of foE Errors With Respect to Ionosonde Value^a

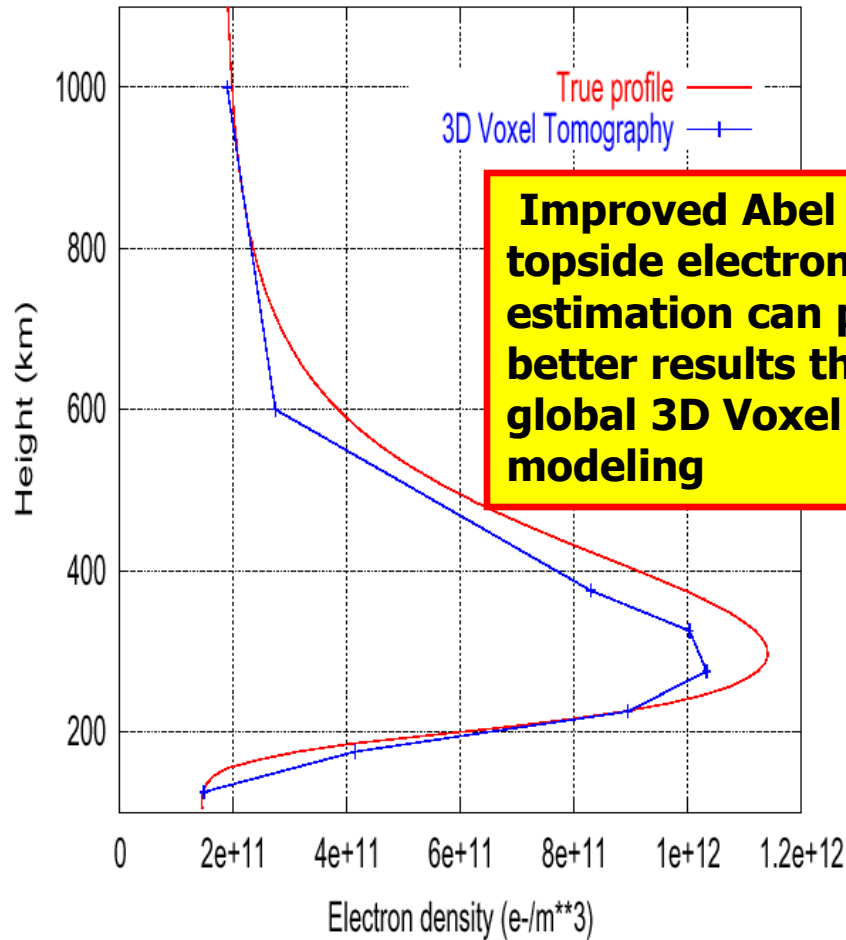
	N.comp	Sep. Hyp. RMS: MHz [%]	Sph. Symm. RMS: MHz [%]
E layer	135	0.4 [17.1]	0.7 [28.5]
Es layer	35	0.5 [16.2]	1.0 [30.4]

^aThe error is Absolute RMS in MHz and Percentual relative RMS difference in brackets.

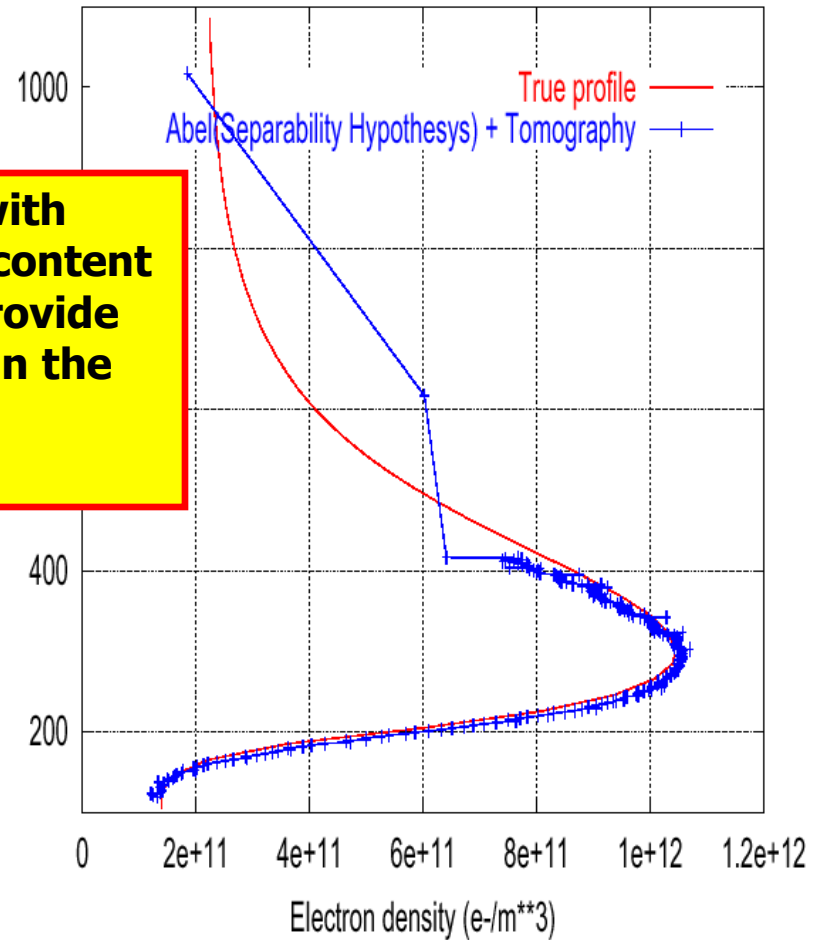
3D voxel profile vs Improved Abel+topside 3D voxels

CHAMP-PRN07: 2001-265, 17h42m (1 Hz)

CHAMP-PRN07: 2001-265, 17h42m (1 Hz)



**Improved Abel with
topside electron content
estimation can provide
better results than the
global 3D Voxel
modeling**

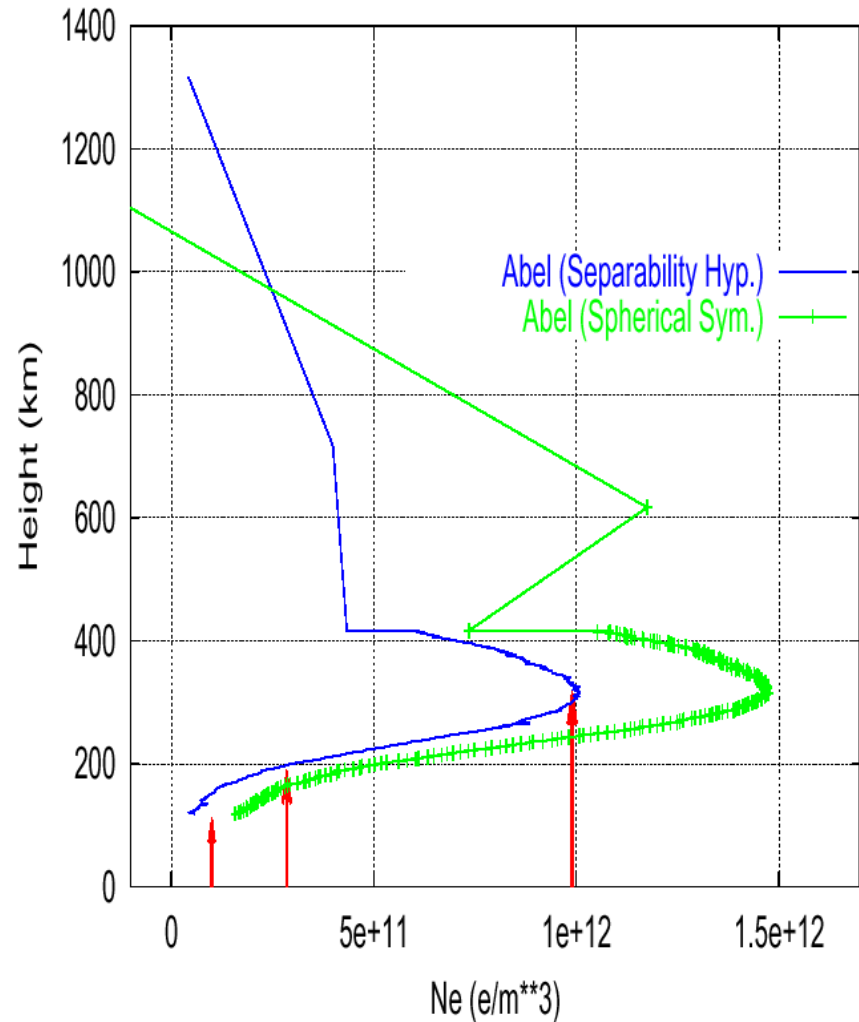


Results with CHAMP data: Separability vs Spherical Symm. with 3D topside model

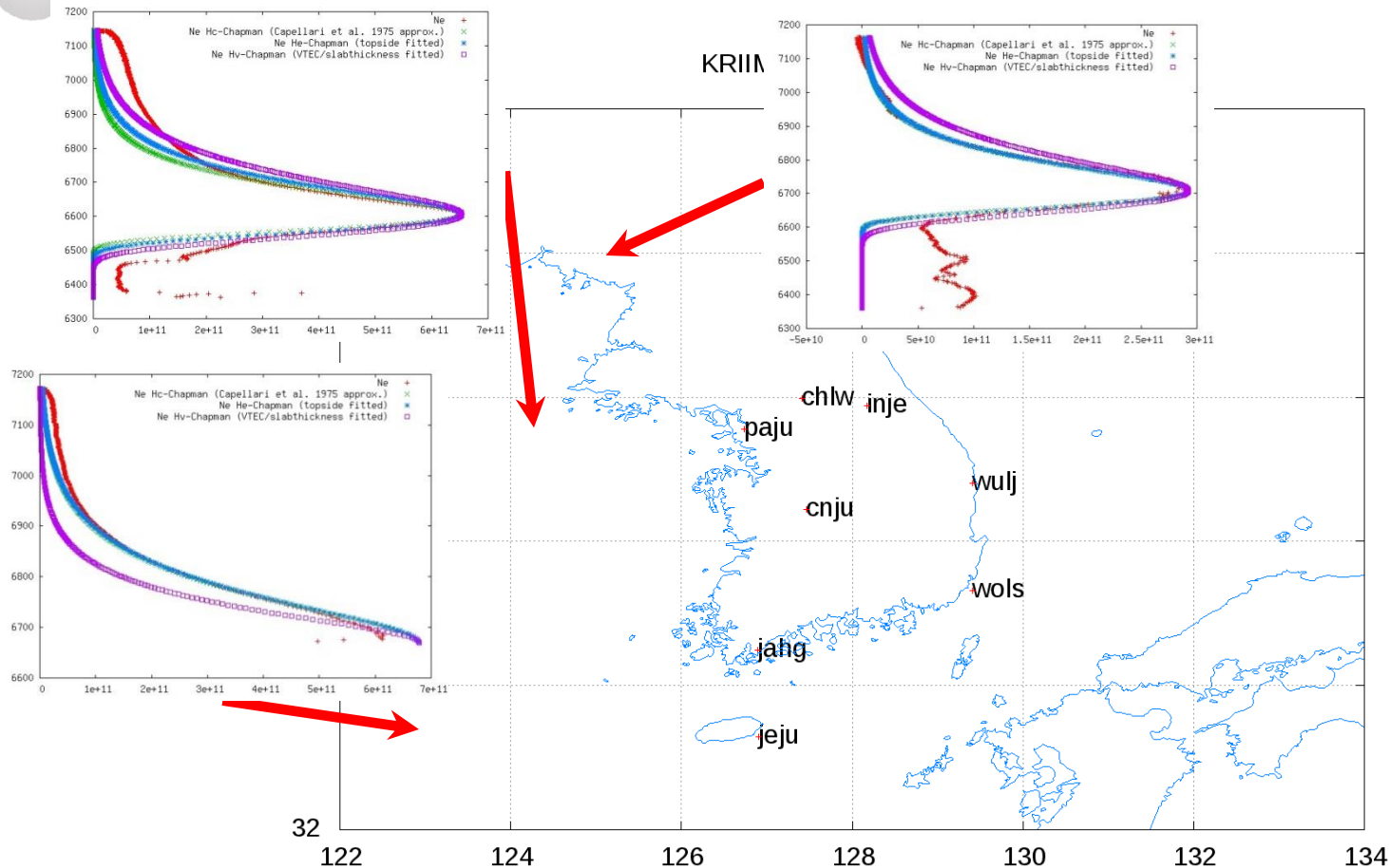
The improved Abel transform, including separability hypothesis and upper ionosphere voxel modelling, provides a quite better performance.

This method provides such results **without a priori solution**, using **only carrier phase data** for both VTEC and occultation, involving **very low computational burden**, and can be applied in **real-time**.

PROFILE at 1042LT LAT=65deg vs. COLLEGE ionosonde (2001 265)

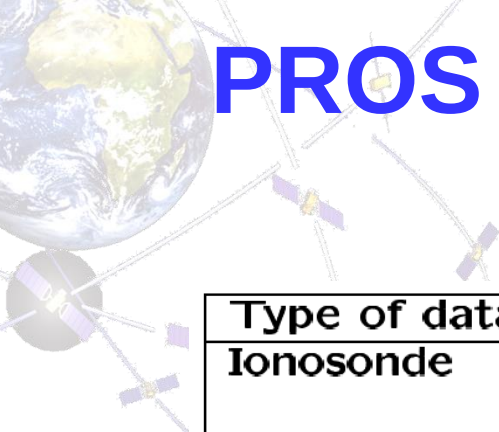


Example: COSMIC occultations over Korea during day 85, 2011





•Ground-based GNSS and Ionosonde data



PROS and CONS of different kind of ionospheric data

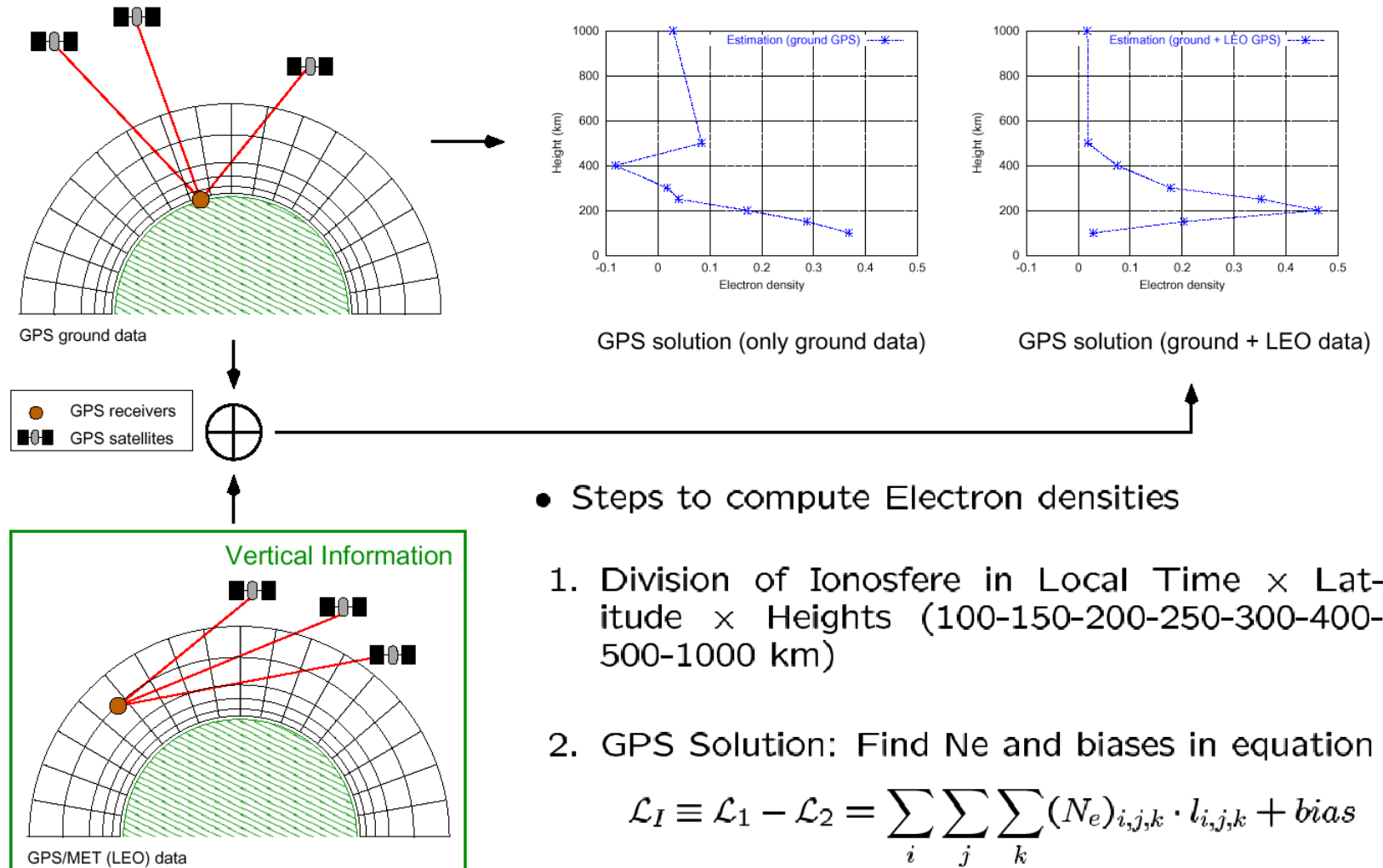
Type of data	Advantages	Disadvantages
Ionosonde	High resolution data up to the maximum	No information beyond hmF2 Only local vertical profiles
Ground GPS	TEC estimations with errors of few TECUs Extensive GPS networks	Bad vertical resolution
LEO GPS	Vertical information of Ionospheric electron density	Occultations only occur for specific Local Time for different locations

Combination $\Rightarrow$ Compensation of weak spots with strong points $\Rightarrow$

1. 3D Ionospheric information possible in either regional or global scale
2. Retrieval of topside information

Occultation + ground GPS data

Data Assimilation (I): ground and LEO GPS data scheme

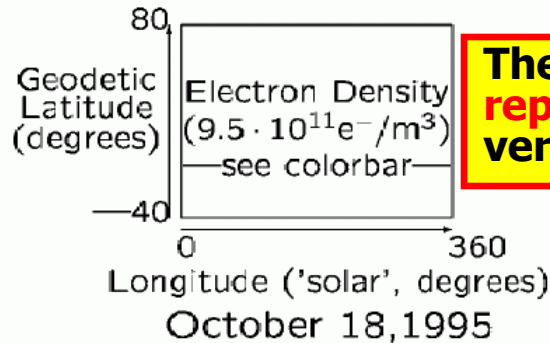


Steps to compute Electron densities

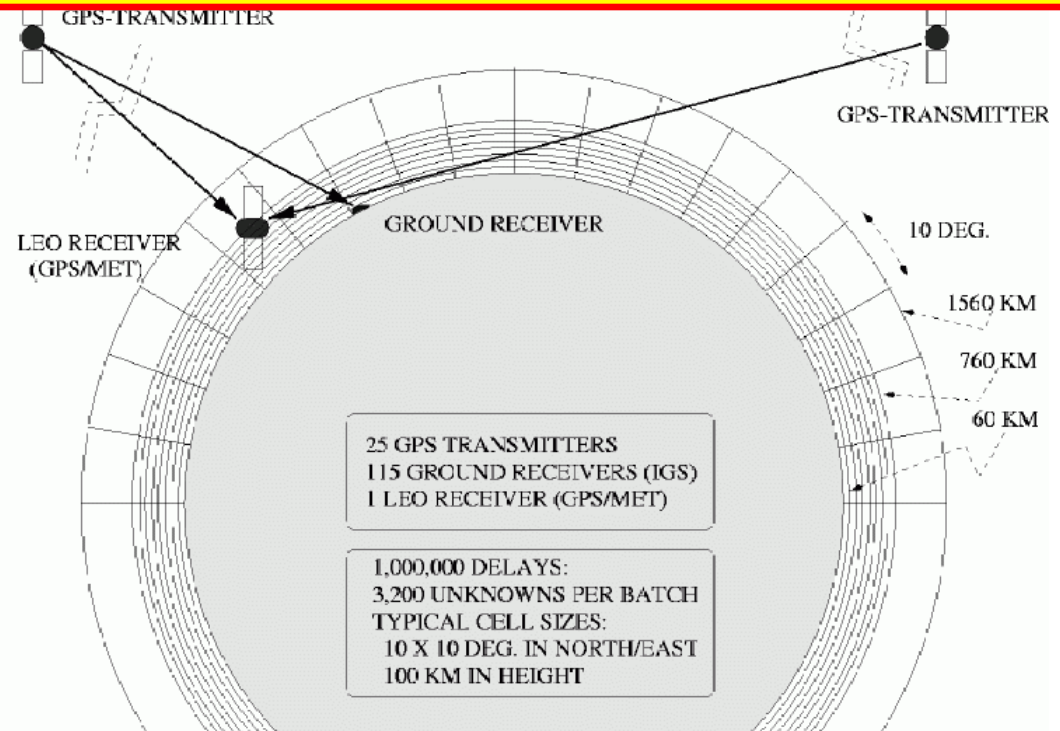
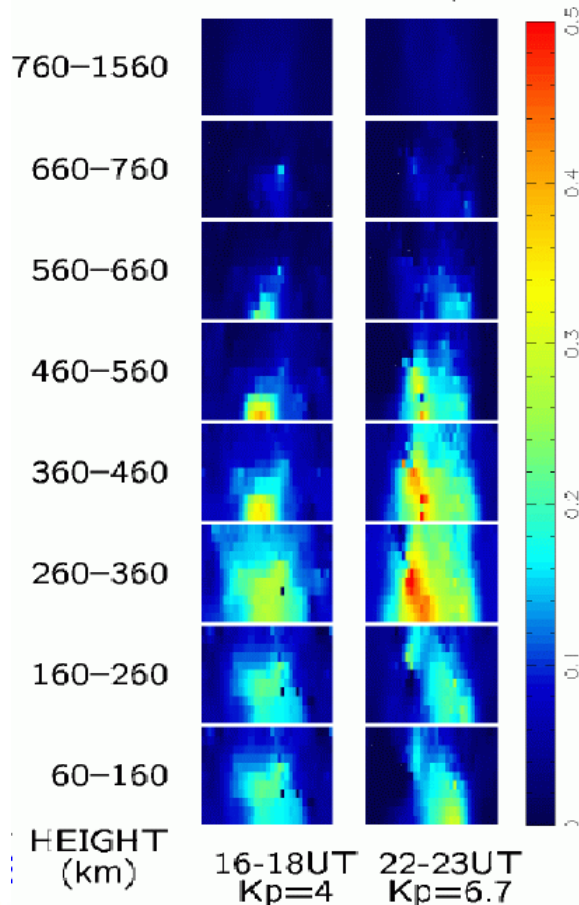
1. Division of Ionosphere in Local Time $\times$ Latitude $\times$ Heights (100-150-200-250-300-400-500-1000 km)
2. GPS Solution: Find N_e and biases in equation

$$\mathcal{L}_I \equiv \mathcal{L}_1 - \mathcal{L}_2 = \sum_i \sum_j \sum_k (N_e)_{i,j,k} \cdot l_{i,j,k} + bias$$

Results with GPS/MET and IGS data



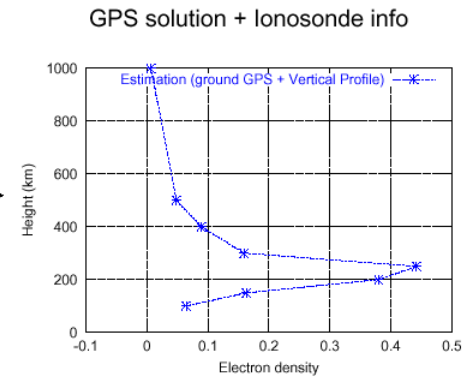
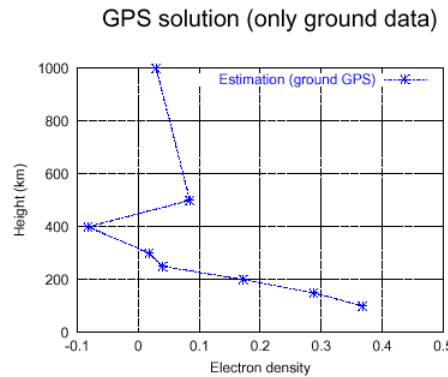
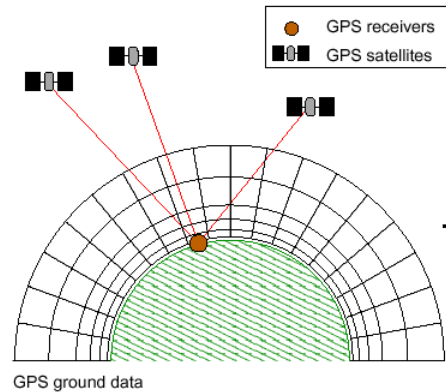
The question is: could the **LEO GPS data** be efficiently replaced by other kind of ionospheric data with vertical information such as **ionosonde data**?



The combination of both **LEO occultation** and **ground IGS GPS data** makes feasible the **global tomography**.



Ionosonde + ground GPS data



• Steps to compute Electron densities

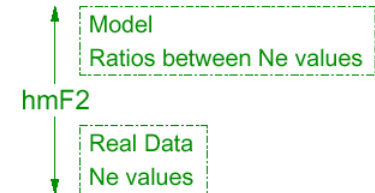
1. Division of Ionosphere in Local Time $\times$ Latitude $\times$ Heights (100-150-200-250-300-400-500-1000 km)

2. GPS Solution: Find N_e and biases in equation

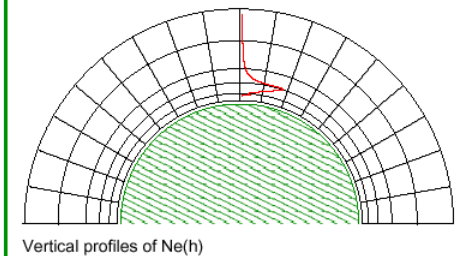
$$\mathcal{L}_I \equiv \mathcal{L}_1 - \mathcal{L}_2 = \sum_i \sum_j \sum_k (N_e)_{i,j,k} \cdot l_{i,j,k} + bias$$

3. Apply **Constraining Scheme** with Vertical profile data from models like NeQuick, IRI...

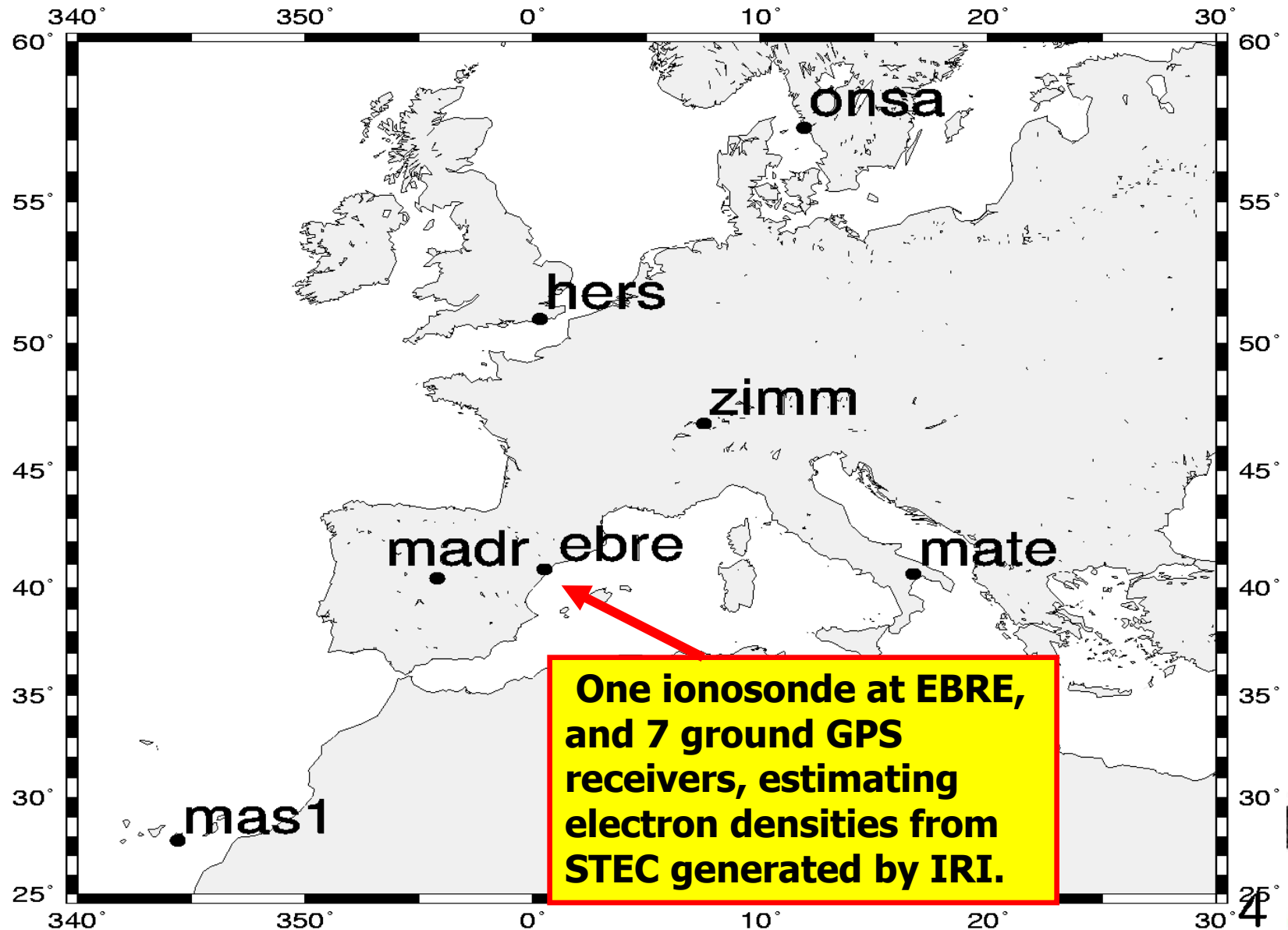
Constraining Scheme



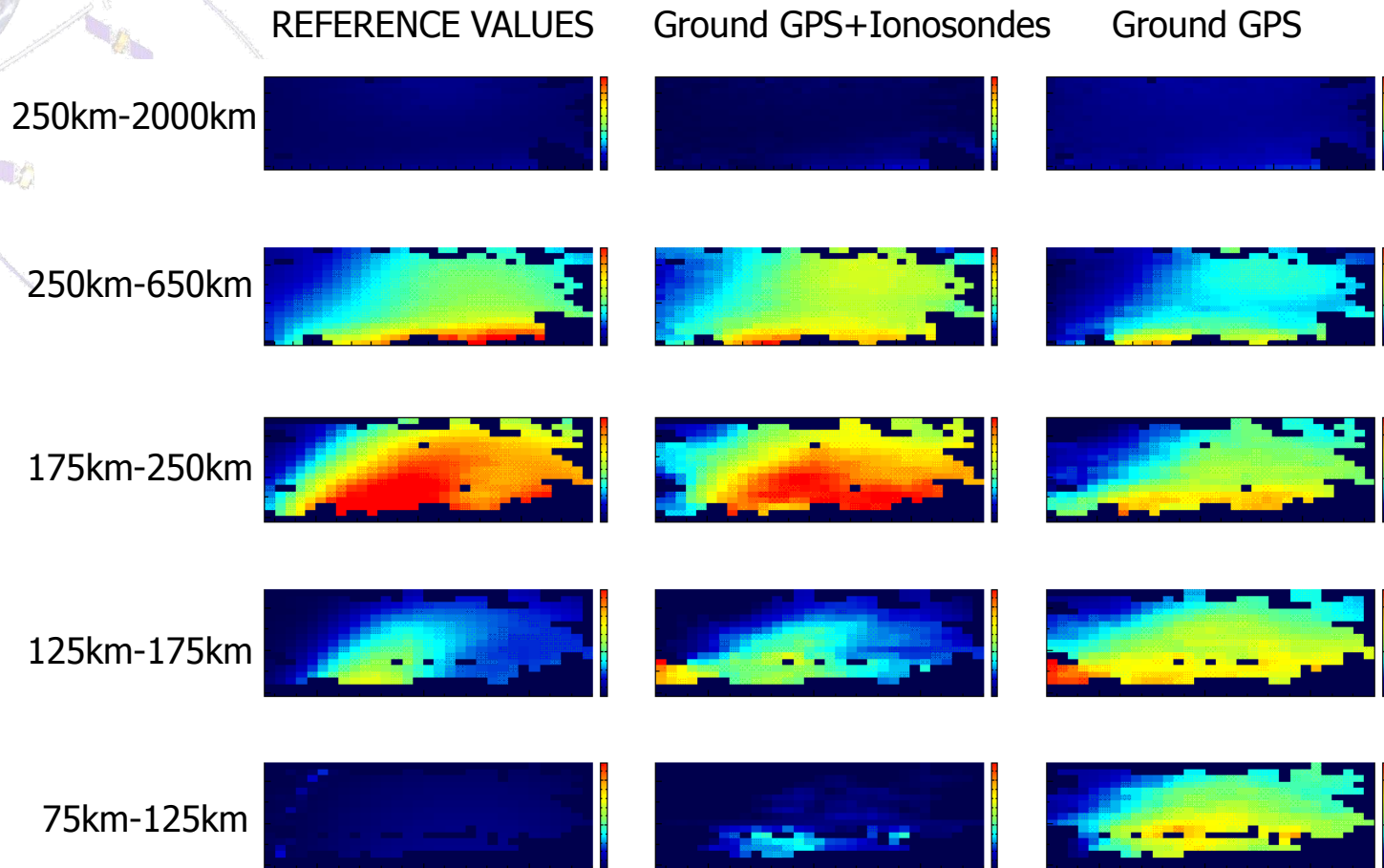
Vertical Information



Ionosondes+ground GPS: Simulation study

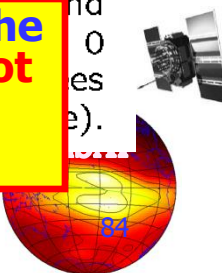


Simulation study: snapshot of results



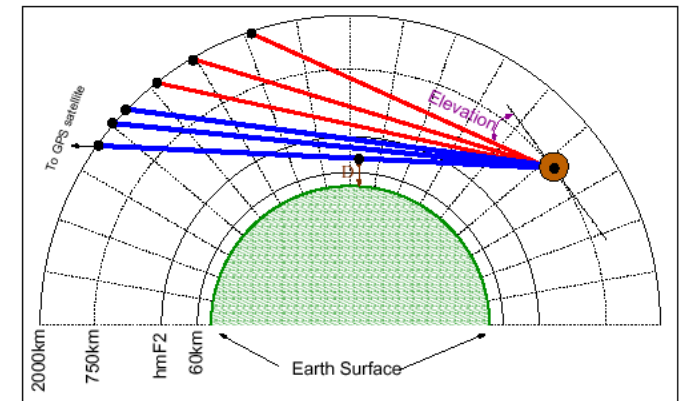
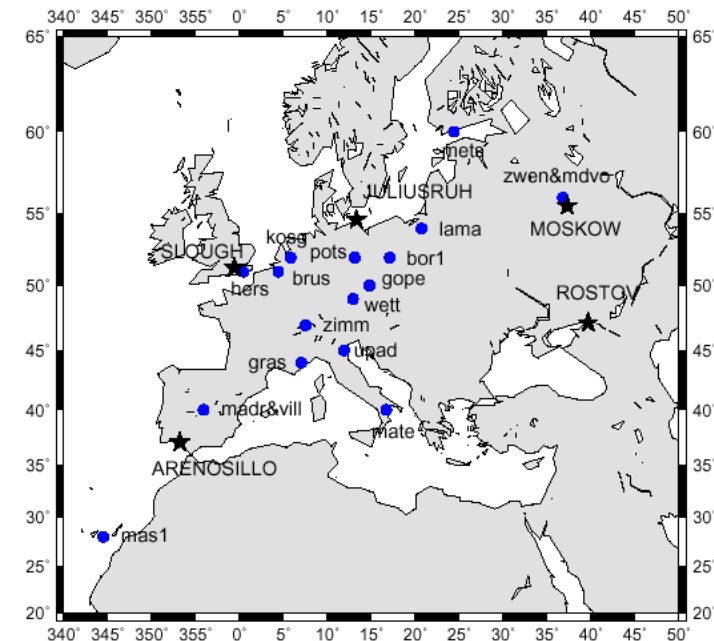
Each plot represent the corresponding mean electron density for layer 1 (bottom) to layer 5 (top) for the 500 km to 2000 km altitude range. The color scale on the right of each plot indicates the electron density in units of 10^{12} electrons/m³. The color scale ranges from 0 (dark blue) to 10 (red).

The mean electron densities obtained with 7 GPS receivers in the ground and with 1 ionosonde are in good agreement. This is not the case using just the ground GPS data (typical correlated solution).



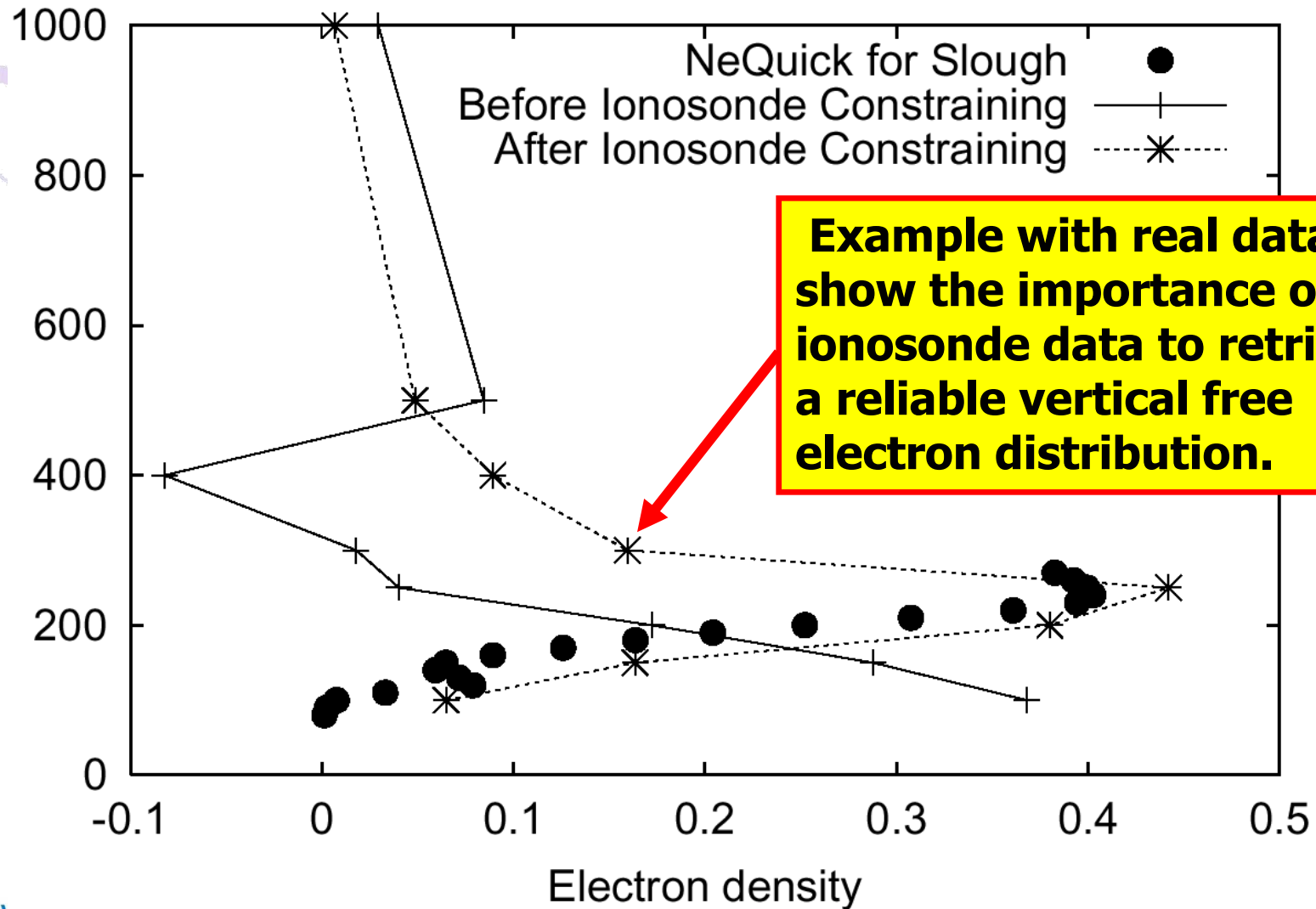
Results with real data over Europe

- Data set obtained from Ionosonde and ground GPS receiver from Europe, day 1995 October 18th.
- Verification procedure after cell densities have been computed:
 1. GPS to GPS/MET rays divided according to the cells crossed
 2. Modelled STEC reconstructed with the obtained densities
 3. Results compared with actual GPS/MET observations



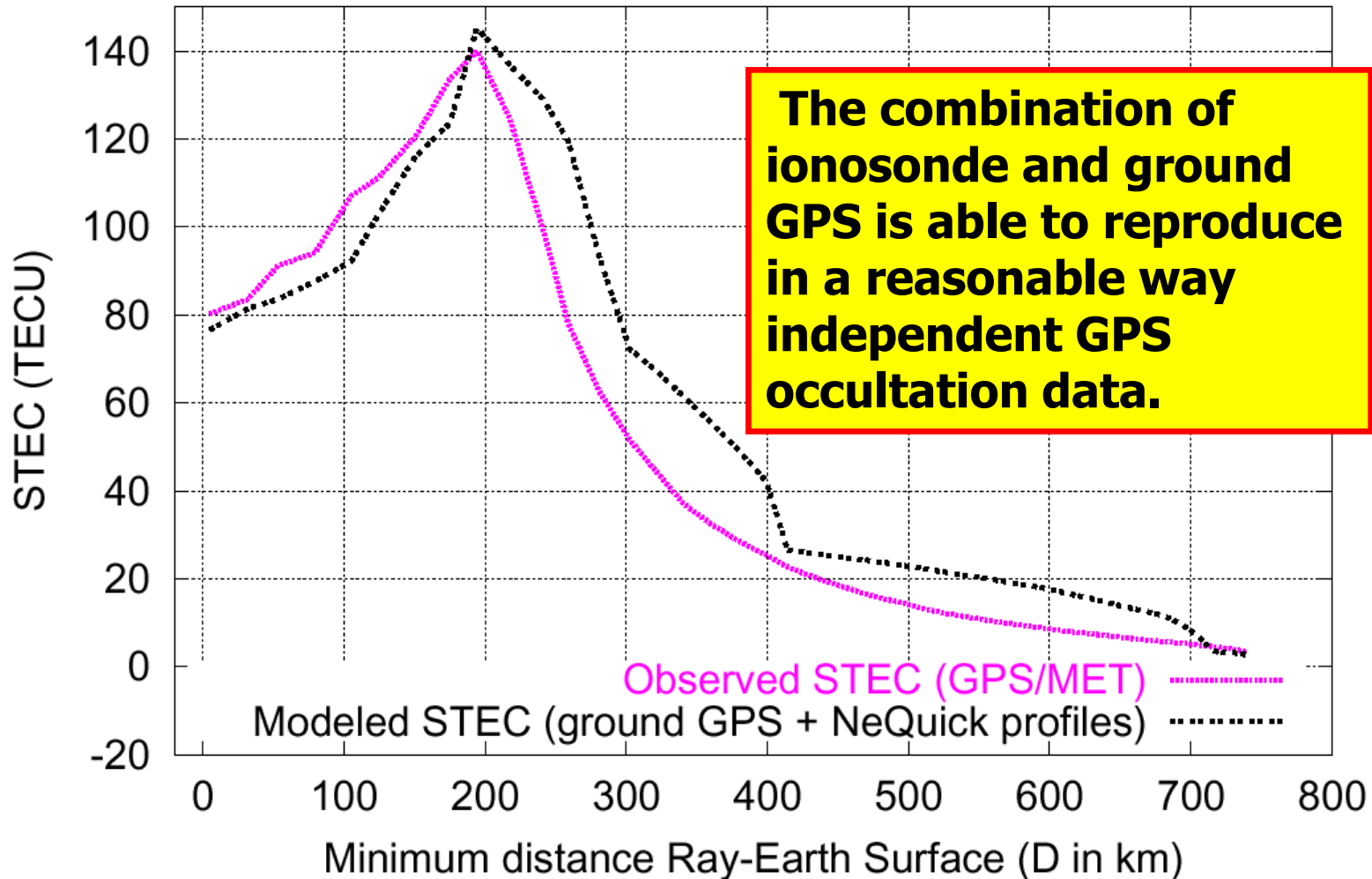
One example of improvement

Estimation for Slough at 10hUT constraining El Arenosillo

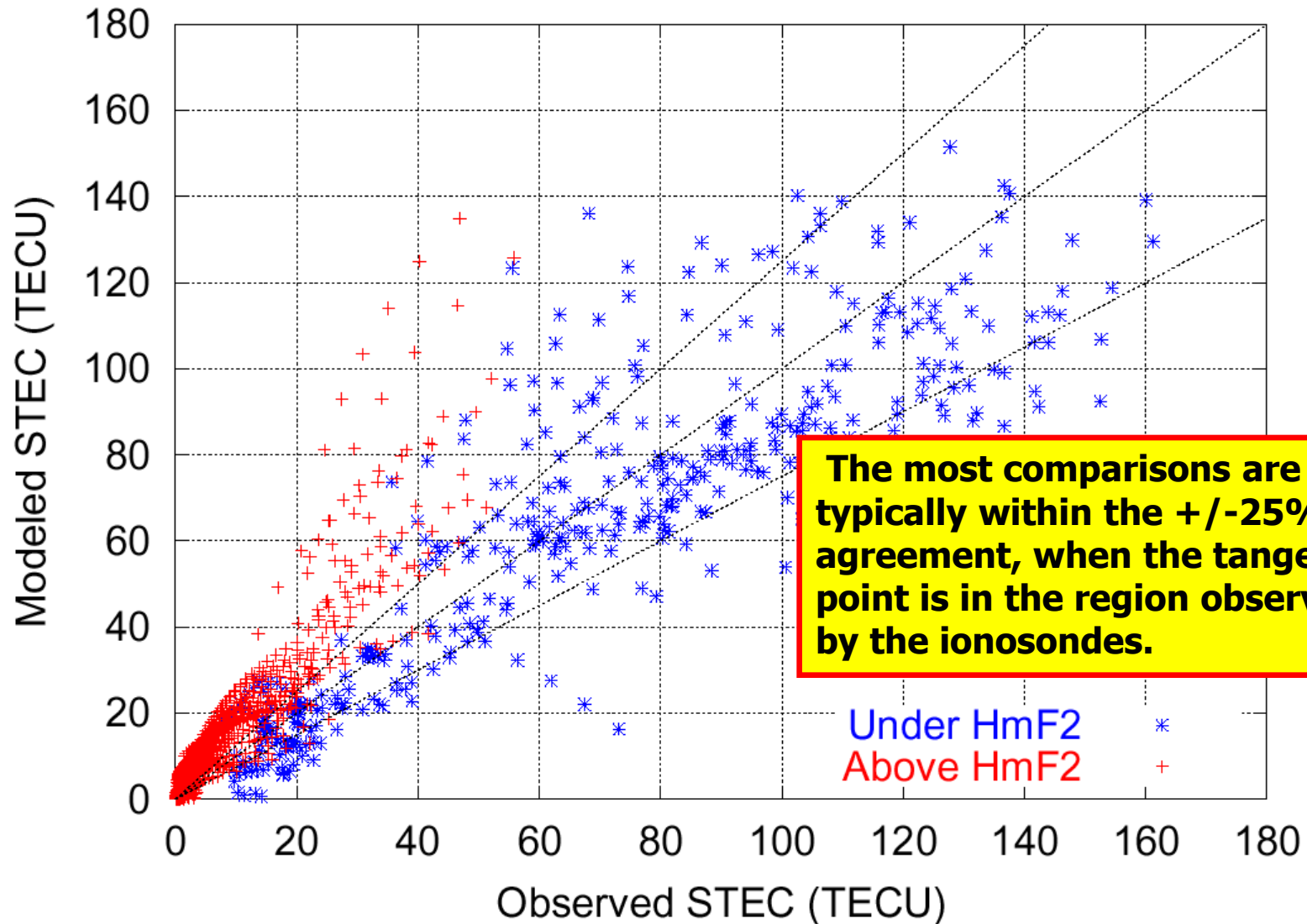


Validating with GPS/MET STEC

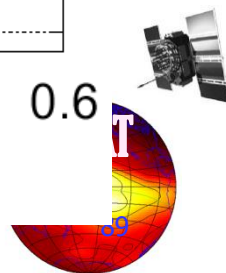
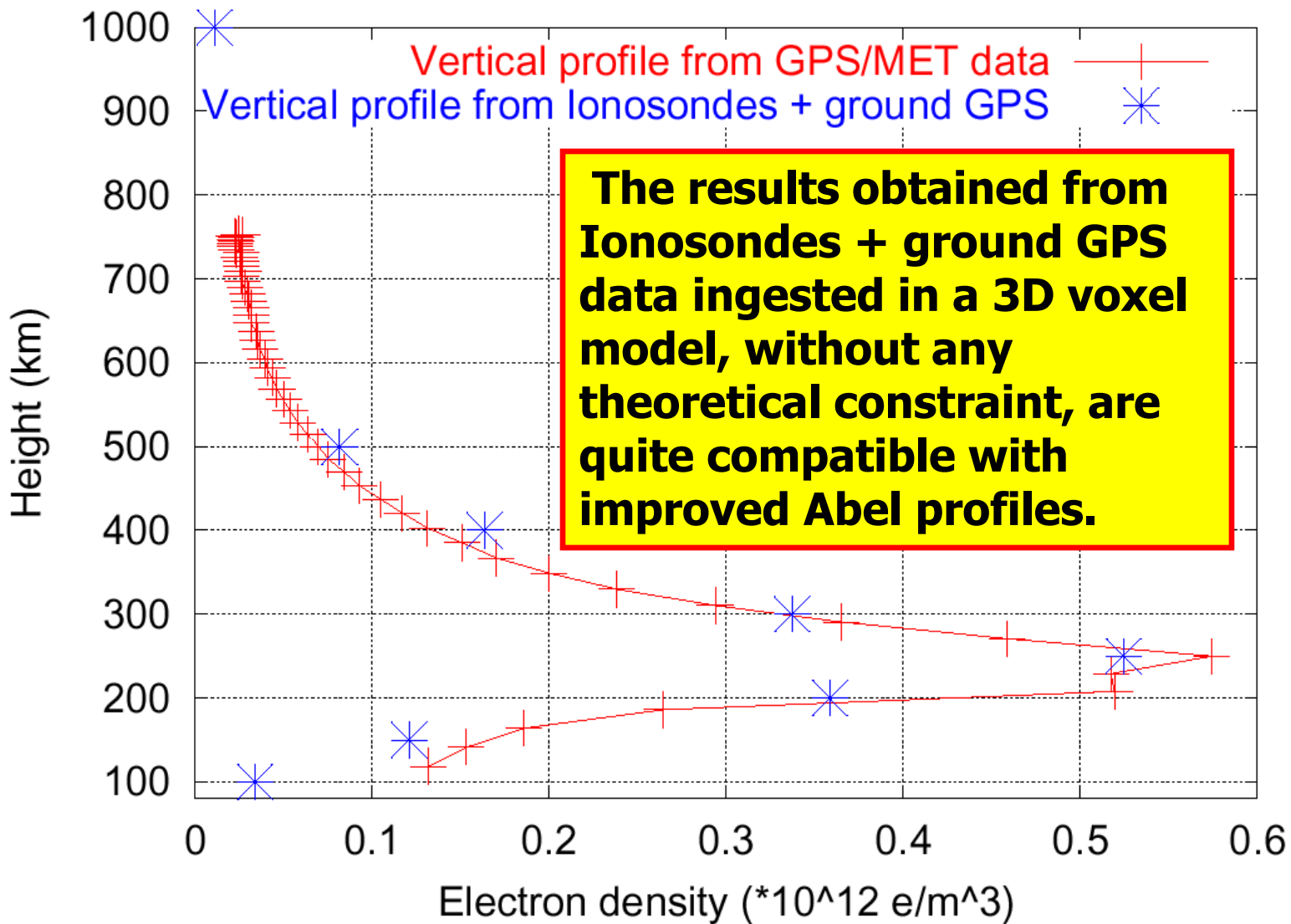
GPS Met occultation for GPS PRN 31 at 05hUT, October 18th, 1995



All the comparisons with GPS/MET STEC

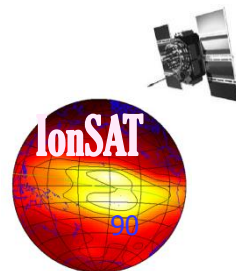


Example of comparison with occultation GPS profile

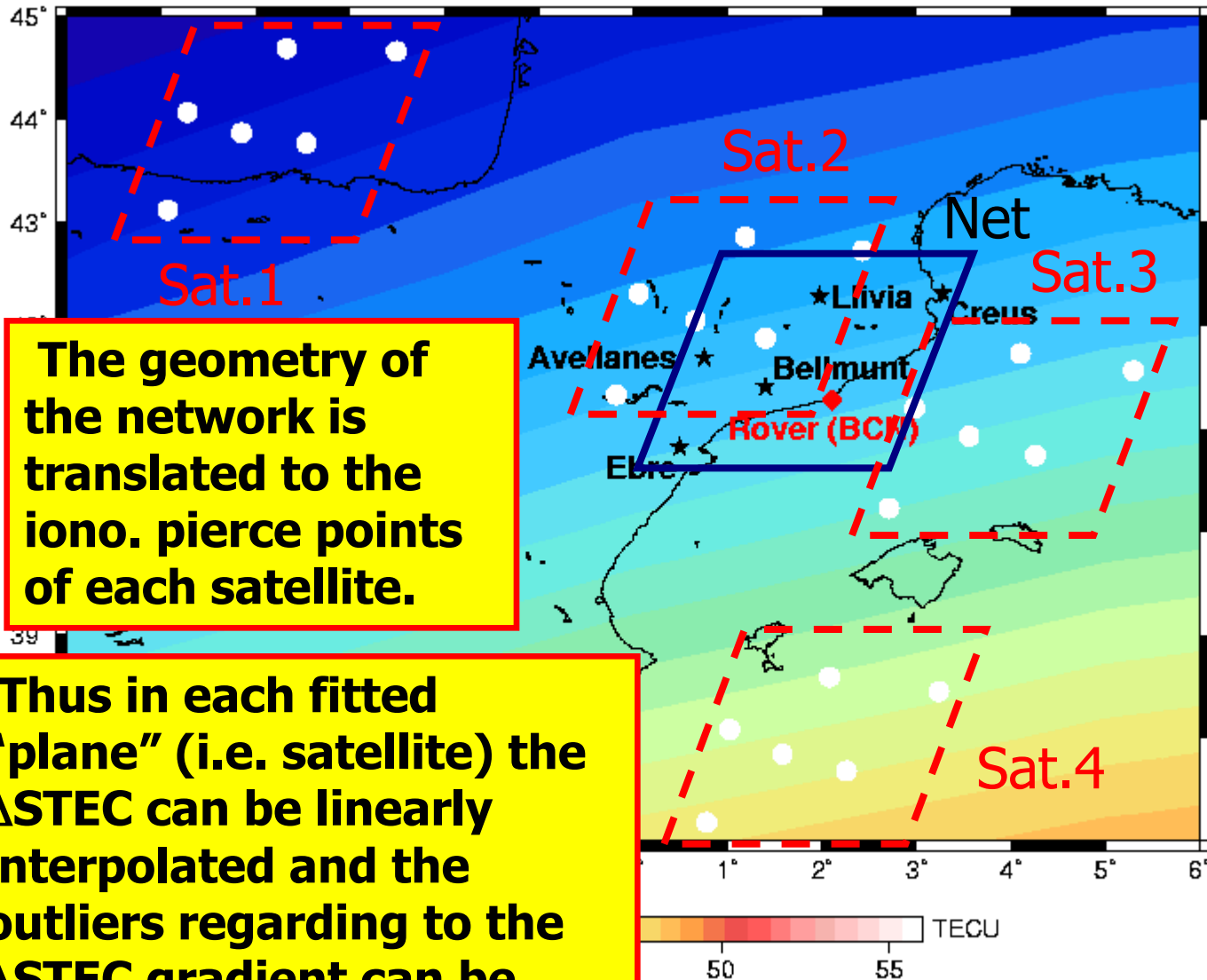




- Ionospheric determination at regional/continental scale

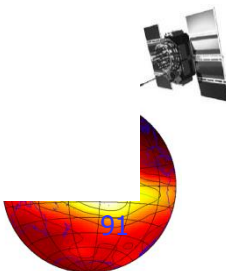


Regional network: Iono. geometry



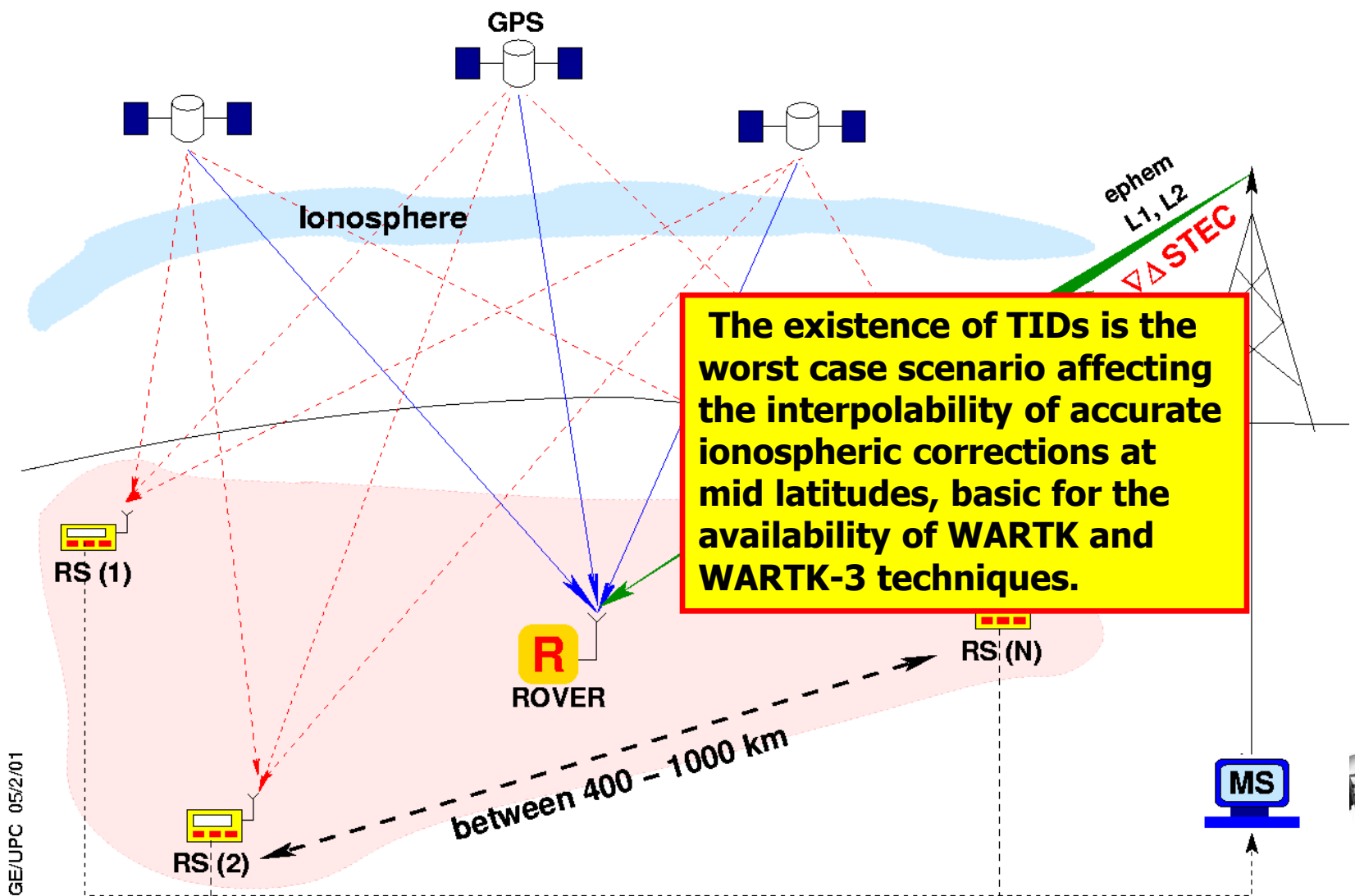
The geometry of the network is translated to the iono. pierce points of each satellite.

Thus in each fitted "plane" (i.e. satellite) the Δ STEC can be linearly interpolated and the outliers regarding to the Δ STEC gradient can be detected.



Wide Area RTK network layout

Long-Baseline (hundreds Km) OTF Ambiguity Resolution

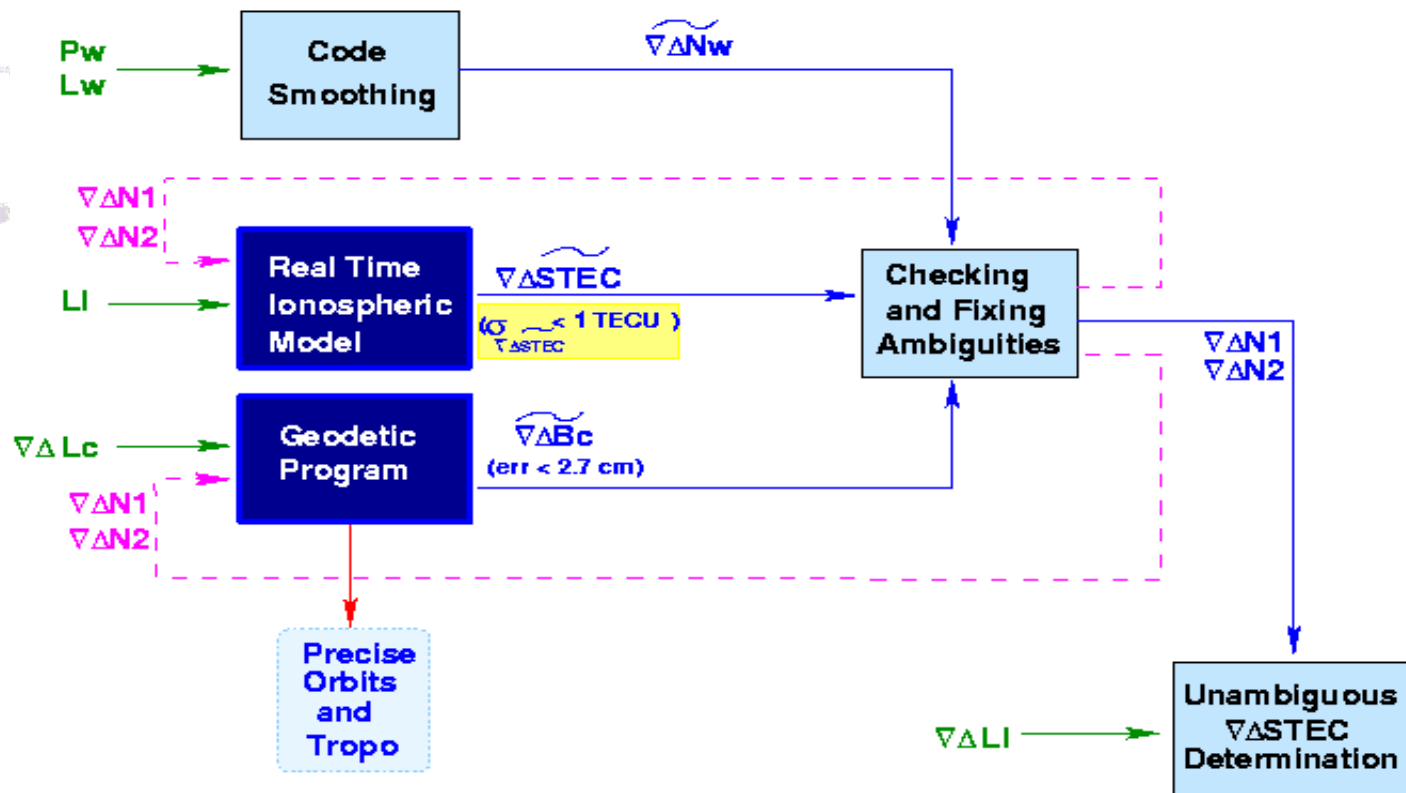


U-GAGE/UPC 05/2/01

WARTK fundamentals

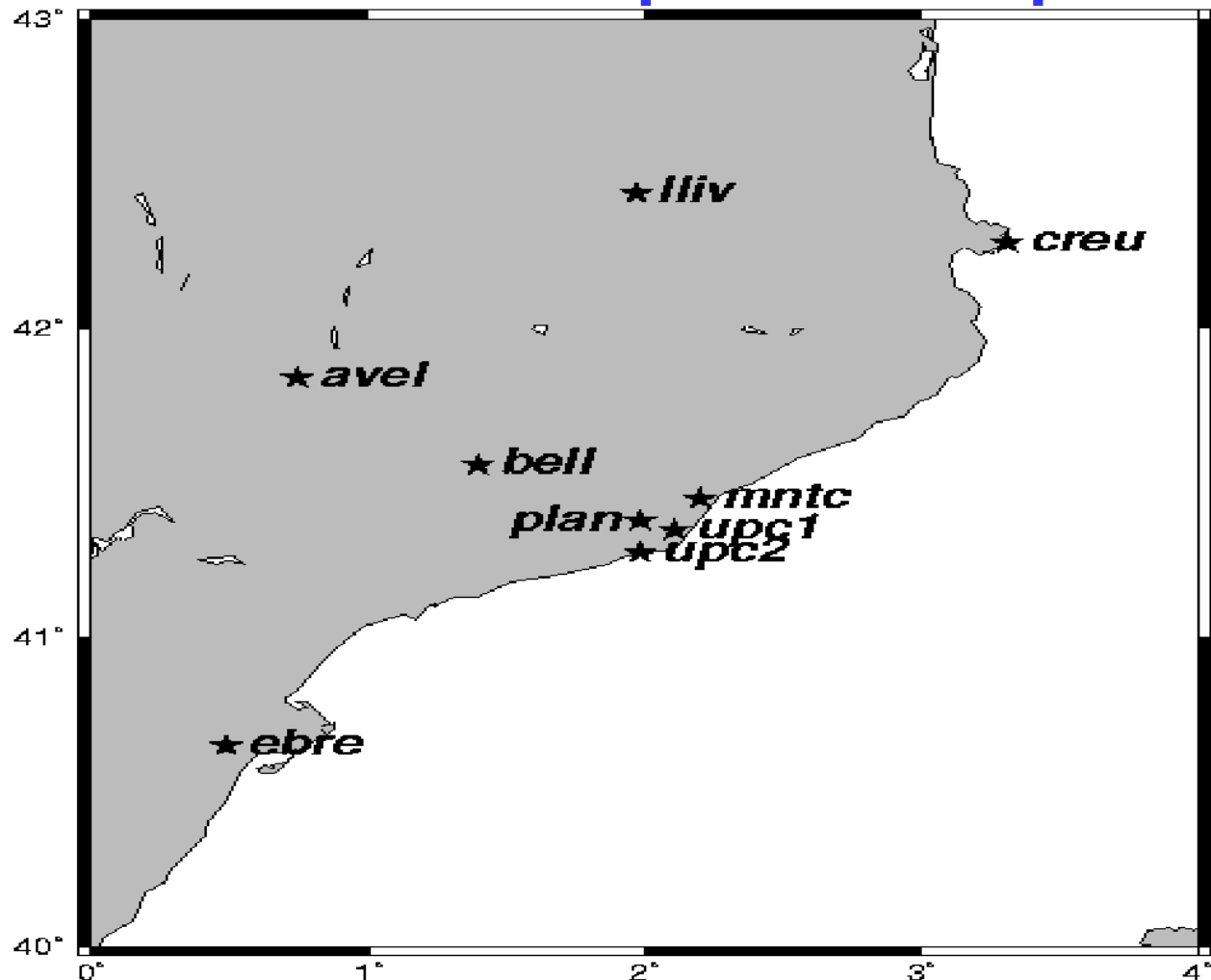
Resolving the Ambiguous $\nabla\Delta$ STEC
in Real Time for the Reference Stations

gAGE/UPC 24/07/01



Both filters, ionospheric and navigation ones, are feed with the reference GNSS stations data, sharing the double differenced carrier phase ambiguity unknowns which can be better estimated and fixed, with two benefits: (1) Improving the geodetic and ionospheric outputs, and (2) Providing very accurate differential ionospheric corrections (at the level of 0.1 TECU of error).

GPS receivers involved in the experiment during 2002 to characterize the main problems in the ionospheric interpolations



Ionospheric interpolation error vs. TIDs occurrence

It can be seen the correlation between poor ionospheric interpolation performance and the occurrence of TIDs during 2002

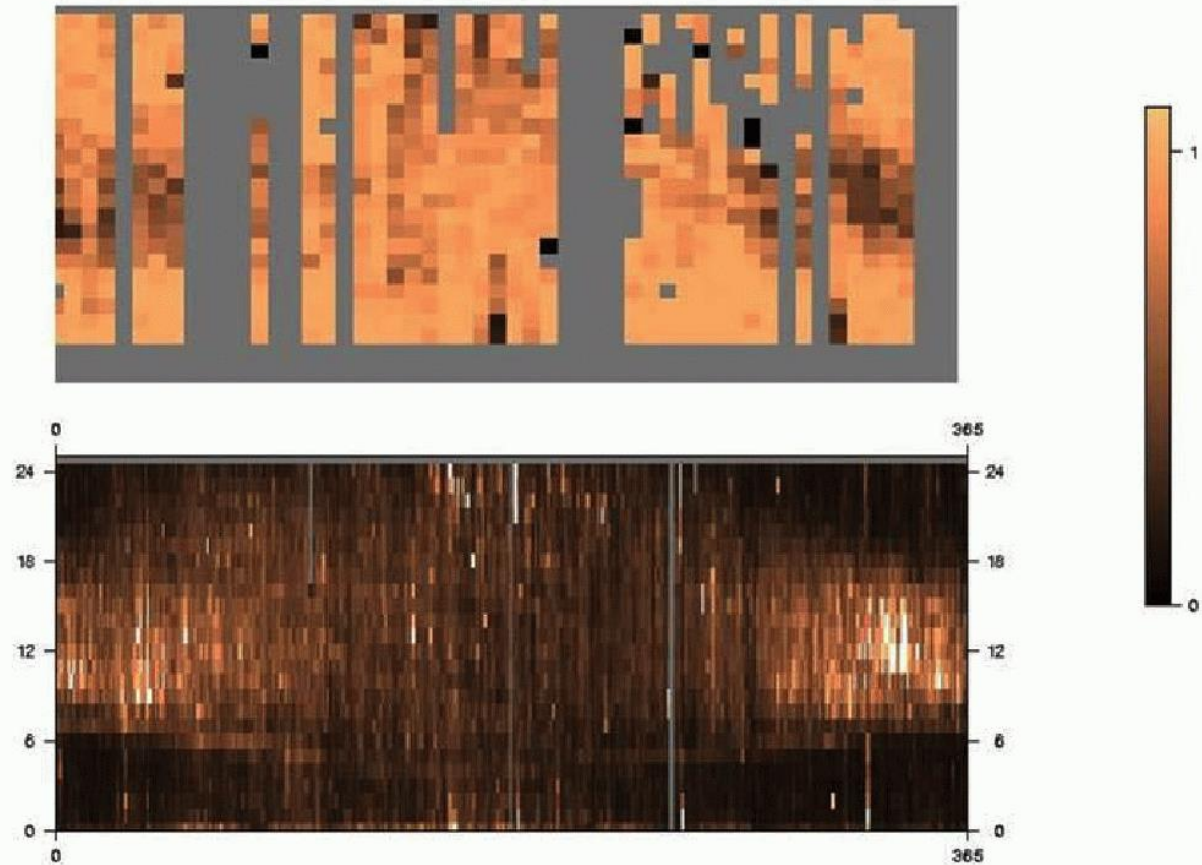
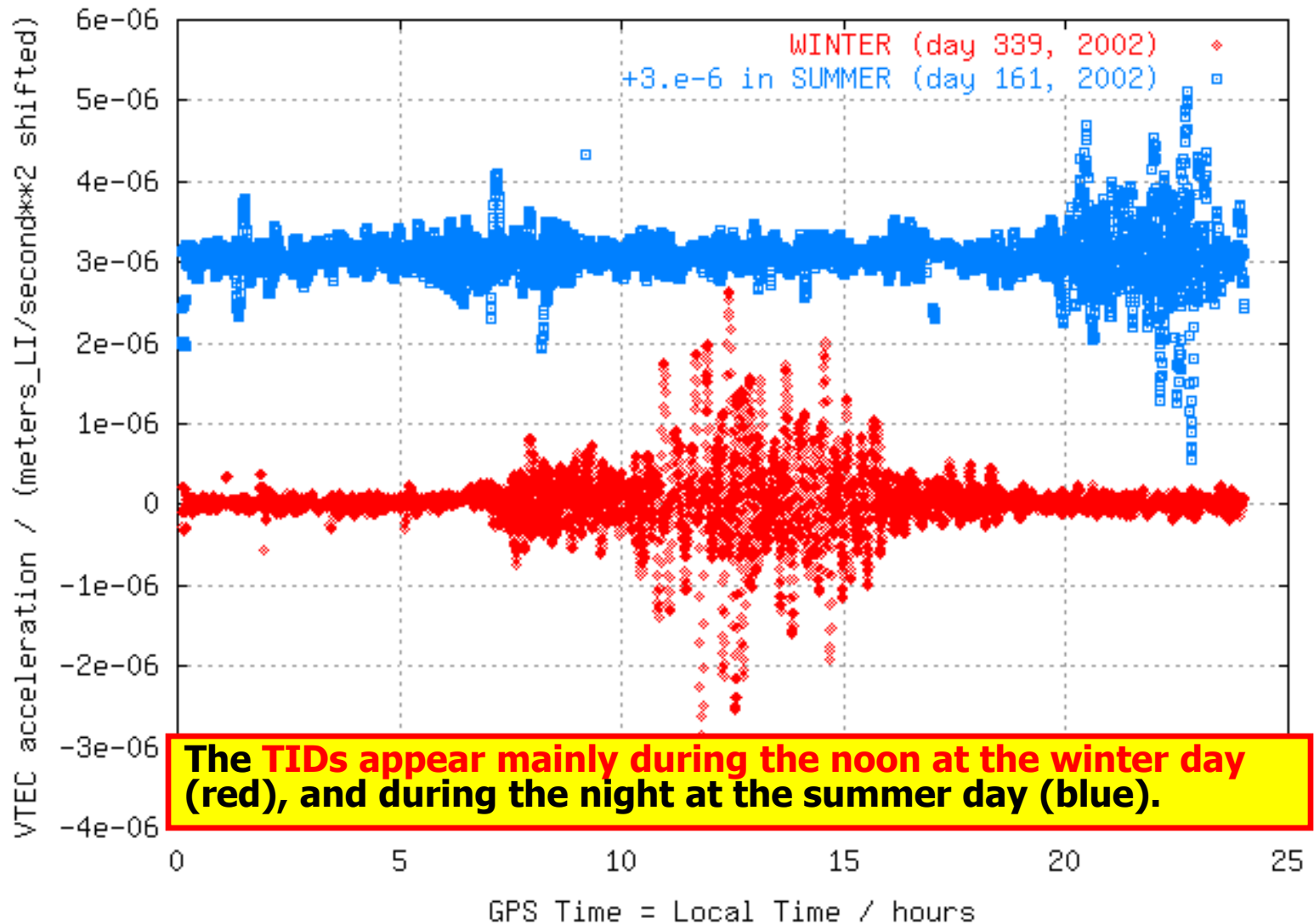


Figure 5: Plot representing the STEC interpolation results at bell (up, extracted from Figure 4), compared with the presence of TIDs (down), computed from **ebre** station VTEC drift rate during 2002 (see details of this computation in Section 3).

Example of TIDs during Winter and Summer

Winter vs. Summer TIDs for Solar Cycle Max. at EBRE (0,41)deg.



The TIDs appear mainly during the noon at the winter day (red), and during the night at the summer day (blue).

TIDs occurrence since the CATNET setup (end 1996)

The TIDs occurrence since end of 1996 repeats the seasonal dependence observed during 2002, modulated by the Solar Cycle (terminator origin).

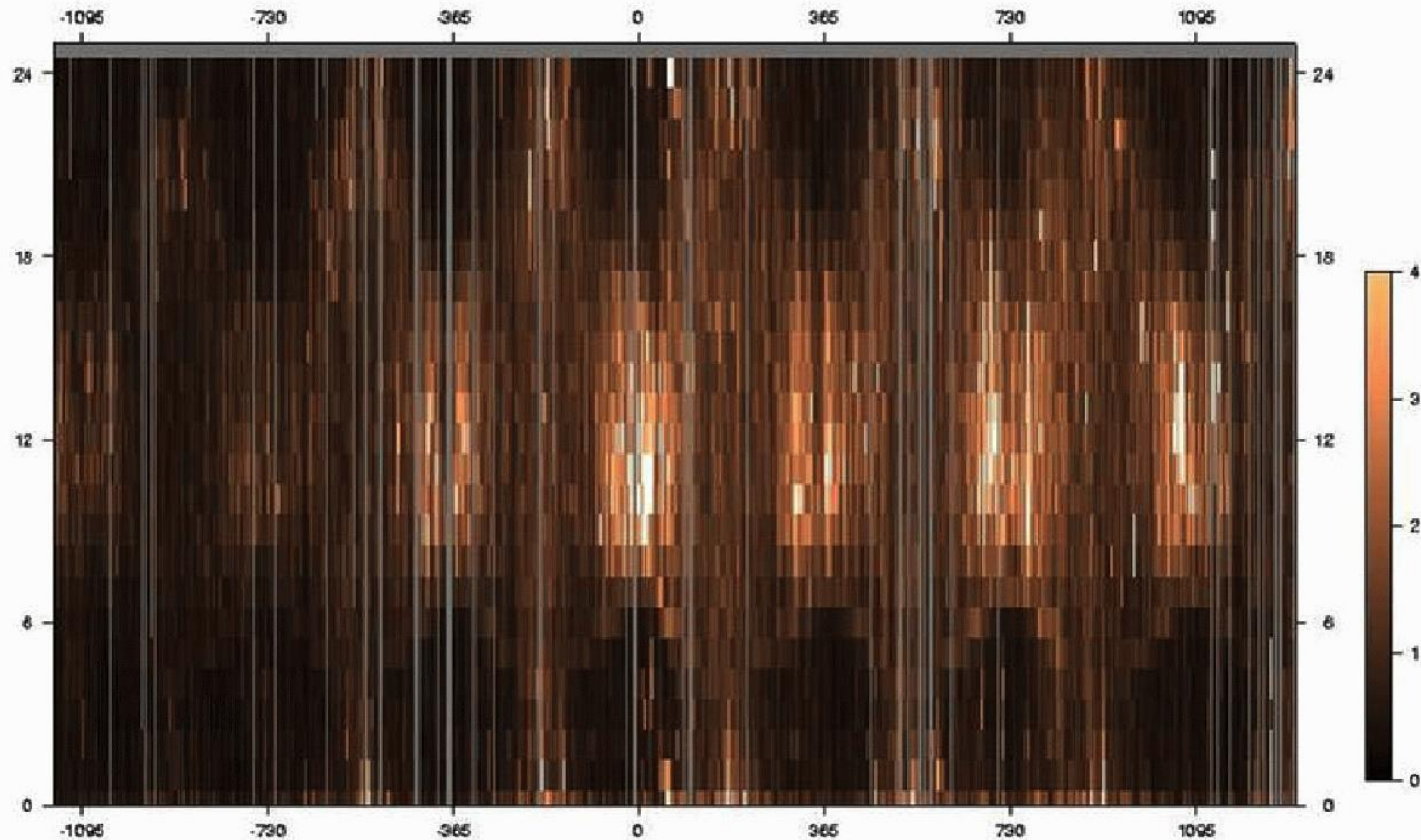


Figure 6: Plot representing the activity's index in TIDs (multiplied by $1e+7$). The X axis represents the day of year referred to January 1st, 2000. It starts at the end of the year 1996, (the changes of year coincide with the maximum number of the TIDs, at noon time). Again the Y axis represents the Universal Time at ebre (approximately coinciding with the local time).

Solution: modelling the TIDs from the reference stations network.



- Determining the electron content at global scale

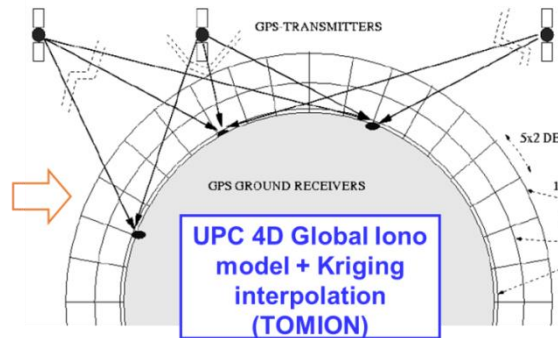
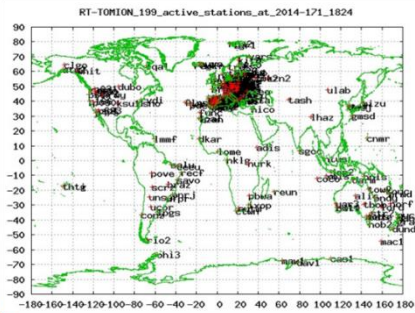
How?: RT-TOMION computing global VTEC maps based on GNSS data

From each obs.
we get one STEC
value: $V = S/M = (L_i - B_i)/M$. [~1500 val. /
30 sec]

Interpol. [*]
by Splines

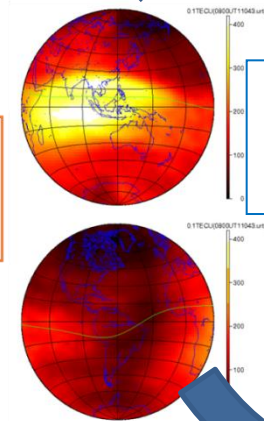
Kriging [*]
Interpolation

RT IGS ground GPS data
(70 to 195 worldwide receivers)



$$L_I = S + B_I \approx \sum_{i=1}^n N_{e,i} \Delta l_i + B_I$$

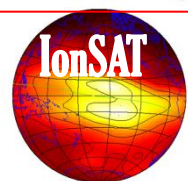
UPC
global
VTEC
maps



[*] Performed in Solar-Magnetic Local Time and Latitude coordinates

Layout summarizing the global VTEC computation from ground GPS data by means of the UPC TOMION software, including the main tomographic equation solved for (data: ionospheric combination of carrier phases L_I , and length intersection within each voxel, Δl_i ; unknowns: its ambiguity B_I , the STEC, S , which includes the mean electron density within each given voxel, $N_{e,i}$).

**New RT global
VTEC maps**



Coordinates to interpolate

The **ionospheric electron content distribution** is closely tied to:

(1) The Sun position, as the main ionizing source in typical geomagnetically quiet conditions.

(2) The geomagnetic field, which field lines are the typical ways used by the charged particles, like the electrons, which can move describing circles around them due to the Lorentz force.

In this context **the usage of a reference system to locate the ionospheric pierce points, where the electron content could be considered as much stationary as possible, is an advantage for interpolation**, allowing the combination of VTEC values during a more extended period of time.

The following type of coordinates goes from **less**, **mid** up to **high** stationarity:

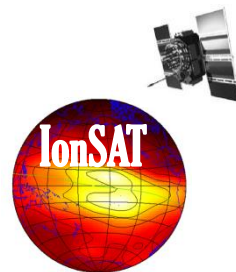
Longitude, Latitude => **Local Time**, Latitude => **Solar-Magnetic Local Time**, **Magnetic Latitude**

From Longitude to Local Time

The transformation from Longitude (LON) to Local Time (LT) can be approximated by the following simple expression (assuming a “mean” Sun):

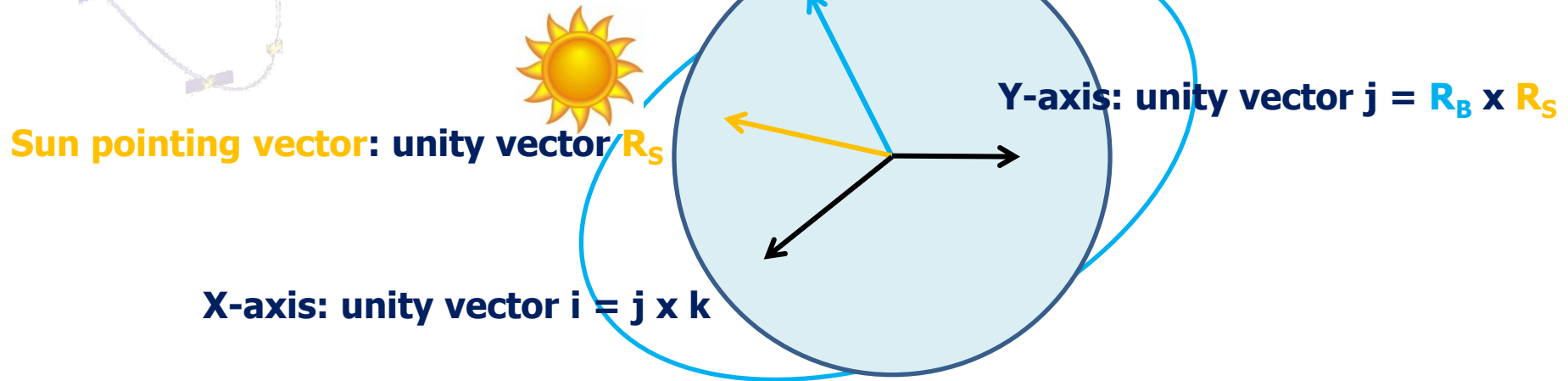
$$LT = UT + LON$$

where UT is the Universal Time.



Solar Geomagnetic coordinates

Z-axis: Boreal Magnetic Pole (from dipolar model), unity vector $\mathbf{k} = \mathbf{R}_B$



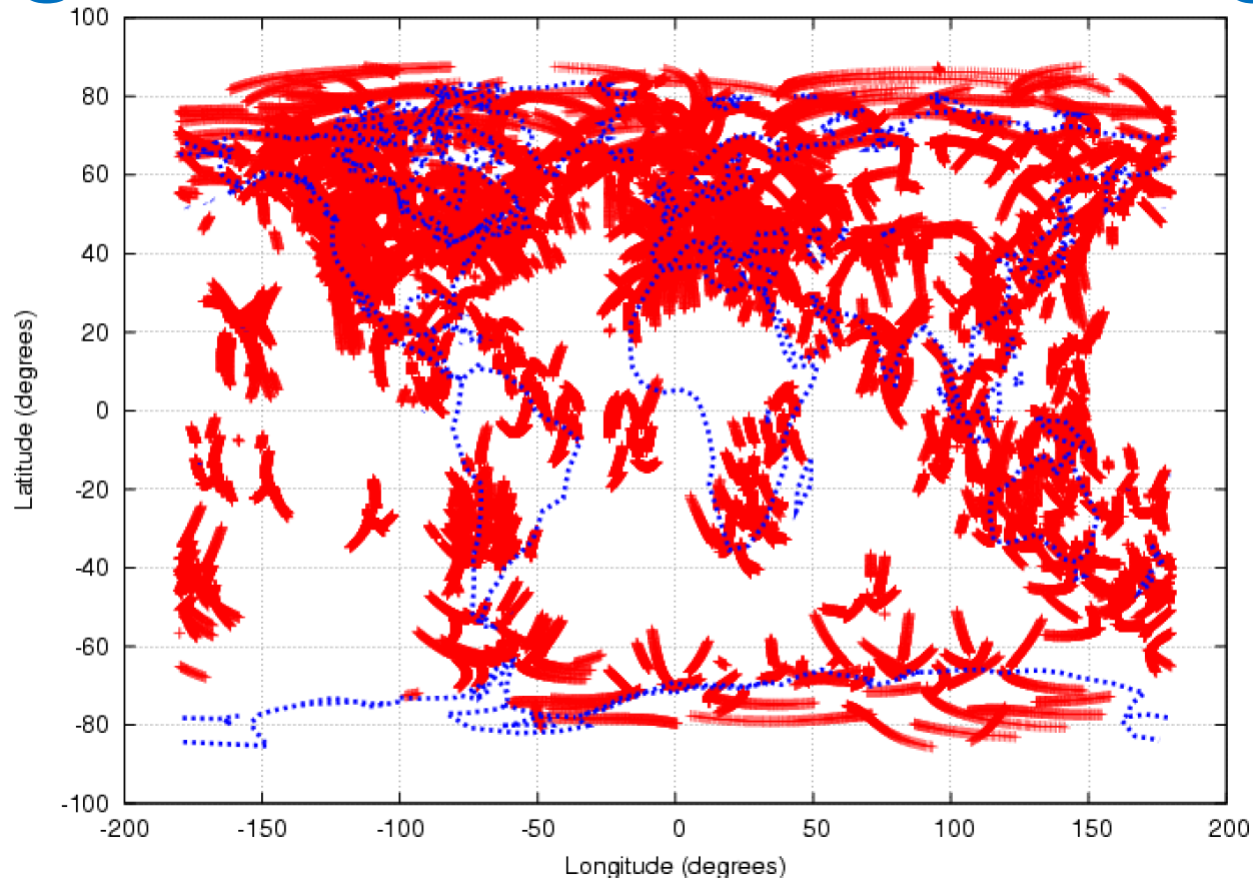
From the projection of the unity vector pointing to the ionospheric pierce point, $\mathbf{R}$, on the new axis, $\mathbf{i}, \mathbf{j}, \mathbf{k}$, we can retrieve the Solar-Magnetic Local Time (SMLT) and Magnetic Latitude, MLAT:

$$\mathbf{R} \cdot \mathbf{k} = \sin(\text{MLAT})$$

$$\mathbf{R} \cdot \mathbf{j} = \cos(\text{MLAT}) * \sin(\text{SMLT})$$

$$\mathbf{R} \cdot \mathbf{i} = \cos(\text{MLAT}) * \cos(\text{SMLT})$$

Kriging motivation in global VTEC modelling (1)



Distribution of the IPPs over a thin shell of 450 km height for an example day of January 3rd 2003 from 3 to 5 UT.



Kriging motivation in global VTEC modelling (2)

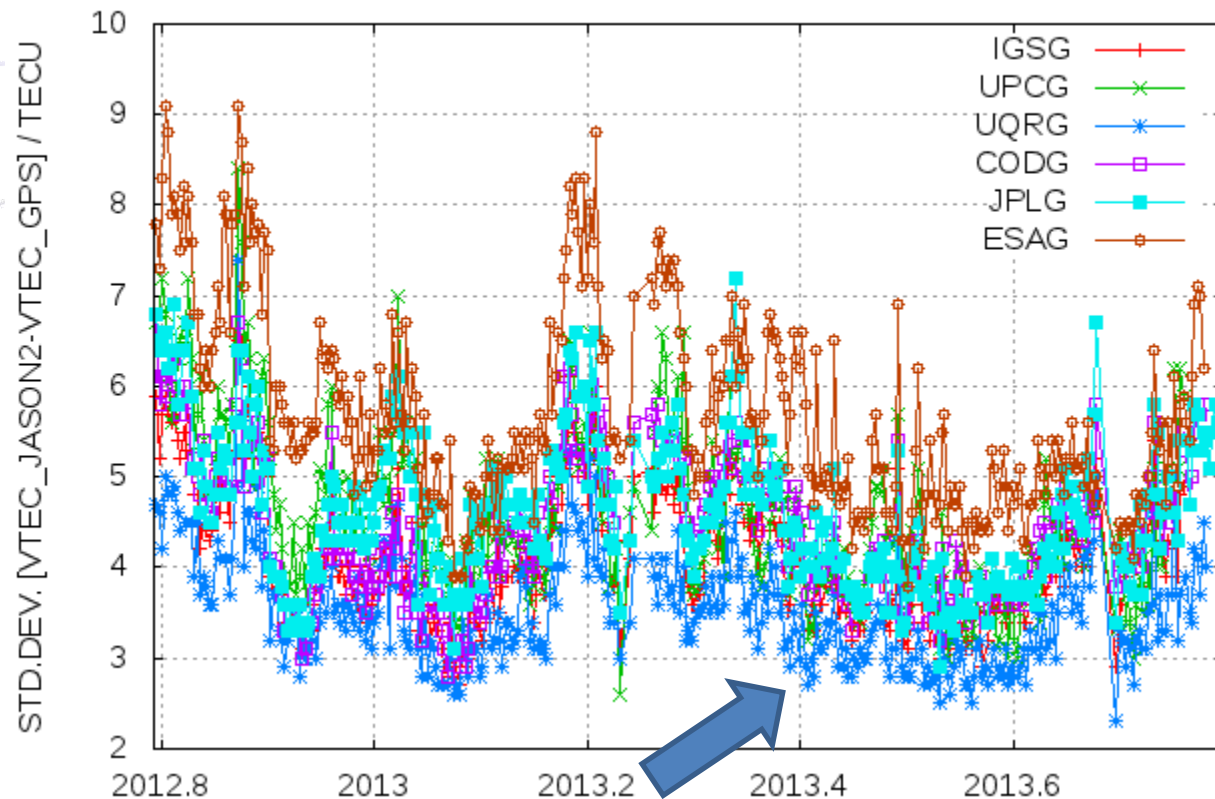
- The large areas without observations makes the method used to interpolate/extrapolate crucial.
- Kriging[*] is a Gaussian Process regression: weights (that combine the observed values for the process of interpolation / extrapolation) are computed by consistency with the expected error decorrelation vs. distance. This improves the results...

[*] We apply “Ordinary Kriging” on the residuals (it can be considered a sort of extended “Universal Kriging”).



Kriging motivation in global VTEC modelling (3)

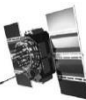
Worldwide JASON2 observations



Std.Dev. [JASON2 VTEC – GNSS VTEC] during one year (worldwide).
New rapid UPC global VTEC with Kriging (**UQRG**) works systematically better than all final IGS VTEC maps, including “classic” UPC product for 1 year-

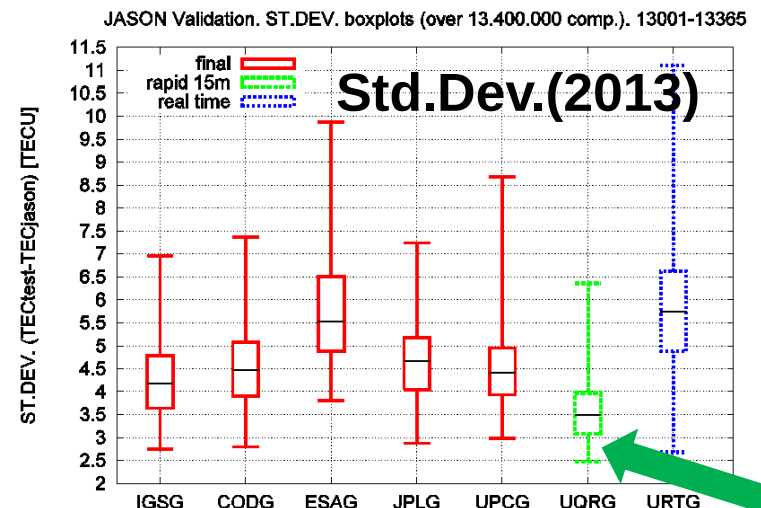
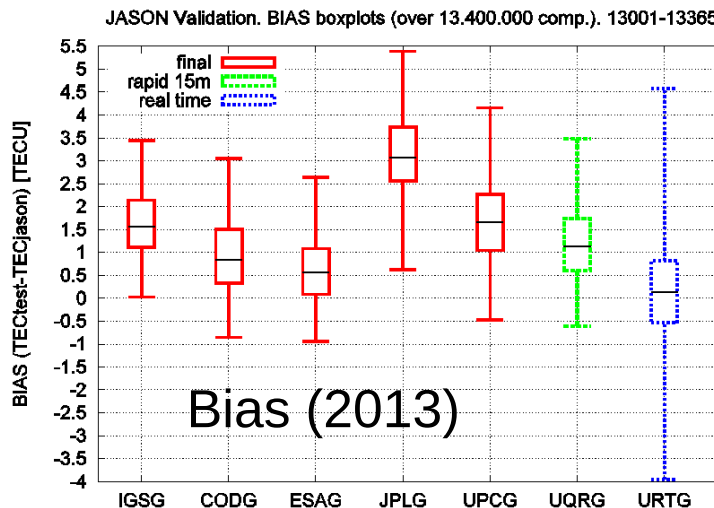
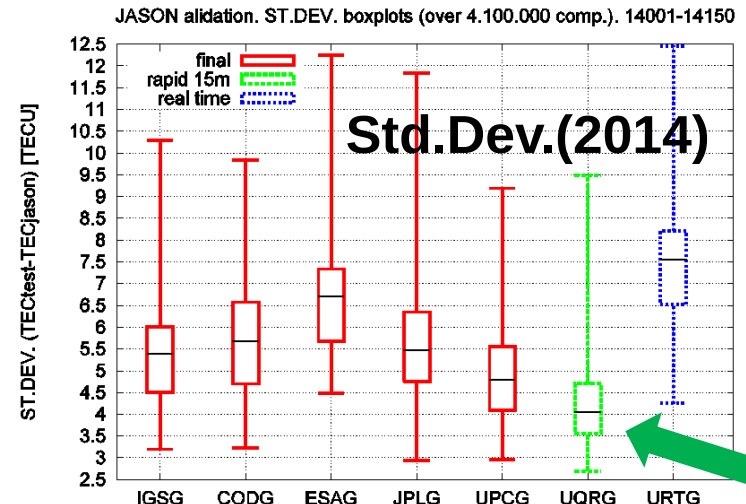
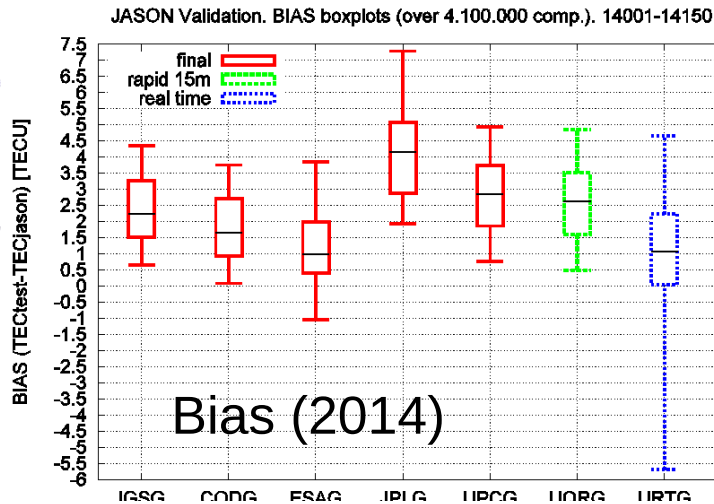
Kriging motivation in global VTEC modelling (4)

- The better behaviour of new UPC global VTEC maps also during the last year confirms the previous comparisons with other models, beyond those of IGS ionospheric analysis centers:
 - Feltens, J., M. Angling, N. Jackson-Booth, N. Jakowski, M. Hoque, M. Hernández-Pajares, A. Aragón-Ángel, R. Orús, and R. Zandbergen (2011), Comparative testing of four ionospheric models driven with GPS measurements, Radio Sci., 46, RS0D12, doi:10.1029/2010RS004584.
 - Gulyaeva T.L., F. Arikan, M. Hernandez-Pajares, I. Stanislawska (2013), GIM-TEC adaptive ionospheric weather assessment and forecast system, Journal of Atmospheric and Solar-Terrestrial Physics, 102 (2013) 329–340.



Assessment with Jason-2 (2013.0-2014.4: ~70 rec. in RT)

Bias (left) and standard deviation (right) boxplots of the differences between the test data and the JASON reference data for the periods 001/2013 to 365/2013 and 001/2014 to 150/2014.



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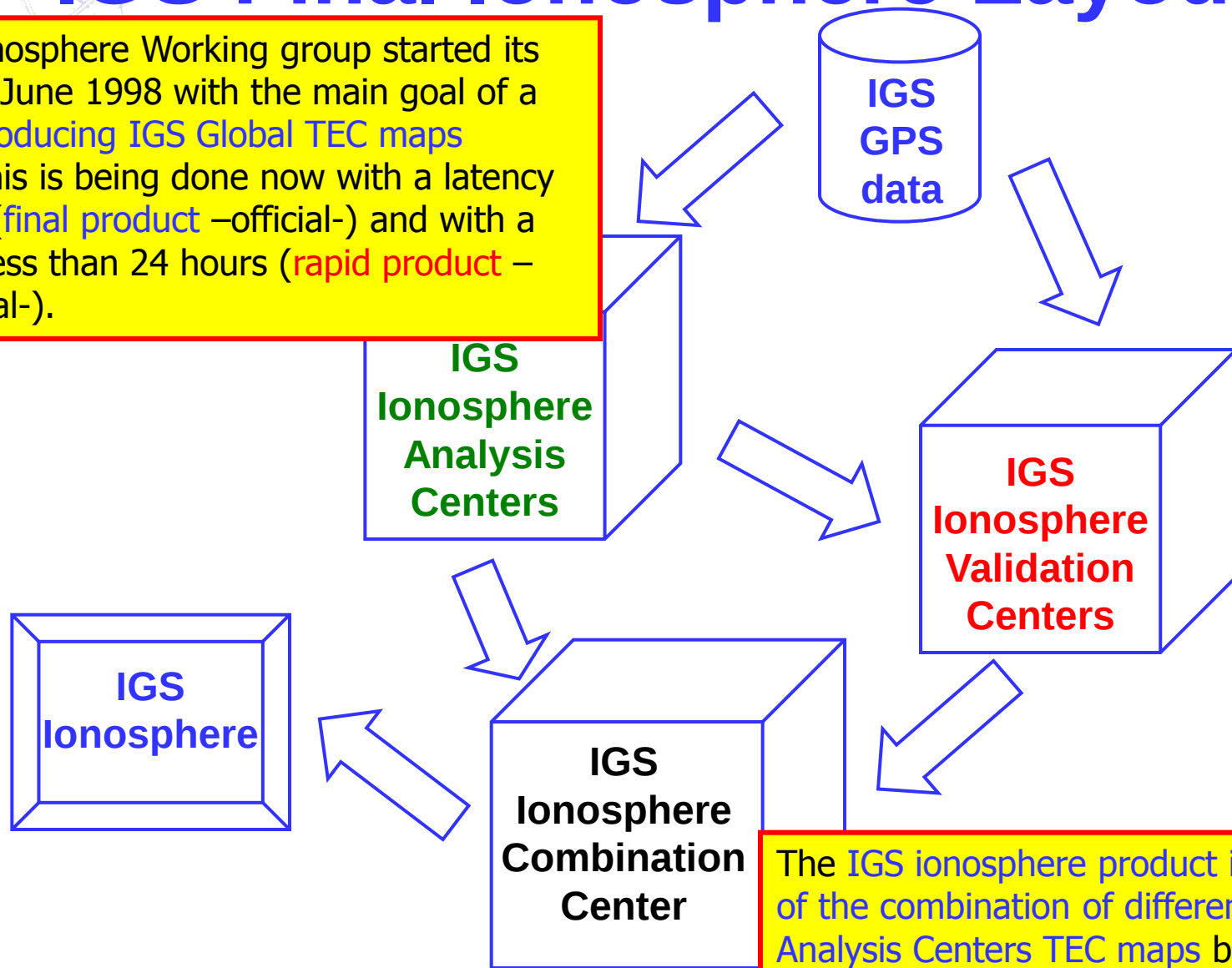
IGS finals

UPC rapid Kriging

107
UPC RT Kriging

IGS Final Ionosphere Layout

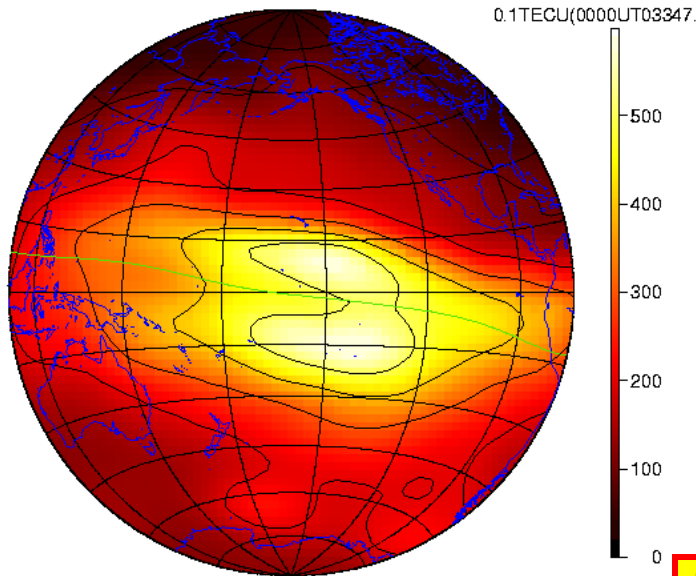
The IGS Ionosphere Working group started its activities in June 1998 with the main goal of a routinely producing IGS Global TEC maps (IGTEC). This is being done now with a latency of 11 days (**final product** –official-) and with a latency of less than 24 hours (**rapid product** –still unofficial-).



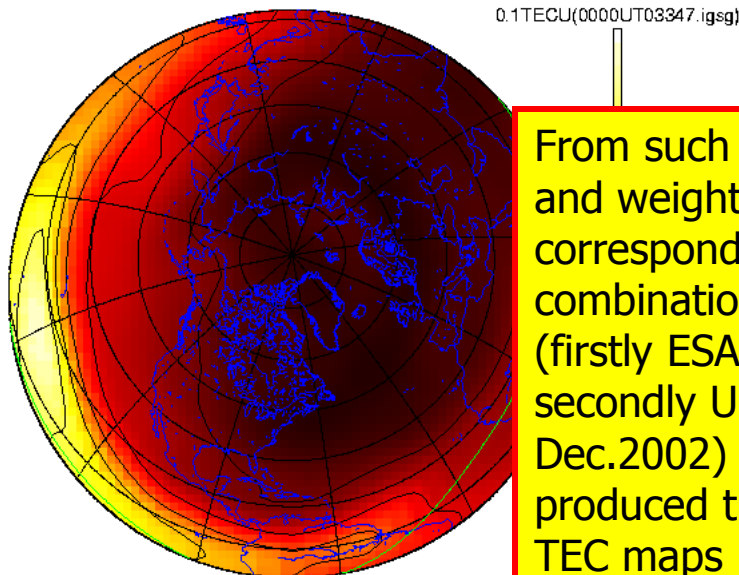
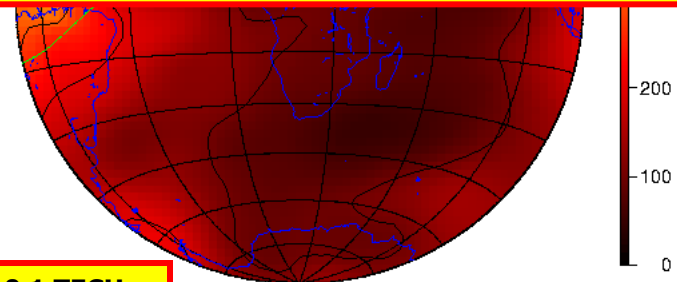
The IGS ionosphere product is a result of the combination of different Analysis Centers TEC maps by using weights computed from GPS data by Validation Centers, in order to get a more accurate product.

Example of IGS Final TEC map: 2003-347-00UT

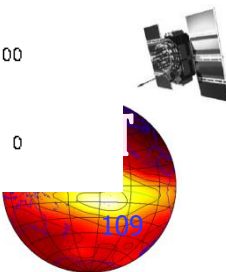
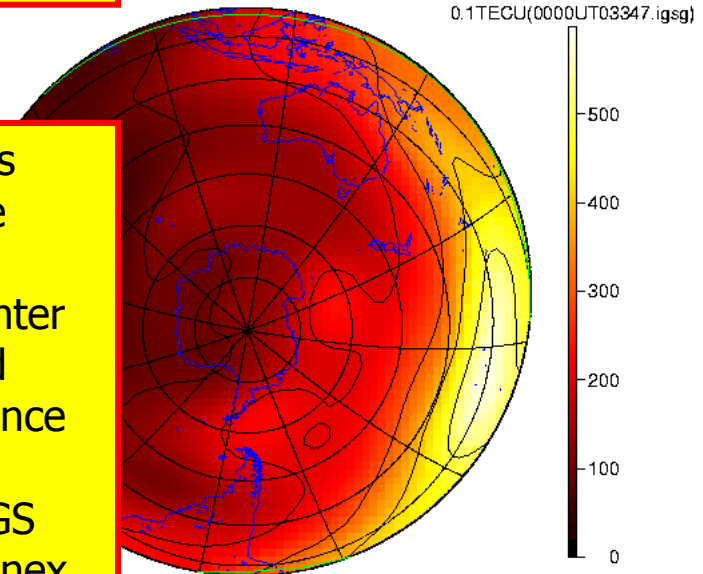
Five Analysis Centers (CODE, ESA, JPL, NRCan and UPC) and 4 Validation Centers (JPL, ESA, NRCan and UPC) have been providing maps (at 2 hours x 5 deg. x 2.5 deg in UT x Lon. x Lat.), weights and external (dual-frequency altimetry-derived) TEC data.



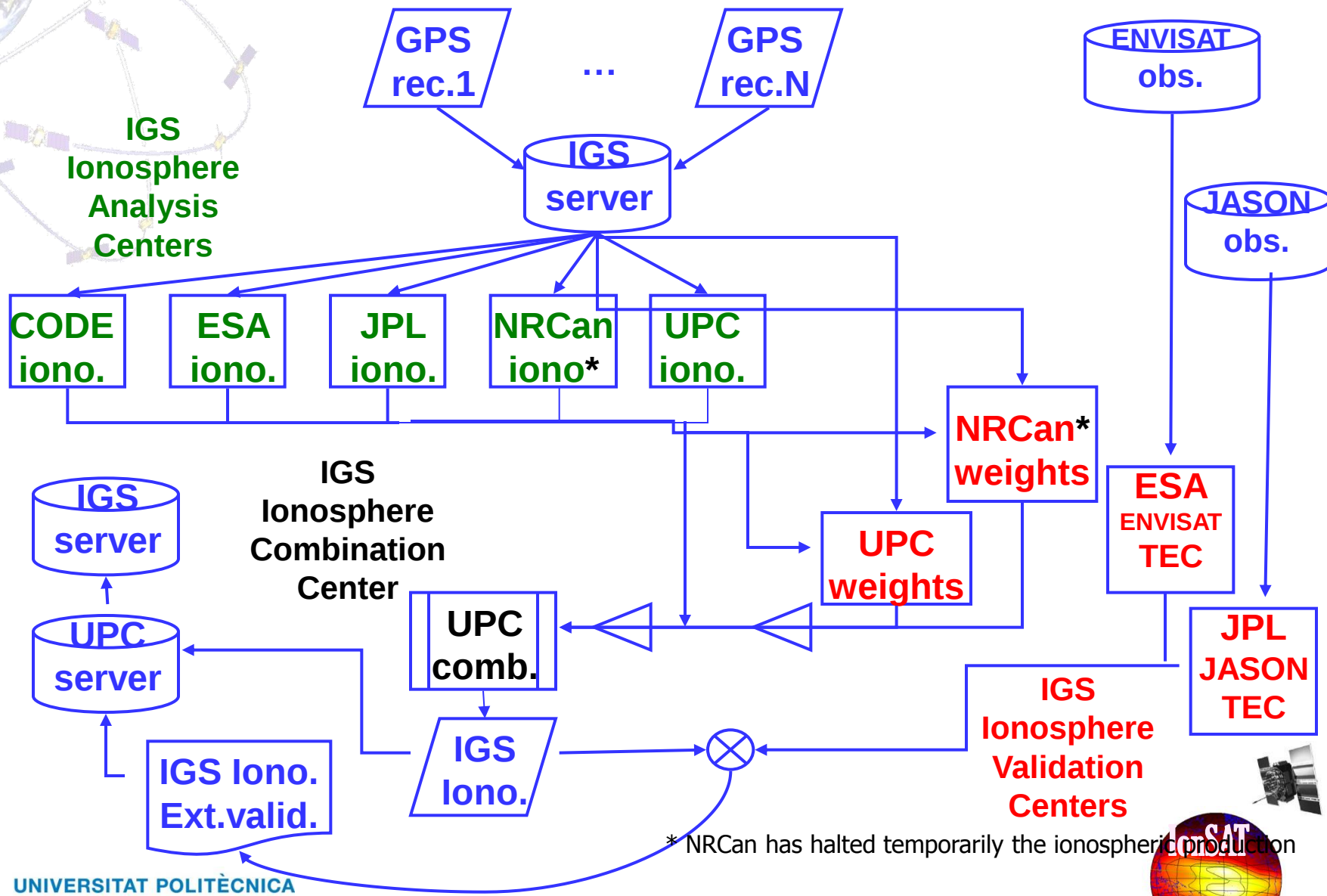
Units: 0.1 TECUs



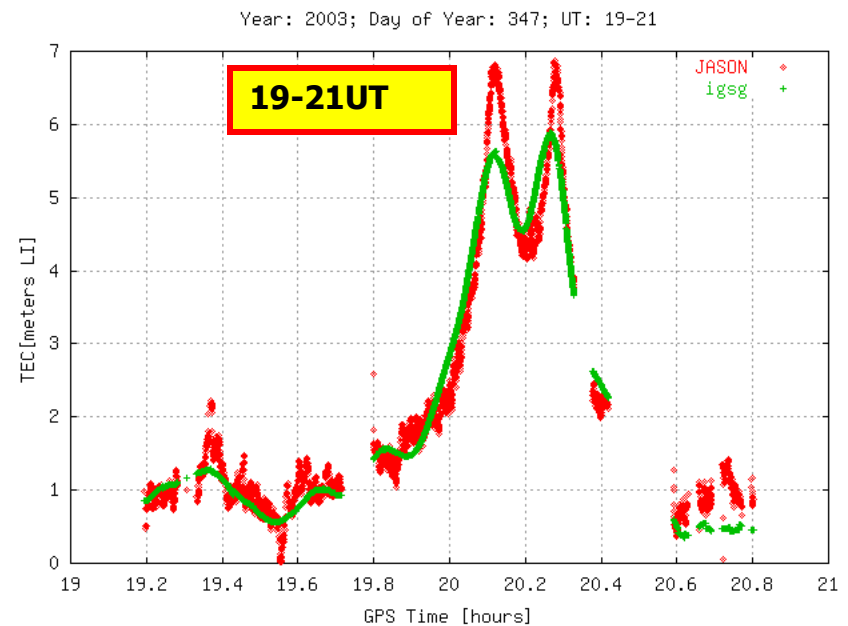
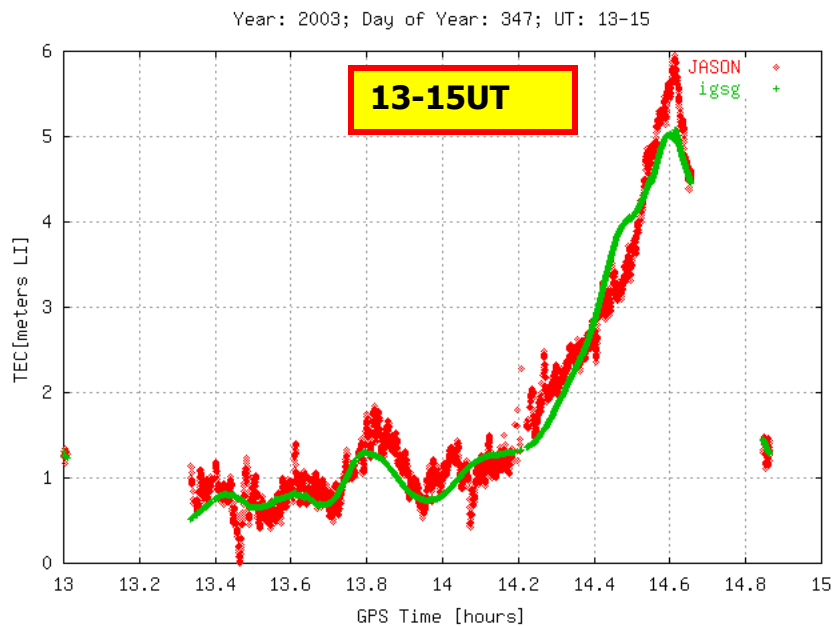
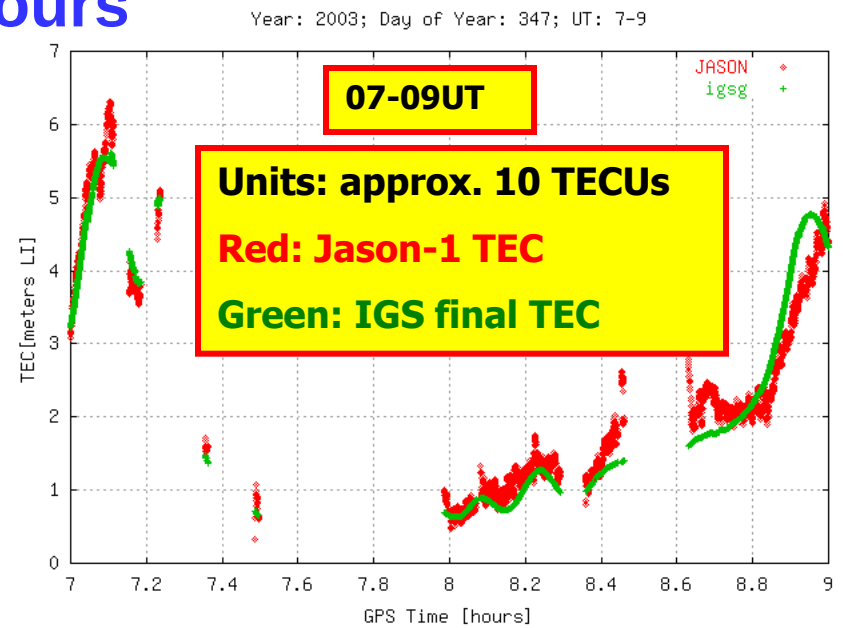
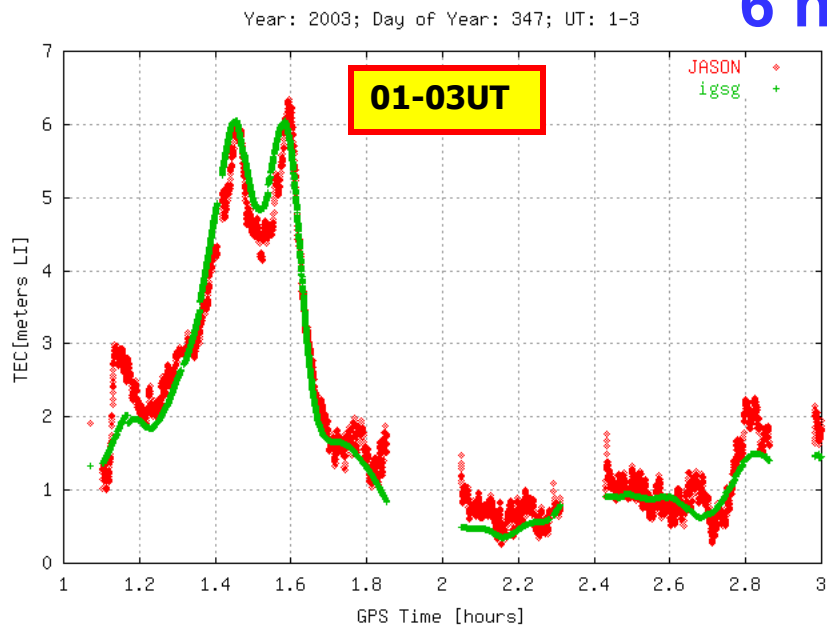
From such maps and weights the corresponding combination center (firstly ESA, and secondly UPC since Dec.2002) has produced the IGS TEC maps in ionex format.



IGS Final Ionosphere Flow-Chart

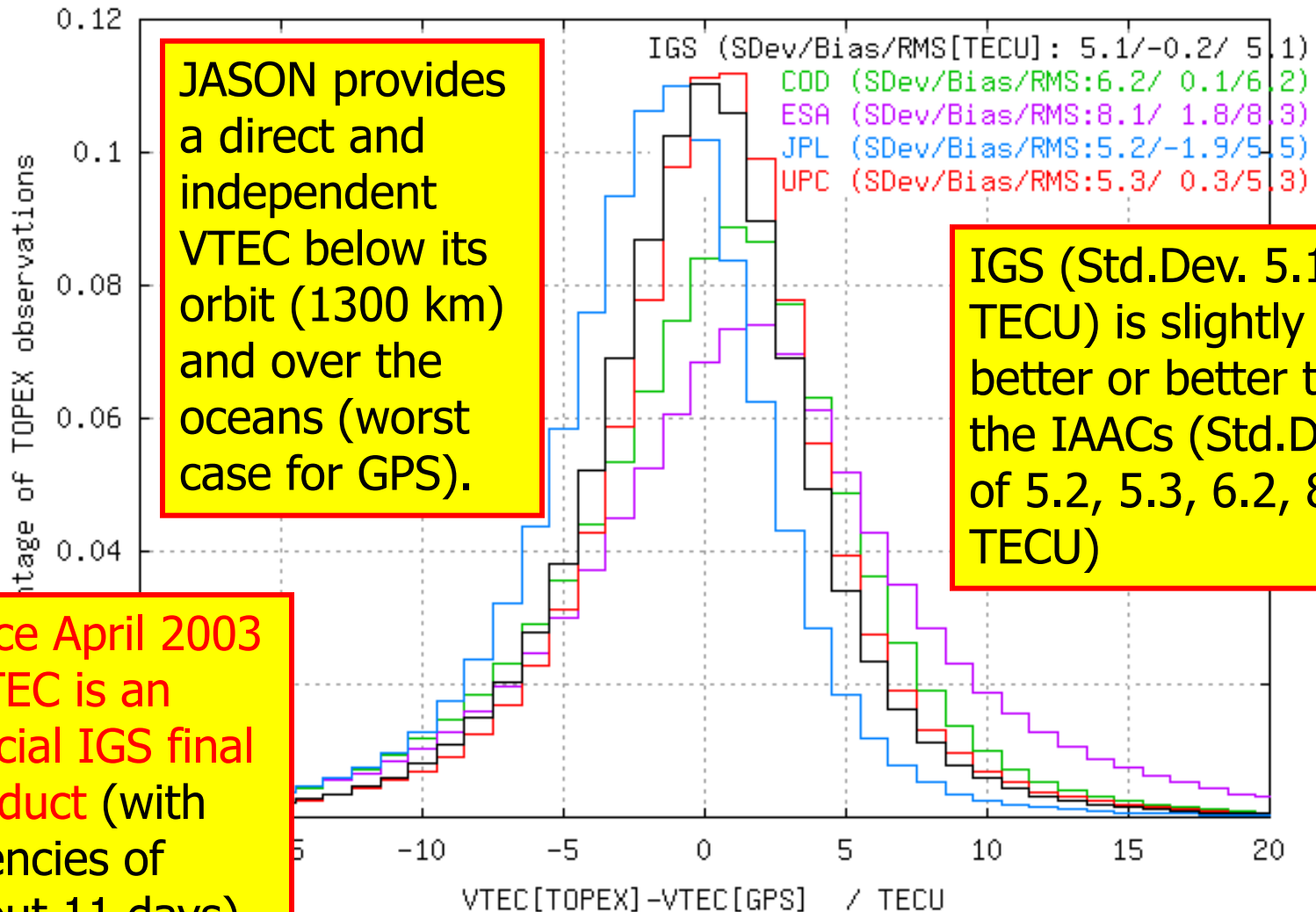


Example of comparison of IGS vs JASON: 2003-347 each 6 hours



Final IGS TEC vs. JASON TEC

GPS vs JASON TEC, 2002 December 15 - 2003 December 13 (14,650,000 obs.)



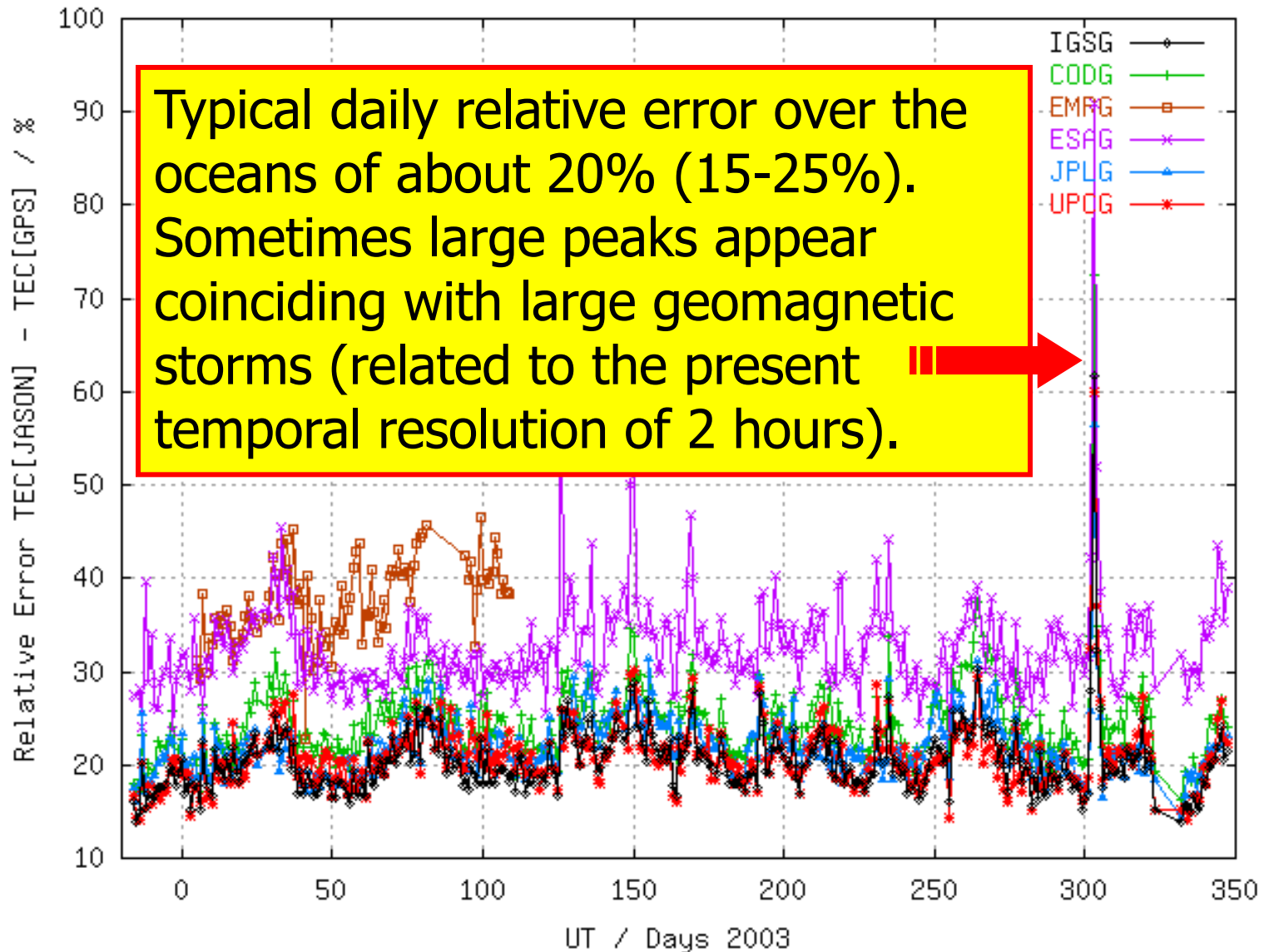
JASON provides a direct and independent VTEC below its orbit (1300 km) and over the oceans (worst case for GPS).

IGS (Std.Dev. 5.1 TECU) is slightly better or better than the IAACs (Std.Dev. of 5.2, 5.3, 6.2, 8.1 TECU)

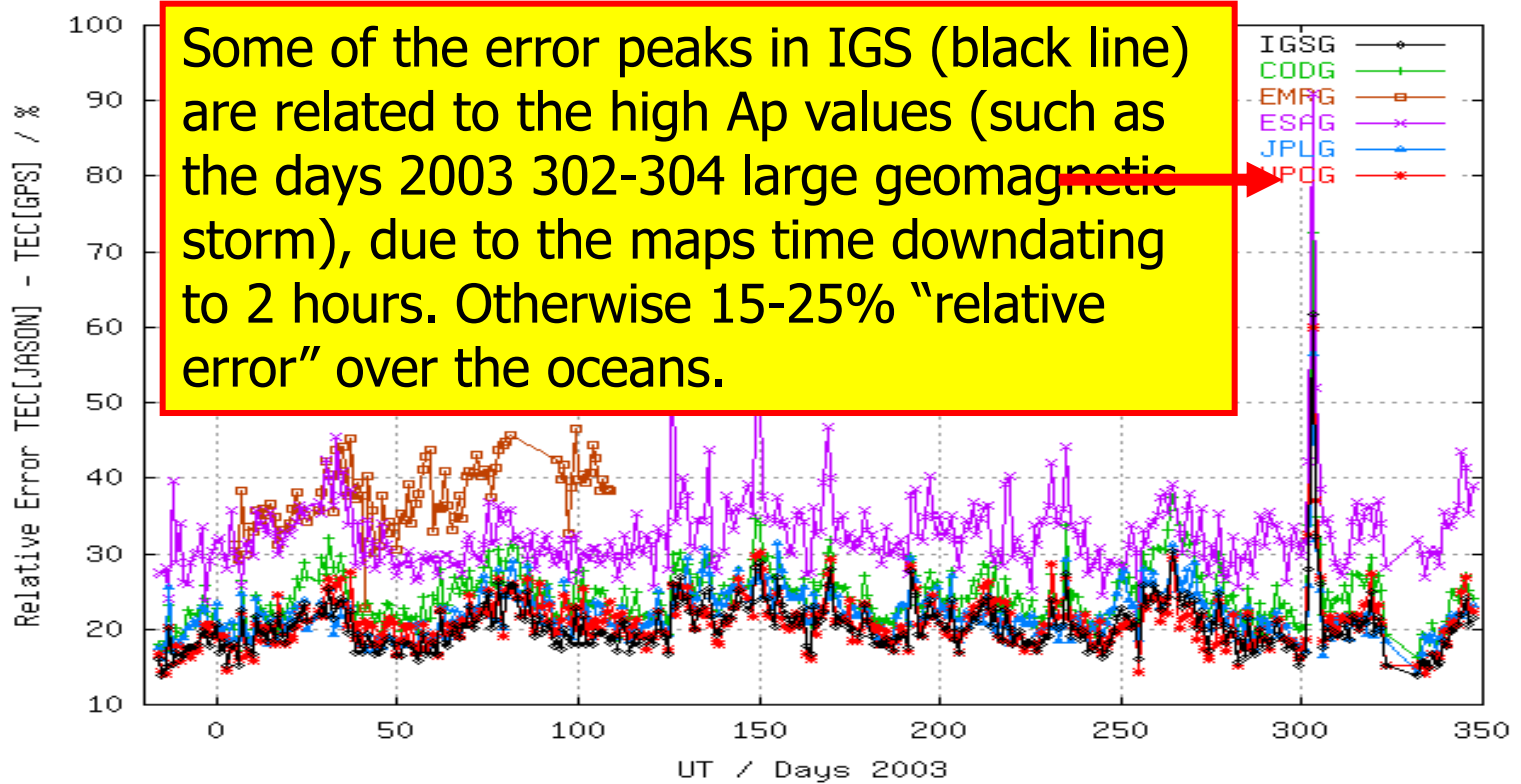
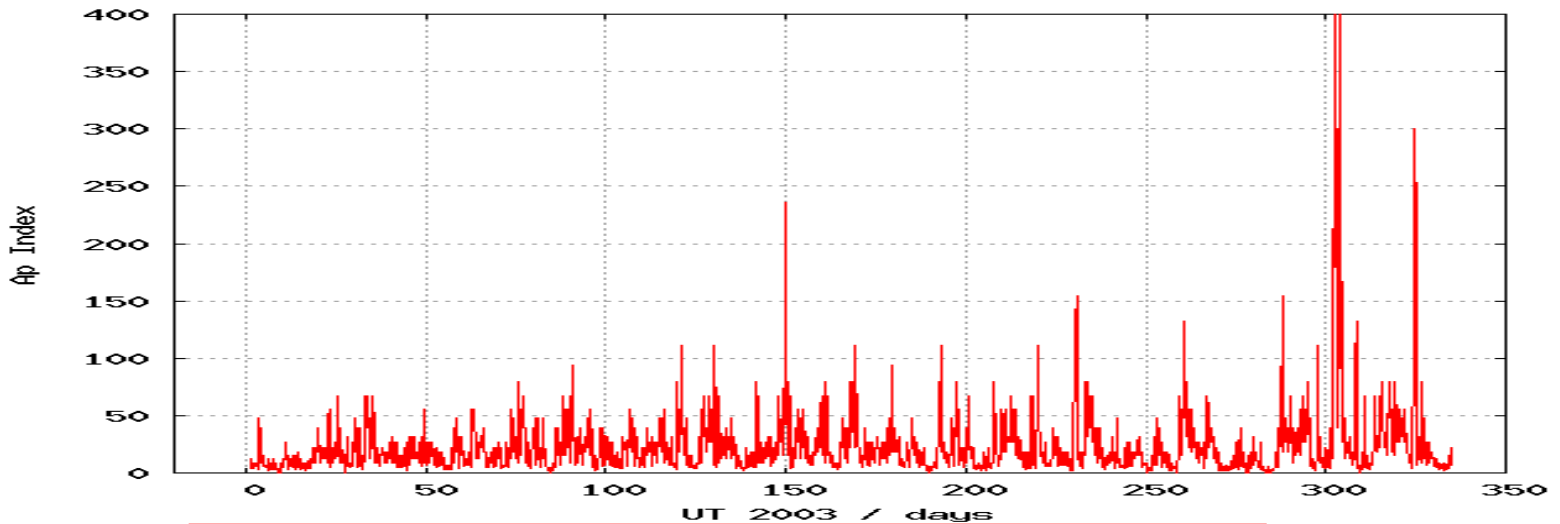
Since April 2003 IGTEC is an official IGS final product (with latencies of about 11 days).



“Relative Error” vs. JASON (daily)

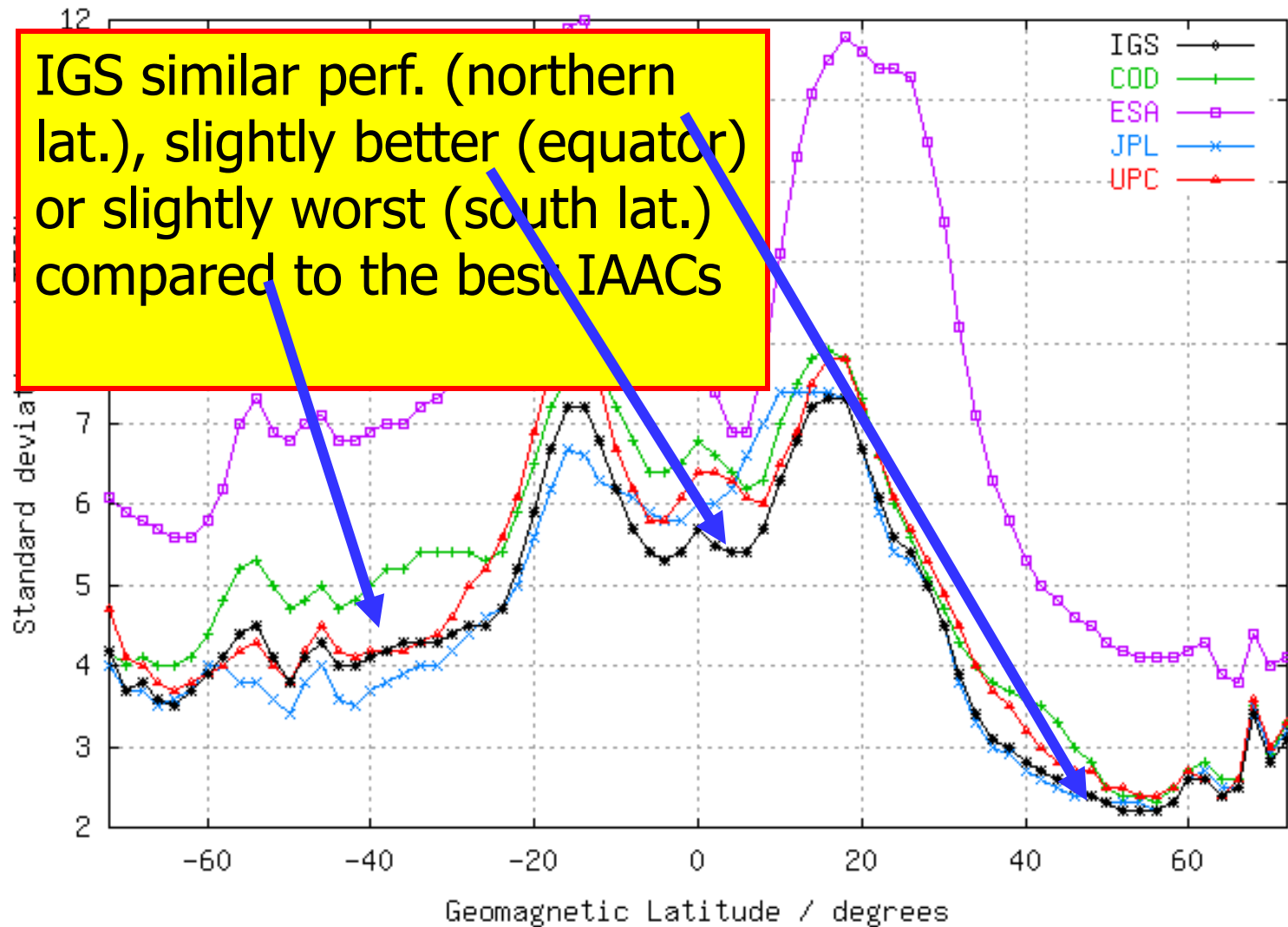


Ap index vs. "Relative Error"



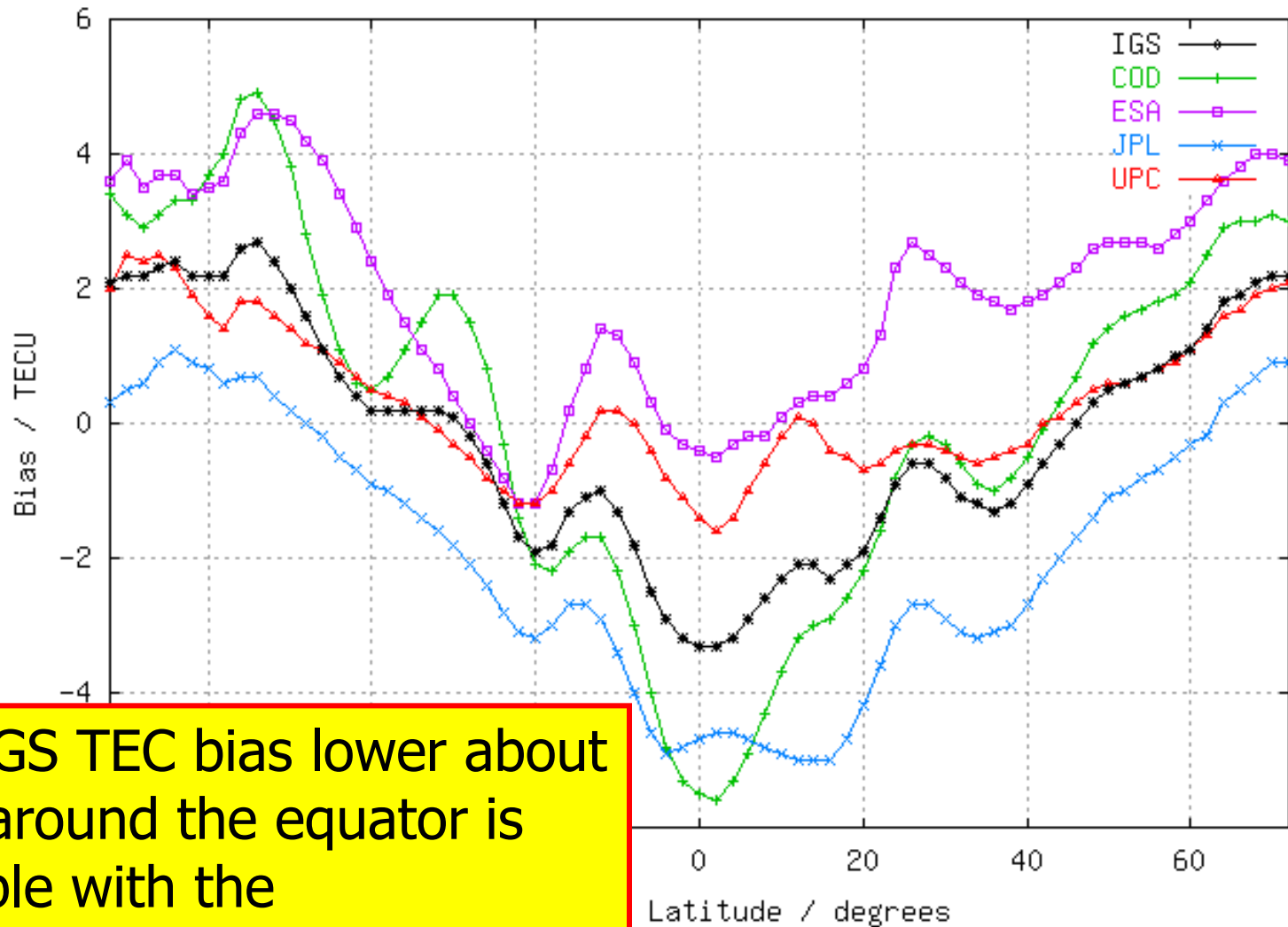
Std.Dev vs. JASON (gmag.lat.)

GPS vs JASON TEC, 2002 December 15 - 2003 December 13 (14,650,000 obs.)

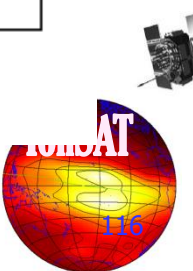


Bias vs. JASON (gmag.lat.)

GPS vs JASON TEC, 2002 December 15 - 2003 December 13 (14,650,000 obs.)

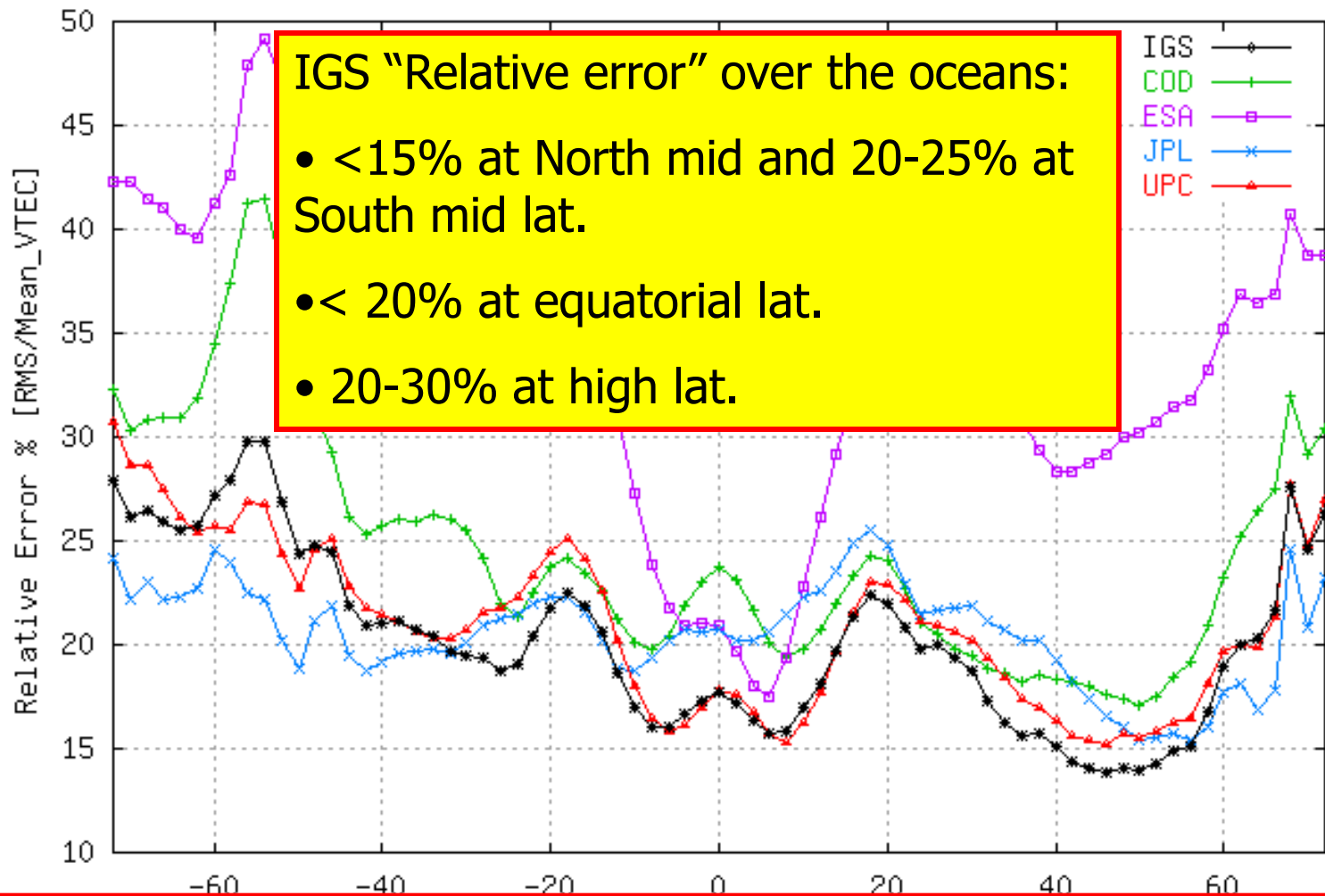


JASON-IGS TEC bias lower about 5 TECU around the equator is compatible with the plasmaspheric component. The absolute bias ref. is still unclear



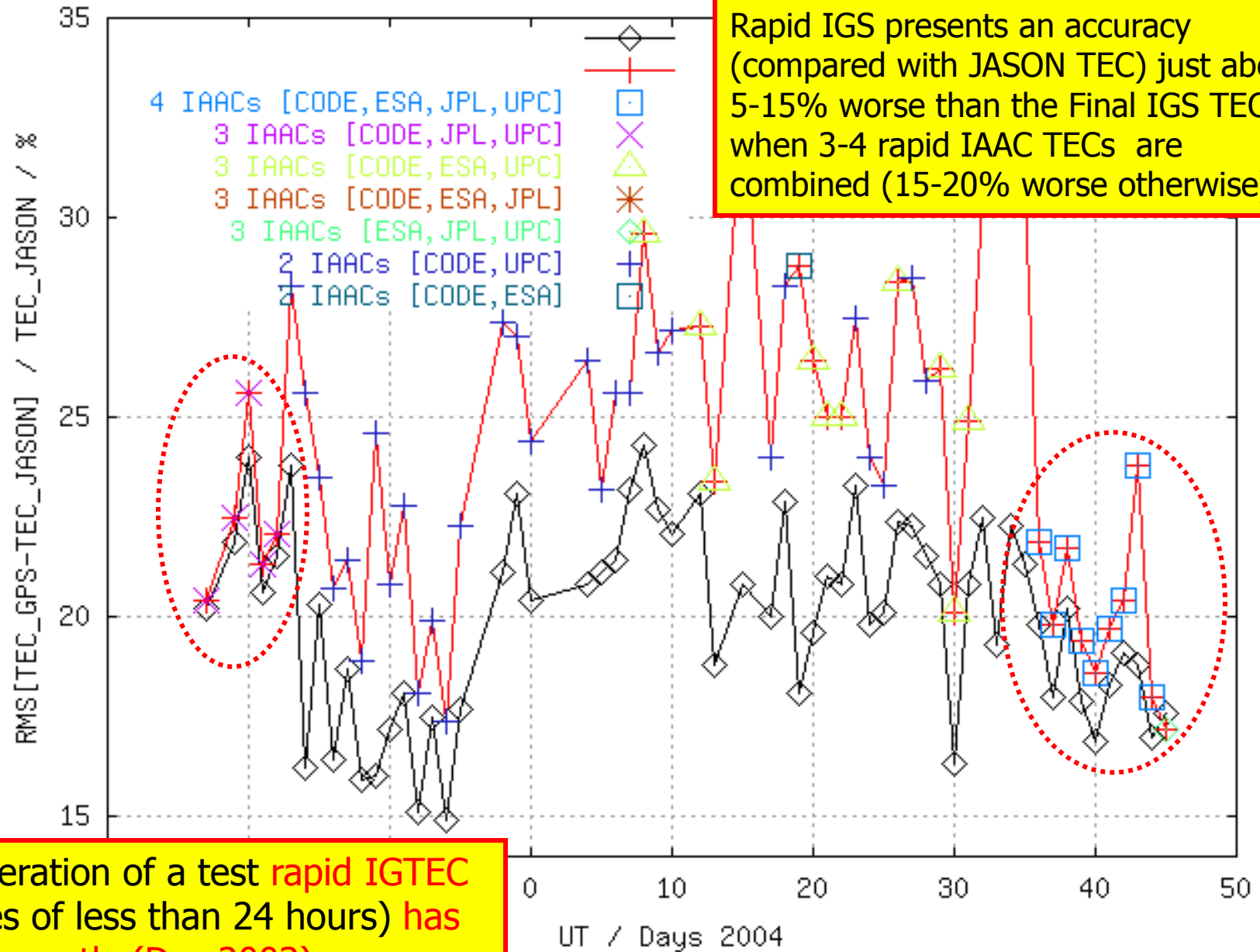
“Rel. error” vs. TOPEX(gmag.lat.)

GPS vs JASON TEC, 2002 December 15 - 2003 December 13 (14,650,000 obs.)



More performance details of the IGS final product (satellite and receiver DCBs, double dif. STEC, latency, with different data scarcity):
http://maite152.upc.es/~ionex3/doc/IGS_IONO_report_April2003_7.pdf

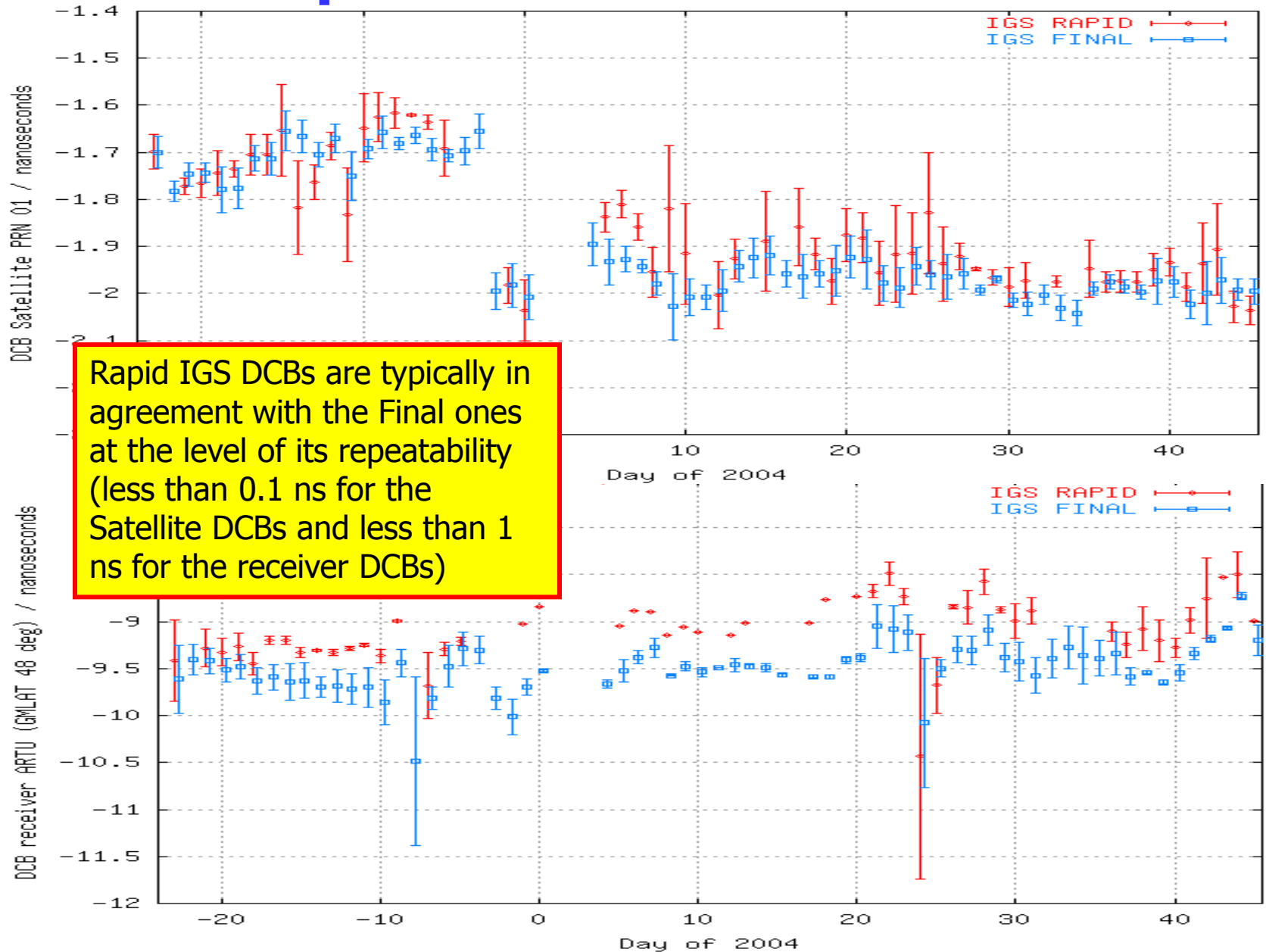
Rapid vs Final IGS TEC “Rel.Error”



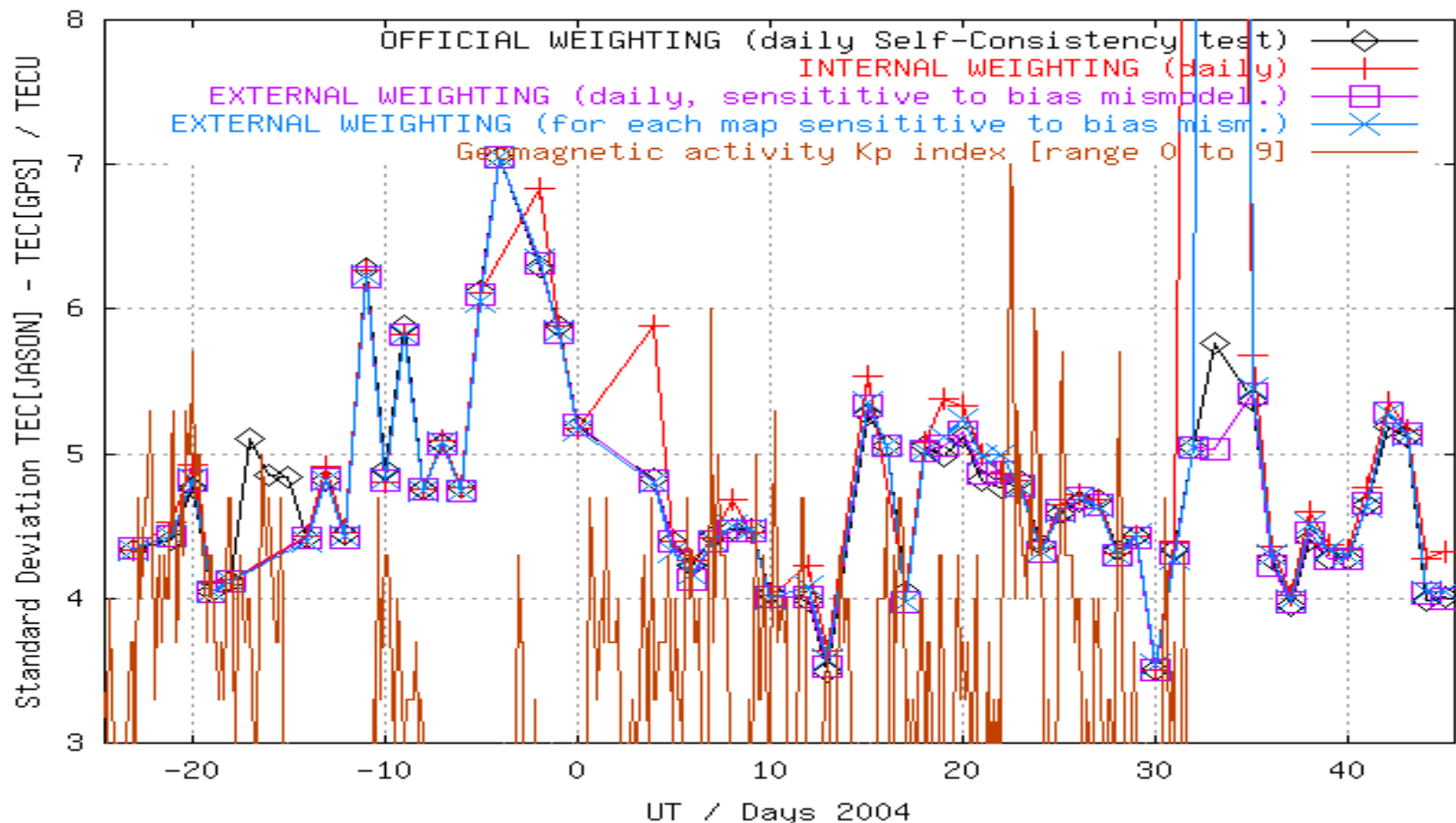
Rapid IGS presents an accuracy (compared with JASON TEC) just about 5-15% worse than the Final IGS TEC when 3-4 rapid IAAC TECs are combined (15-20% worse otherwise).

The generation of a test rapid IGTEC (latencies of less than 24 hours) has started recently (Dec.2003).

Rapid vs Final IGS DCBs

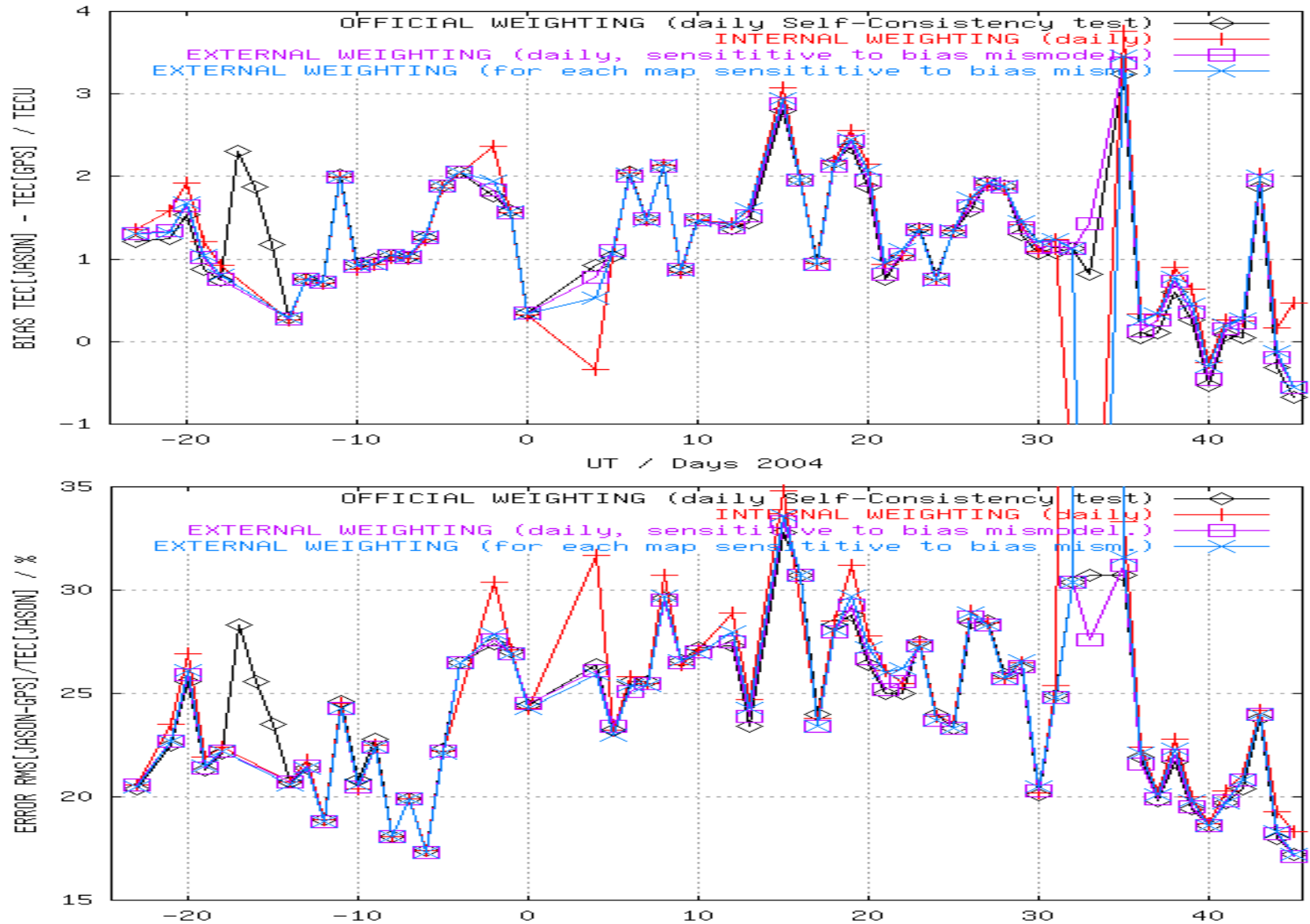


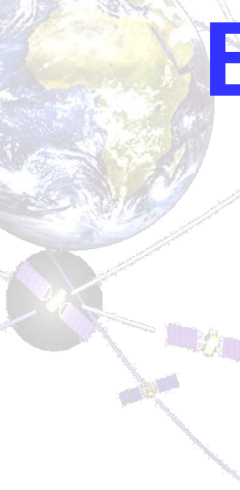
Alternative weighting schemes



- The performance between daily weighting (magenta squares) and independent map weighting (blue crosses) are very similar, including geomagnetic activity periods (days -23 to -19 and 22 to 28, 2004).
- The performance of the official weighting algorithm (bias insensitive) is very similar to that using a bias-sensitive weighting algorithm (truth: STEC referred to the max. elevation ray STEC, both observed).
- It is confirmed the best performance of external weighting strategies (presently used), compared with the internal weighting, from the deviates to the plain mean of IAAC TEC maps (red crosses).

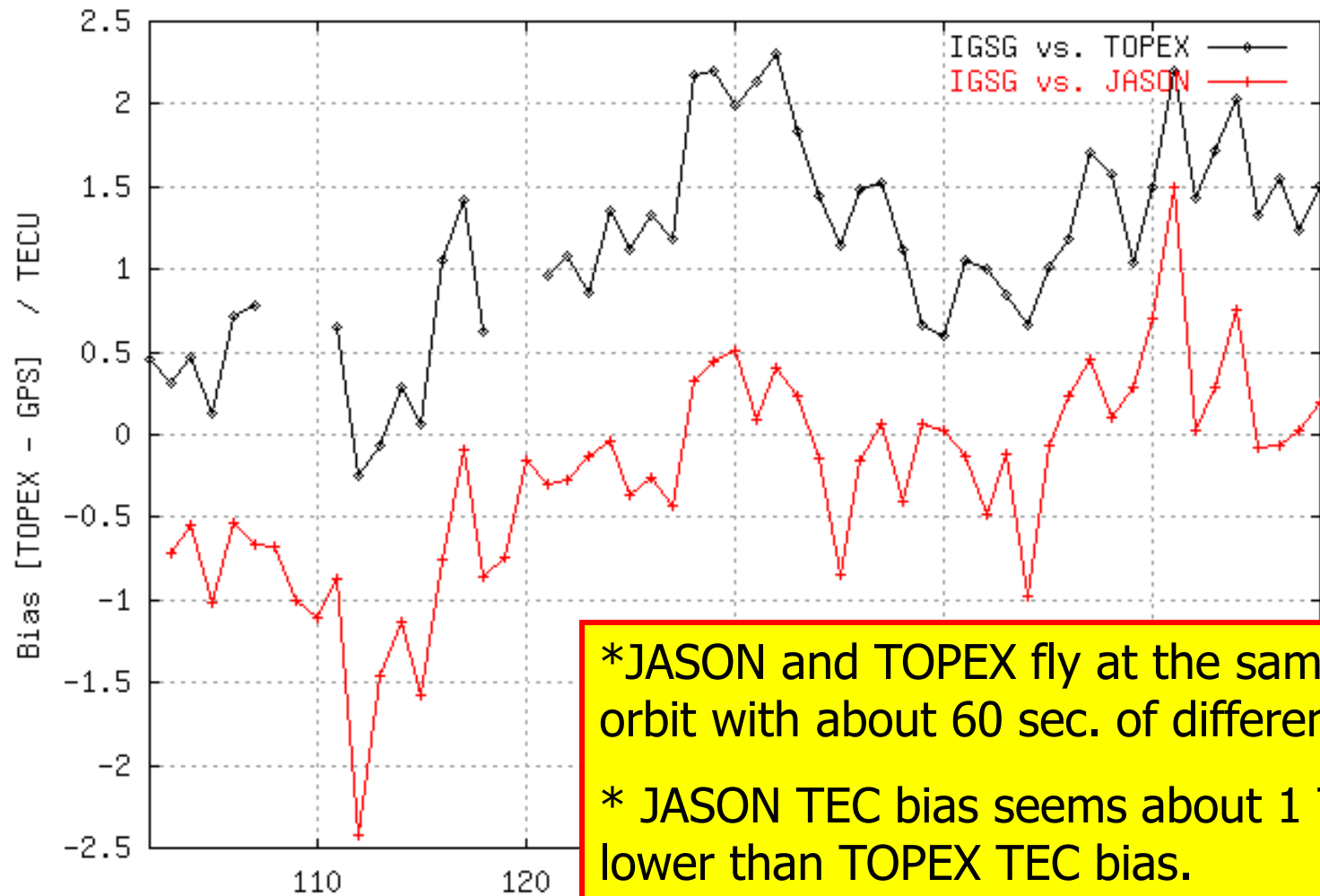
Alternative weighting schemes (2)





Bias vs. JASON & TOPEX (daily)

IGS vs TOPEX TEC and JASON TEC (about 2,000,000 obs.)

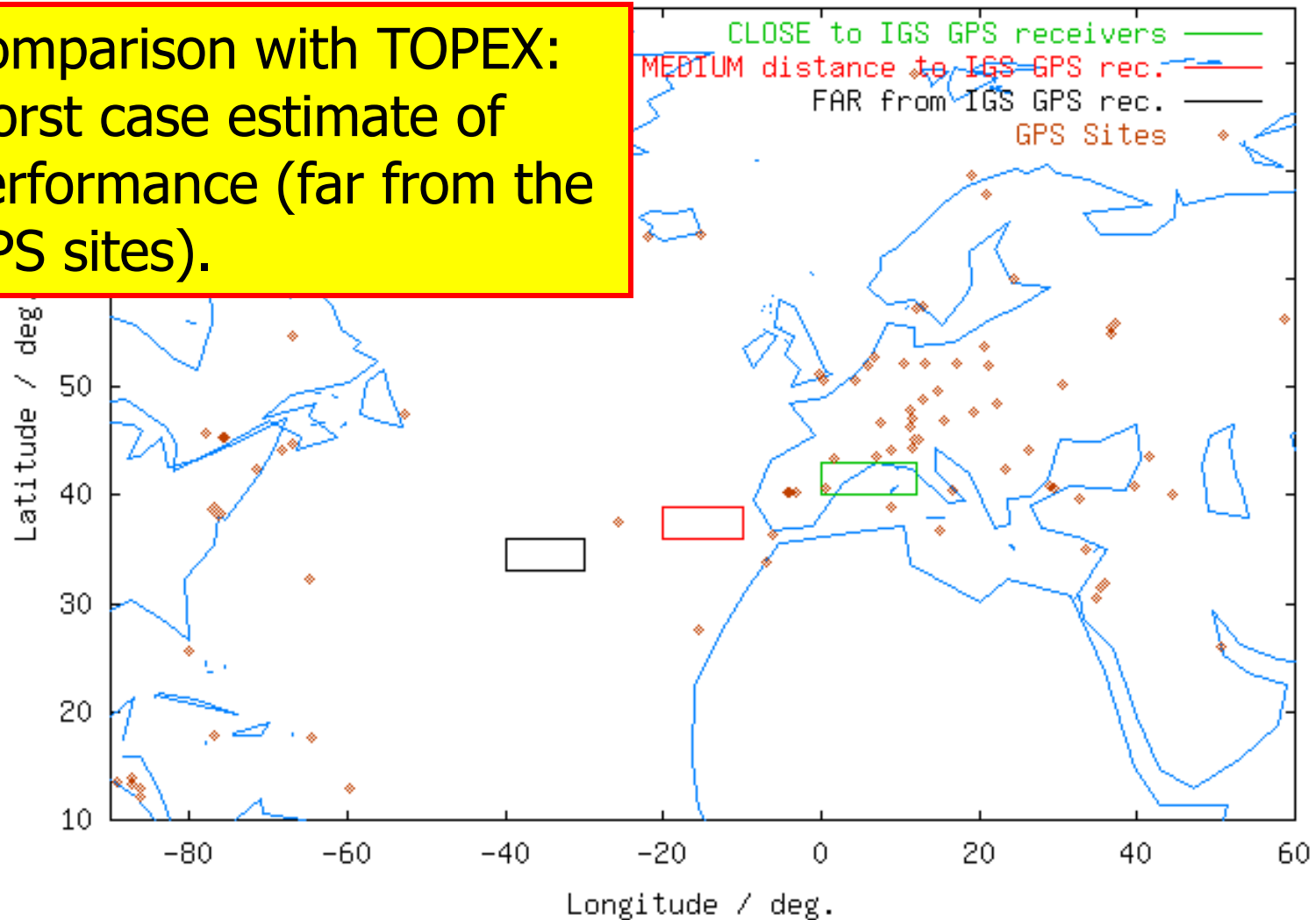


- *JASON and TOPEX fly at the same orbit with about 60 sec. of difference.
- * JASON TEC bias seems about 1 TECU lower than TOPEX TEC bias.
- *JASON TEC seems more compatible with IGS TEC than TOPEX TEC.

TEC performance vs. Distance (1)

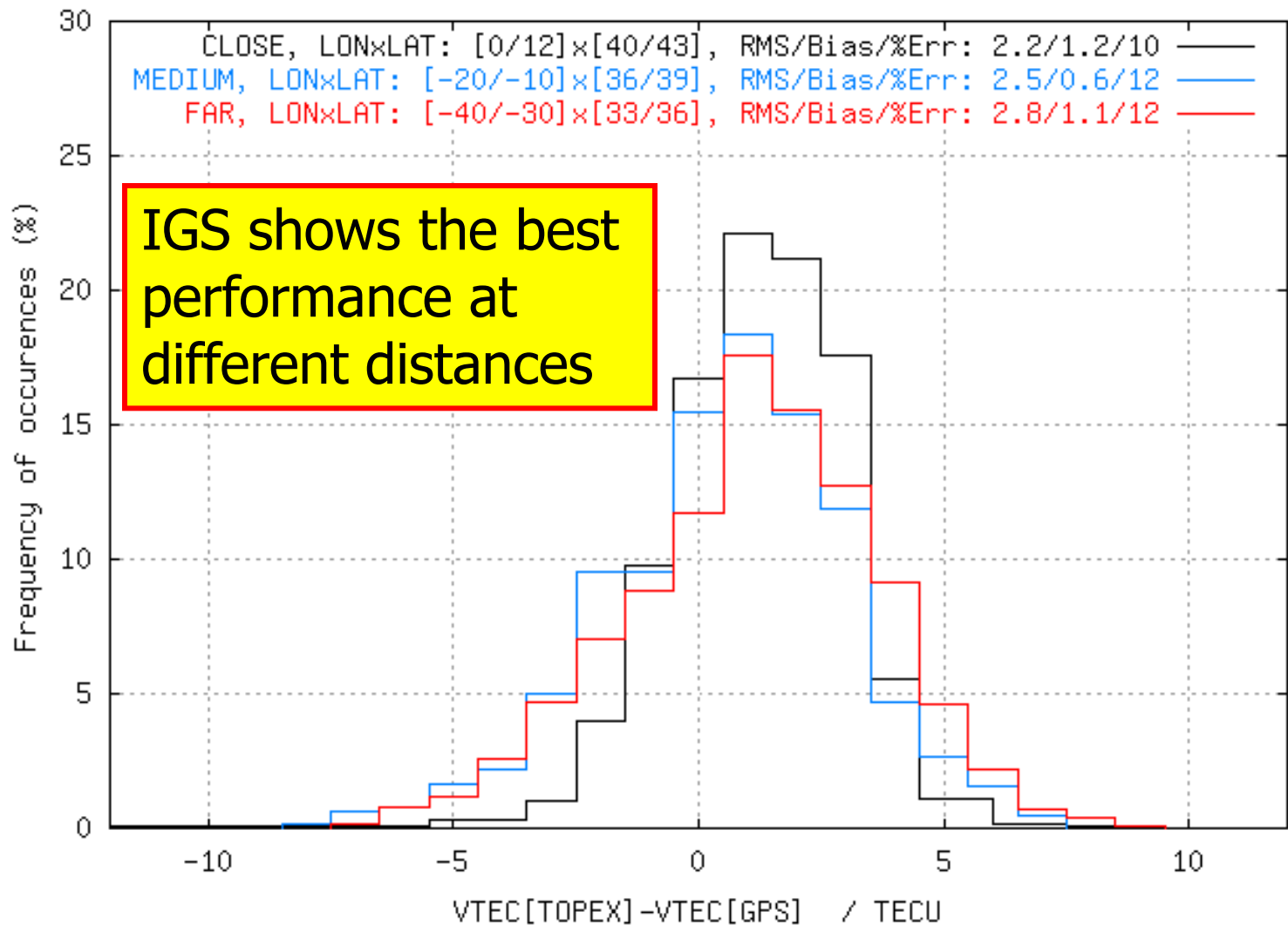
Selected areas at 43 deg. Geom.Lat

Comparison with TOPEX:
worst case estimate of
performance (far from the
GPS sites).



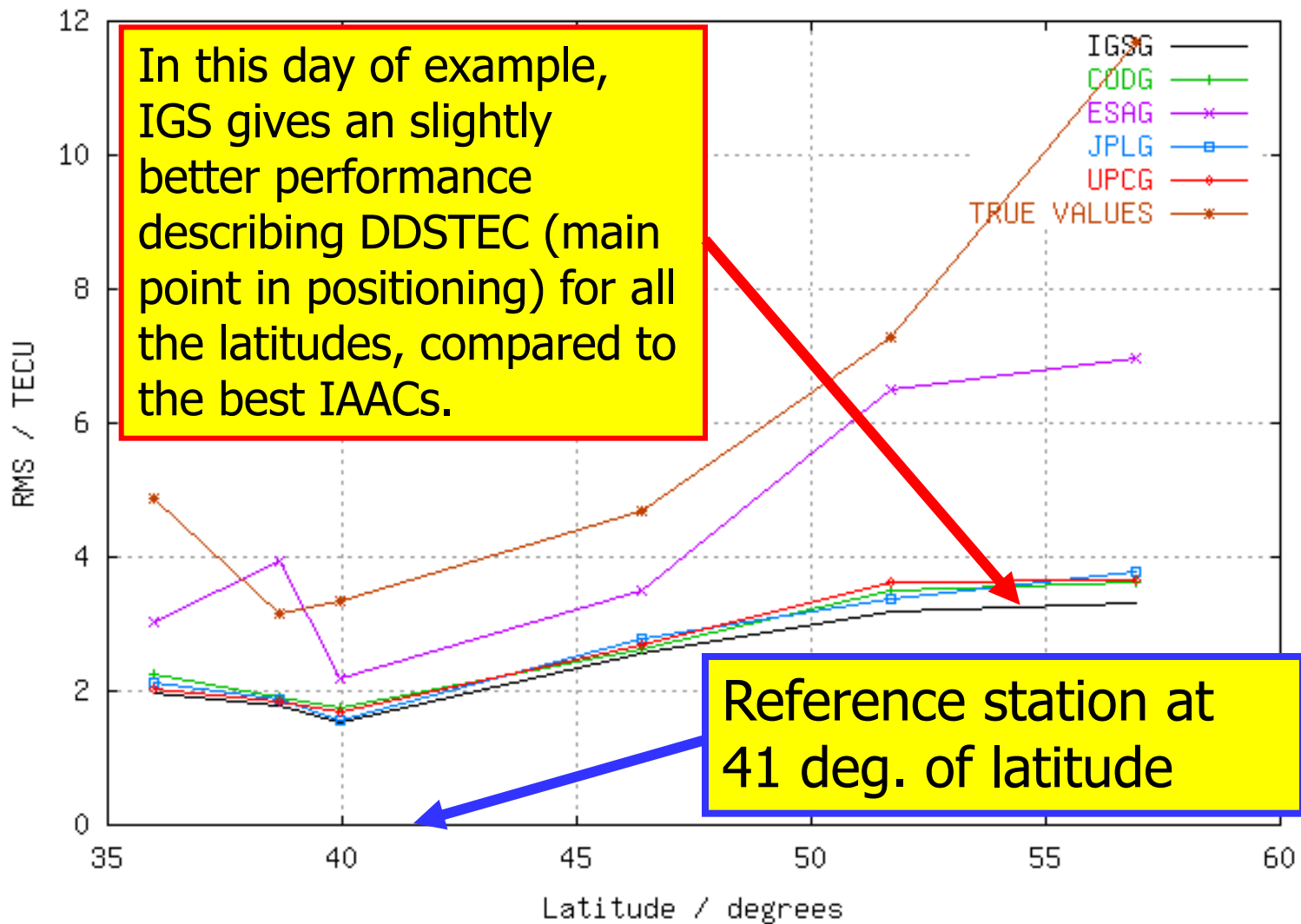
TEC performance vs. Distance (2)

02 Dec.15 / 03 March: IGSG (3 regions: 10x2 deg at 43 deg. gmag. lat.)



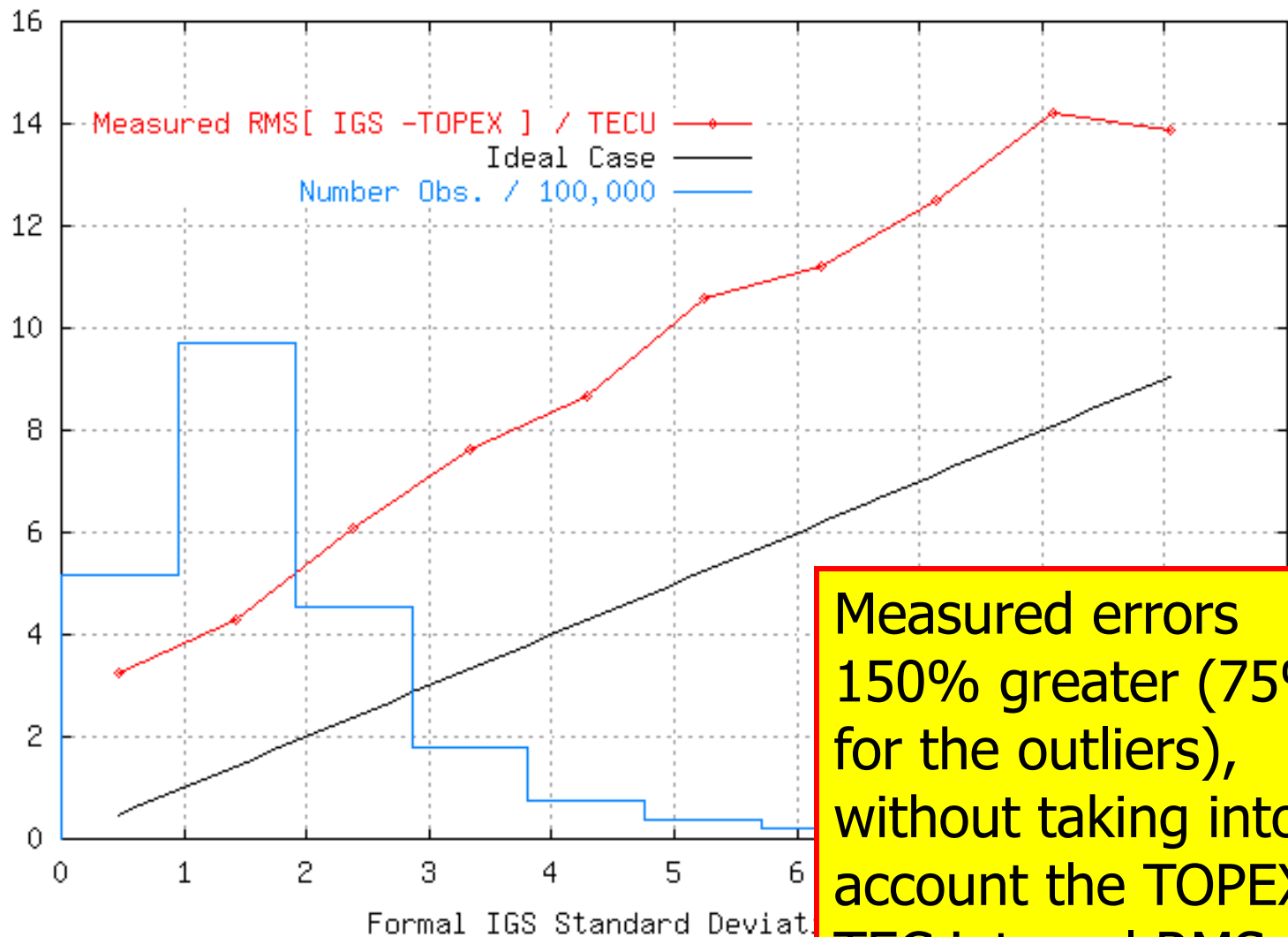
Double Diff. STEC performance

Double Diff. STEC predicted vs. true values (Europe, Day 353, 2002)



TOPEX RMS vs Formal Std.Dev.

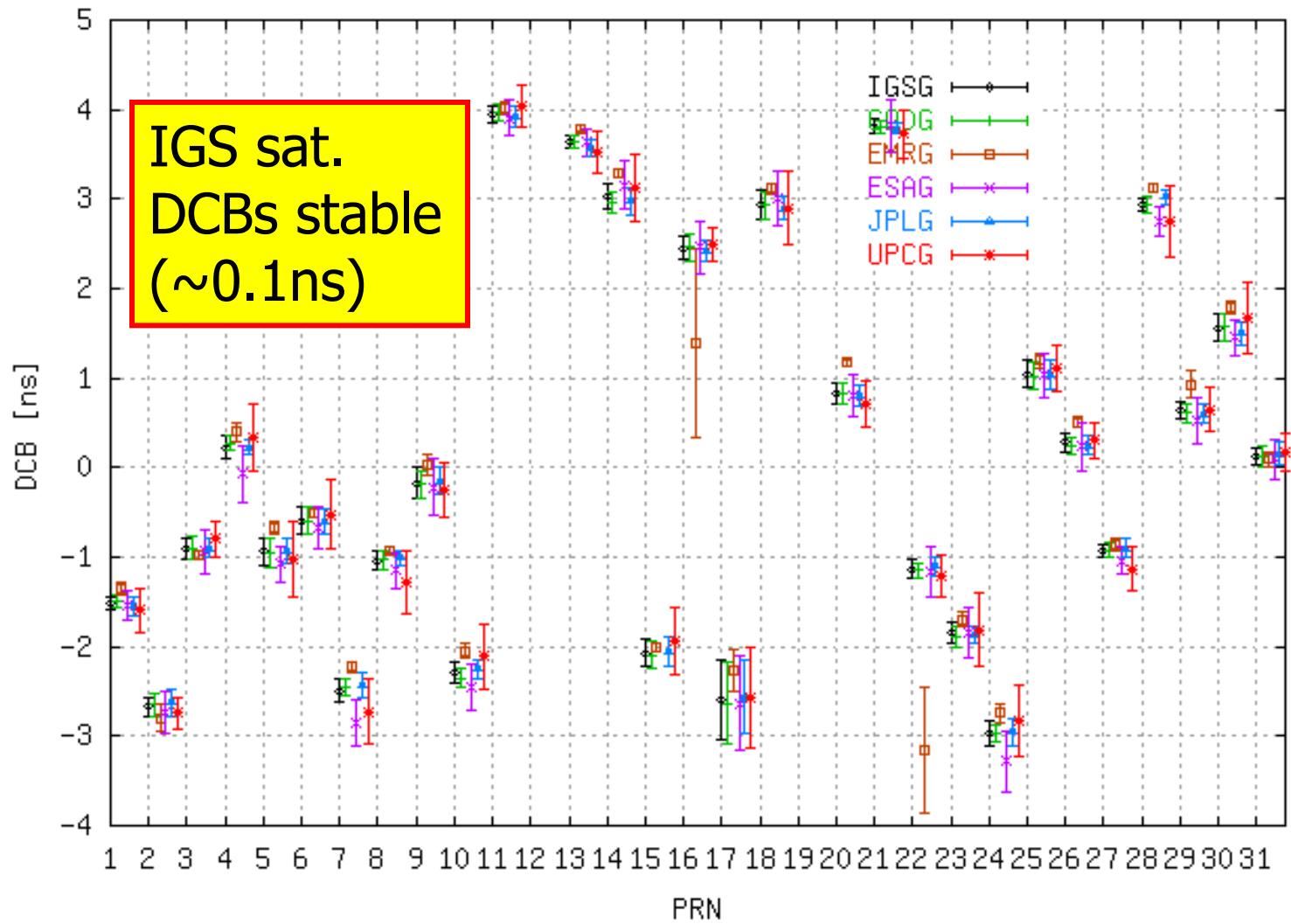
Formal vs. Measured IGS Std.Dev.: 02'Dec15-03'March (2,500,000 TOPEX obs.)



Measured errors
150% greater (75%
for the outliers),
without taking into
account the TOPEX
TEC internal RMS.

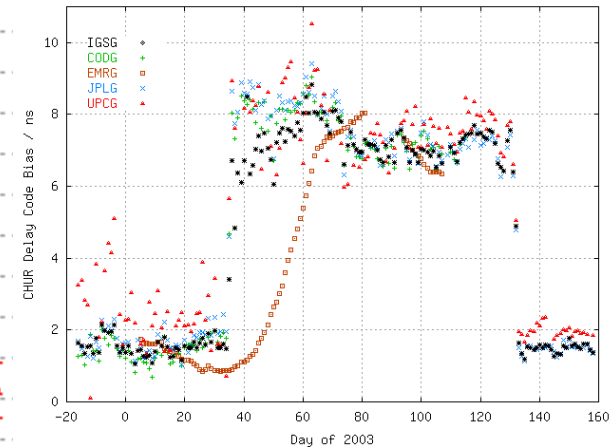
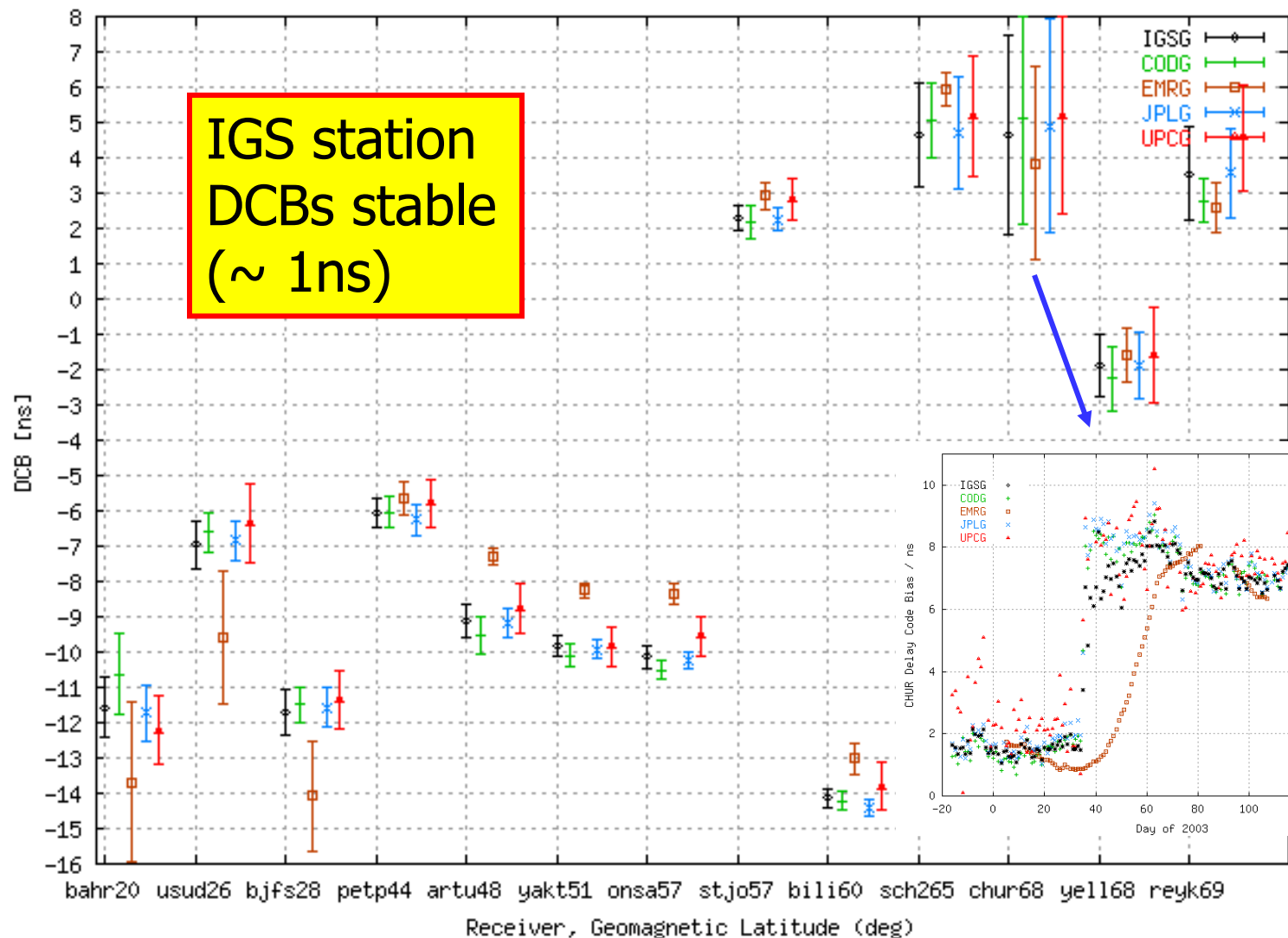
Satellite DCB's stability

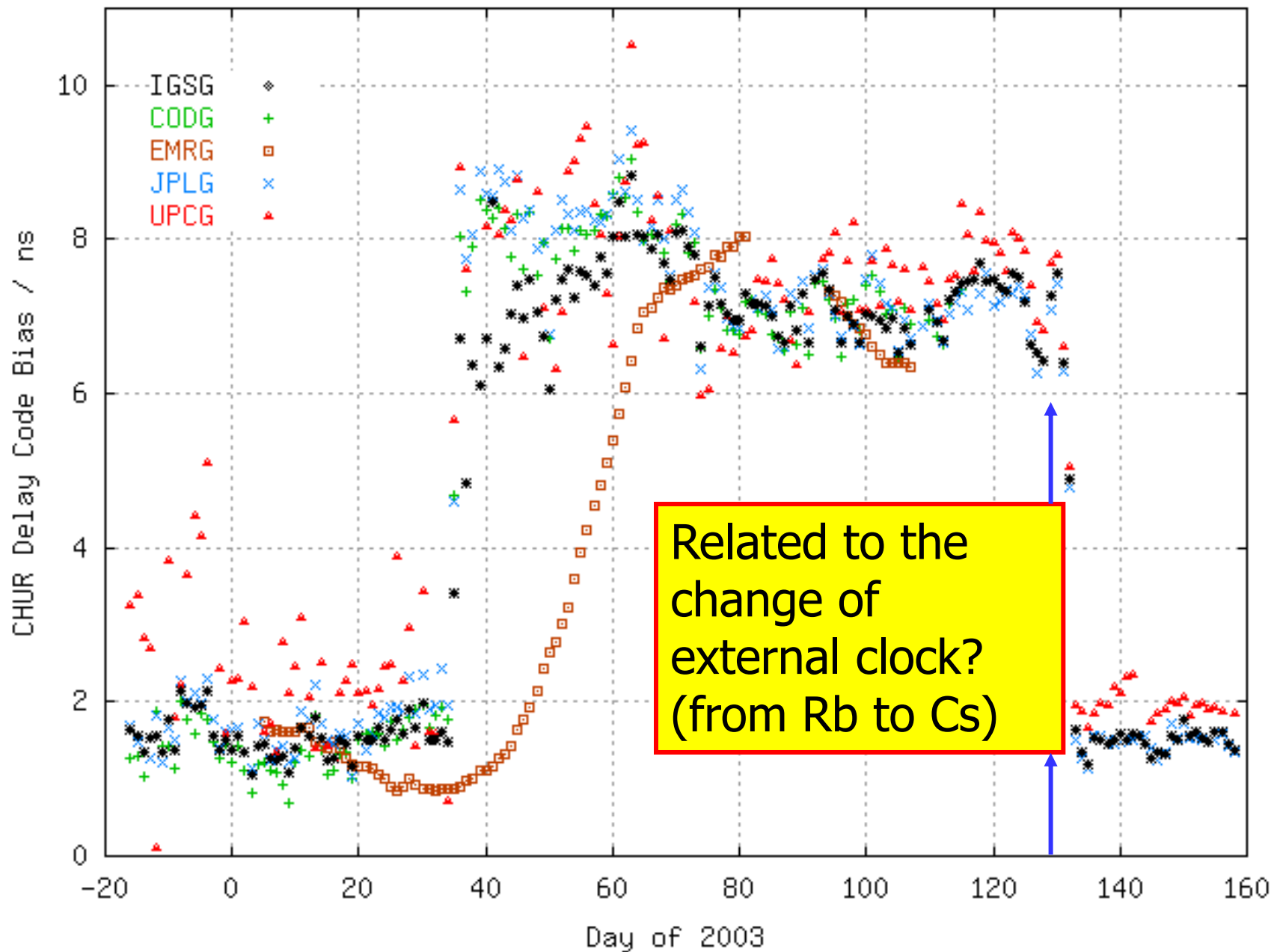
2002 December 15 - 2003 June 7



Station DCB's stability

2002 December 15 - 2003 June 7 (more than 70 daily estimations)





IGS Ionospheric Activities status

- After several improvements performed in the IAACs and IGS Ionosphere Map combination algorithms in 2003, the IGS final combination of TEC maps shows a good performance that is slightly better or even better than the individual IAAC maps, that justified our efforts to start producing the final iono product in April 2003.
- The rapid IGS TEC maps generation started, in testing mode, in December 2003 (latency less than 24 hours, as opposed to 11 days of the final product). The first maps (and DCBs) show a good accuracy (about 10% worse than the final maps) when 3-4 rapid IAACs TEC maps are combined.
- Regarding alternative weighting strategies:
 - The performance of the official weighting algorithm (bias insensitive) is similar to that using a bias-sensitive algorithm.
 - It is confirmed the best performance of external weighting strategies (presently used), compared with the internal weighting.
 - The performance between daily and independent map weighting seems very similar, including geomagnetic activity periods.

Final Ionosphere product use (2003, from cddisa)

- 453 distinct IPs (hosts) downloaded IONEX files in 2003.
- 154,000 files downloaded.
- 29 countries.
- Top “Non-IAAC” users (60% of files downloaded from them, E=Education, C=Commercial, G=Government, N=Network):
 - APL, US (E), TU, Vienna, Austria (E), NAO, Japan (E), CRL, Japan (G), CABLESPEED, U.S. (C/N), ENST-BRETAGNE, France (E), UWM, Poland (E), WAT, Poland (E), U.TRIESTE, Italy (E), NRAO, U.S. (E), STATE OF TENNESSEE, U.S. (G), IGN, France (G), ISP, Russia (N), ALCATEL, France (C), ISP, S.Korea (N), U.MUNICH, Germany (E).
- 68% files downloaded from “Non-IAAC” users.





Some recommendations (updated from the IGS Iono. WG splinter meeting happened at the IGS Technical Meeting held at Bern, March 2004)

- The use of the final IGS product is quite large. However for the rapid product, started in Dec.2003, very few downloads are registered from its temporarily server at UPC. In this context to promote its use, the next actions items have been adopted:

1. To send a new e-mail to the IGS e-mail list announcing the rapid product.
2. Moving the igs-iono e-mailing list to igs.cb.
3. Moving rapid product server from UPC to CDDISA.

- After receiving inputs from VLBI, Altimeter and Timing users, it has been decided:

1. To maintain the present generation of both final and rapid IGS TEC maps.
2. To include the list of GPS receivers used for timing in the list of IAAC used stations to compute the ionospheric product, in order to ensure IGS DCB estimations for such receivers.

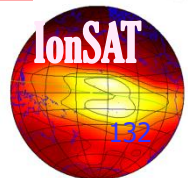
- There has not been consensus between the IAACs on increasing the temporal, spatial resolution/densification of the present ionospheric products.

IGS Final Ionosphere ionex files at
<ftp://cddisa.gsfc.nasa.gov/gps/products/ionex/>

Thank you!



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•Application Example: WARTK and WARTK-3

Introduction

- A GNSS navigation service with errors of few centimeters at continental scale will enhance existing applications (civil eng....), allowing new ones (real-time mapping...).
- The differential ionospheric refraction limits typically its coverage to baselines <20 km (RTK).
- Until now, the authors had combined an accurate ionospheric model: (1) with RTK in two-frequency systems (GPS: WARTK), and (2) with TCAR in three-frequency systems (Galileo and Modernized GPS: WARTK-3, extending the Local Area Galileo services to continental scales).
- The main goal of this presentation: to show the WARTK/WARTK-3 consolidation and improvement algorithm, WARTK-3 Laboratory campaign data generation and results, taking into account two main improvements:
 - *Improving the integrity of the ionospheric corrections.*
 - *Integration in a navigation filter (WARTK-3.2).*

Why does a GPS user need external ionospheric corrections better than 2.7cm (0.25 TECU) to navigate with sub-decimeter accuracy?

- We can get sub-decimeter accuracy when we are able to convert the precise carrier phases in precise and accurate pseudoranges by estimating and fixing the integer ambiguities. This speeds up very much the convergence and the accuracy by eliminating such ambiguity unknowns.
- One typical dual-frequency user can navigate with ionosphere-free combination of carrier phases (Lc), but the iono-free ambiguity (Bc) is not integer, and cannot be fixed and removed.
- Solution: to estimate accurately a second independent ambiguity, the geometric free one (BI), which is linear combination of both L1 and L2 integer ambiguities, thanks to very accurate external double-differenced ionospheric corrections.

$$\nabla \Delta B_I = \nabla \Delta L_I - \nabla \Delta STEC_{ext}$$

- Then the L1 ambiguity can be estimated (the widelane ambiguity Nw can be estimated from a rough Bc and BI).

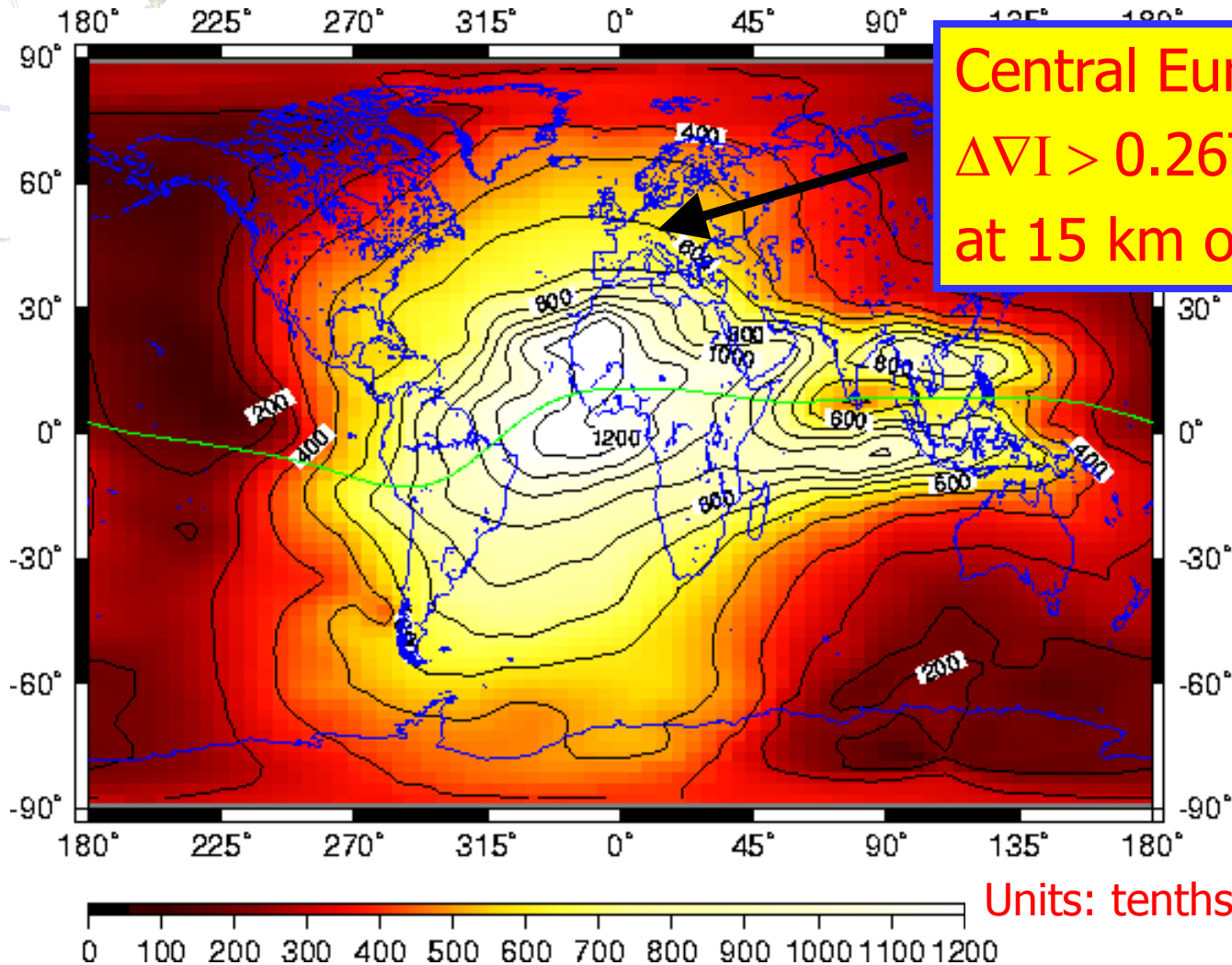
$$\nabla \Delta N_1 = \frac{-1}{\lambda_2 - \lambda_1} \nabla \Delta B_I + \frac{\lambda_2}{\lambda_2 - \lambda_1} \nabla \Delta N_w$$

- And N1 can be fixed and removed if the iono. error is lesser than 2.7cm=0.25TECU

$$\frac{1}{\lambda_2 - \lambda_1} \varepsilon[\nabla \Delta B_I] < 0.5 \text{ cycles} \Rightarrow \varepsilon[\nabla \Delta B_I] \approx \varepsilon[\nabla \Delta STEC_{ext}] < \frac{\lambda_2 - \lambda_1}{2} = 2.7 \text{ cm}$$

Vertical Ionospheric Refraction

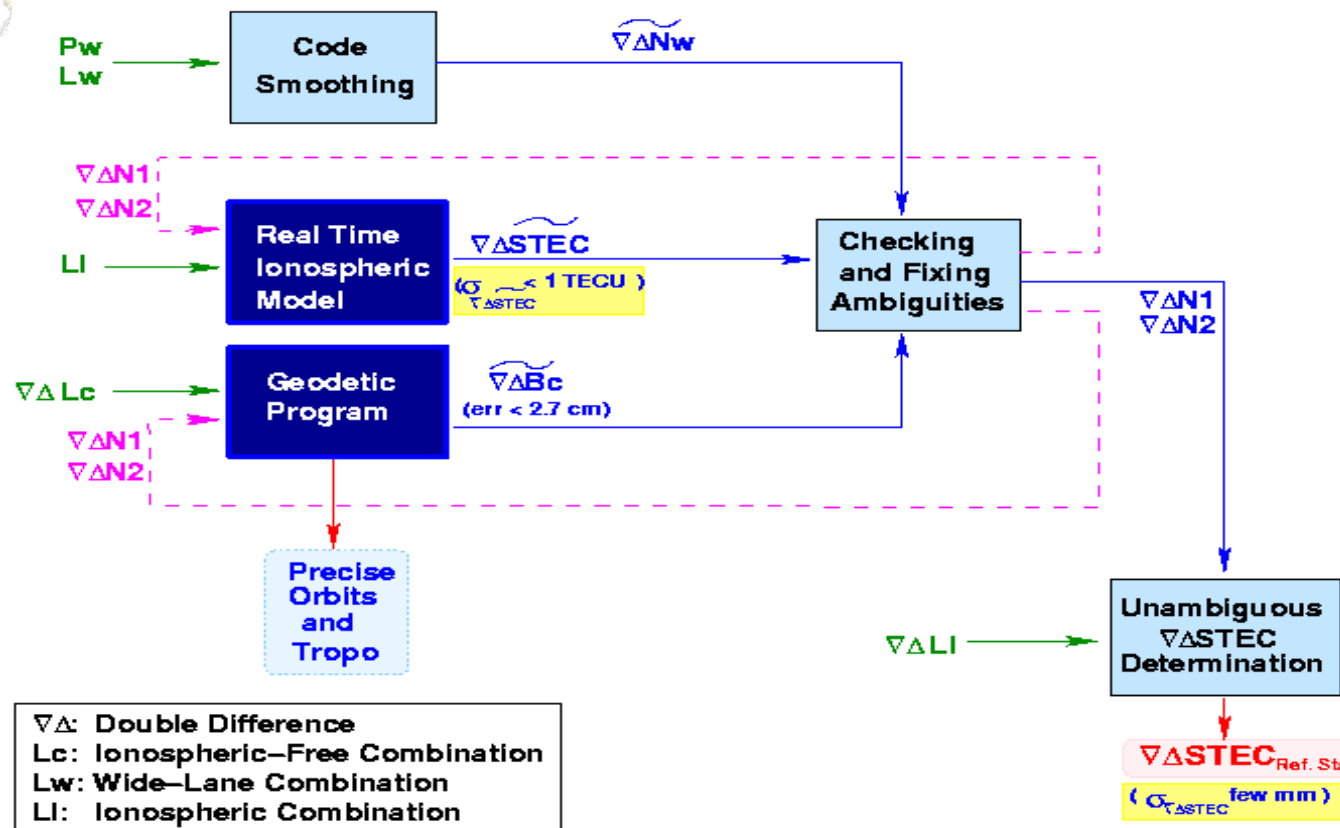
UPC Global Ionospheric Map: 077 2000, 15UT



WARTK: Combining real-time Ionospheric & Geodetic models

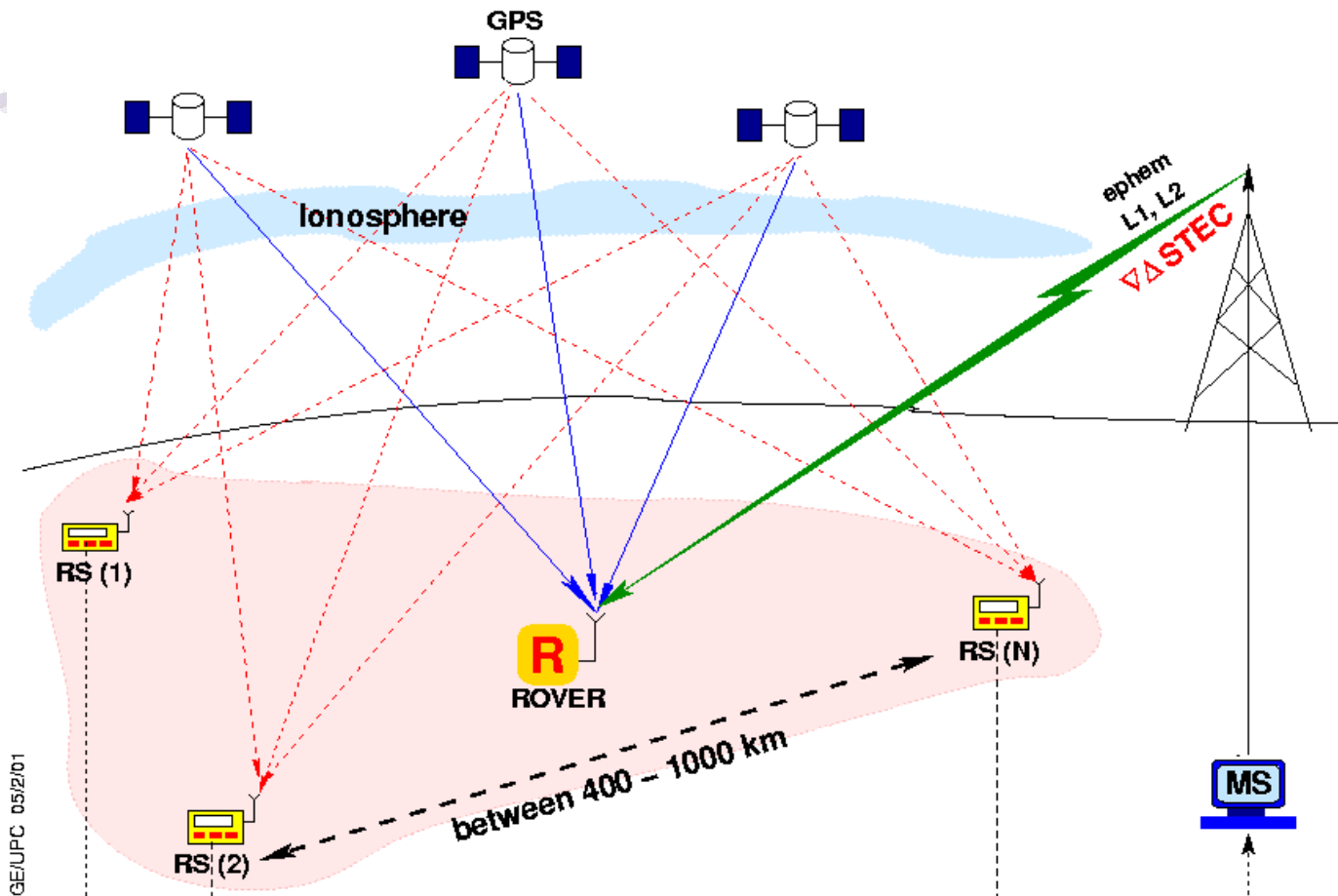
Resolving the Ambiguous $\nabla\Delta$ STEC in Real Time for the Reference Stations

gAGE/UPC 24/07/01



WARTK: Can be based on Wide Area GPS Networks (i.e. EGNOS)

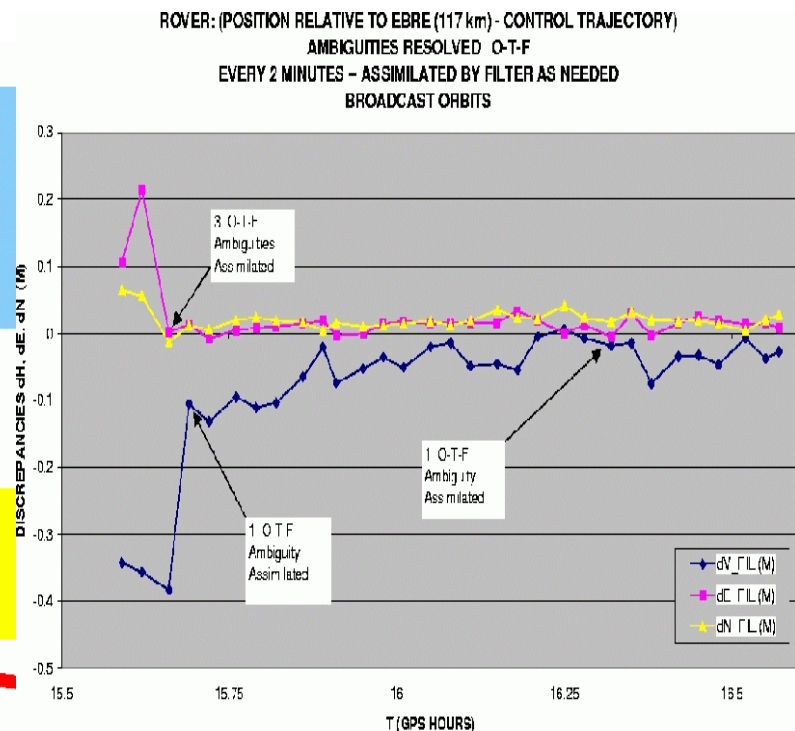
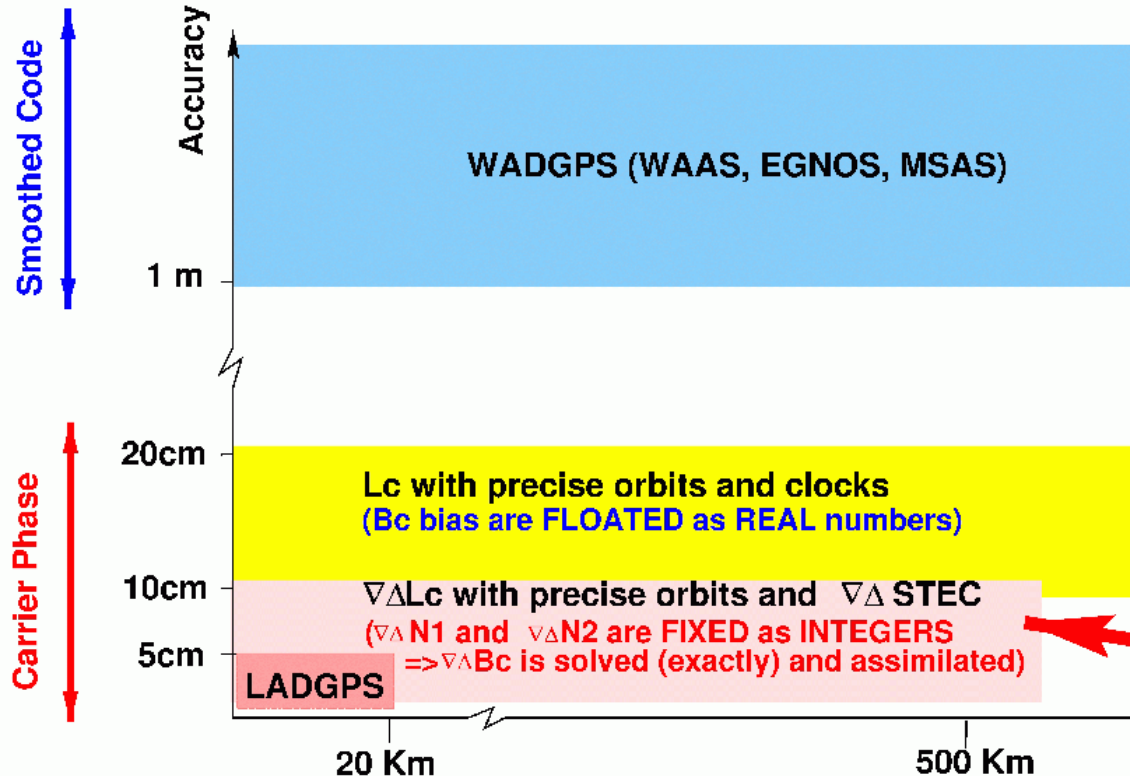
Long-Baseline (hundreds Km) OTF Ambiguity Resolution



IMAGE/UPC 05/2/01



WARTK(GPS): cm-Navigation at hundreds km far



LC: Ionospheric-Free Combination

gAGE/UPC 05/2/01

WARTK*: experiments & results

Experiment	Shortest baseline/km Rover/Fixed	Ionospheric Activity	Fixed Rec. Ambiguity success %	Roving Rec. Ambiguity success %	Kind of rover	Region	Dates	Reported in
BellKin99	116 / 286	Mid.Solar Cycle & Quite	97	80-100	4x4 Car	Catalonia, NE Spain	23-03-99	Colombo et al. 99 (ION)
NWPacific1	400/900	Mid Solar Cycle & Active high lat. (Kp=6)	90-100	80	IGS Site	NWCanada-USA	03-05-98	Hernández et al. 00a, Colombo et al. 00 (GRL, PLANS)
NWPacific2	162/900	Mid-Low Solar Cycle & Irreg.	95-100	80-90	IGS Site	NWCanada-USA	28-04 to 01-05-98	Hernández et al. 00b (ION)
SolarMax1	130/500	Solar Maximum	85-95	80	IGS Site	Central Europe	19 to 22-04-00	Hernández et al. 01 (GRL)
SolarMax2	130/500	Solar Max. & Supestorm	50-95	80	IGS Site	Central Europe	12 to 15-07-00	Hernández et al. 00b (ION)
Baltic99	144/285	Travelling Iono. Disturb. (TIDs)	97	83	IGS Site	North Europe	25-08-99	Hernández et al. 01b (ION)
Equator01	1000-3000/.	Solar Max. & Equator & Very Active (Kp to 9)	90	---	IGS Sites	Central Asia to Oceania	06-03 to 02-04-01	Hernández et al. 02 (JGR)
TCARdata (simulated)	130/300	Solar Max.	100	92 (single-epoch)	Sim. car	Central Europe	17-03-00 (noon)	Hernández et al. 03a-b (IEEE TGARS, Navigation)
UNBAR01	70-115/100	Solar Max. & Strong TIDs	100	~80 (with integrity)	Car	Barcelona, NE Spain	11-06-01	This project
WARTK3 Lab.Test 1-2	178-238 /250-600	Solar Max.	100	100	1:Car 2:Air.	Iberian Peninsula	31-03-90 (ionos.)	This project
WARTK3 Lab. Test 3	416/250-600	Solar Max.	100	99	Fixed Site	Iberian Peninsula	31-03-90 (ionos.)	This project

Conclusions

- Different techniques and results summarized in this presentation show the **maturity of the GPS Ionospheric Modelling techniques in order to provide an unique view of the Ionosphere (in terms of temporal and spatial resolution) and to make feasible new applications.**

Thank you!

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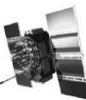
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